

# ROSETTA MARS EXPRESS VENUS EXPRESS

## Radio Science Experiments RSI / MaRS / VeRa

DSN ODF (Orbit Data File) Calibration Software:  
Doppler Level 1b to Level 2  
Software Design Specifications

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**Rosetta Radio Science Investigations RSI****Mars Express Orbiter Radio Science Experiment MaRS****Venus Express Radio Science Experiment VeRa****DSN ODF (Orbit Data File) Calibration Software : Level 1b to Level 2**

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## ACRONYMS

A/D	Analog/Digital
AGC	Automatic Gain Control
AGVTP	Archive Generation, Validation and Transfer Plan
AOL	Amplitude Open Loop
ATDF	Archival Tracking Data Format
CD-ROM	Compact Disk - Read Only Memory
CL	Closed-Loop
DDS	Data Delivery System
DSN	Deep Space Network
DVD	Digital Versatile Disk
ESA	European Space Agency
ESOC	European Space Operation Center
ESTEC	European Space Technology Center
FOL	Frequency Open Loop
G/S	Ground Station
HGA	High Gain Antenna
ODF	Intermediate Frequency Modulation System
JPL	Jet Propulsion Laboratory
LCP	Left Circular Polarization
LGA	Low Gain Antenna
LOS	Line Of Sight
MaRS	Mars Express Radio Science Experiment
MGA	Medium Gain Antenna
MGS	Mars Global Surveyor
NASA	National Aeronautics and Space Administration
ODF	DSN Original Data File
ODR	Original Data Record
OL	Open-Loop
ONED	one-way dual-frequency mode
ONES	One-way single-frequency mode
PDS	Planetary Data System
POL	Polarization Open Loop
RCP	Right Circular Polarization
RSR	Radio Science Receiver
RX	Receiver
S/C	Spacecraft



SIS	Software Interface Specification
S-TX	S-Band Transmitter
SPICE	Space Planet Instrument C-Matrix Events
TBC	To Be Confirmed
TBD	To Be Determined
TWOD	Two-way dual-frequency mode
TWOS	Two-way single-frequency mode
USO	Ultra Stable Oscillator
X-TX	X-band Transmitter

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# 1 INTRODUCTION

## 1.1 SCOPE

This document specifies the requirements for the development of the Orbit data File (ODF) calibration software, transferring Level 1b ODF Doppler data towards Level 2. The software shall analyze radio Doppler tracking data recorded at the DSN ground stations.

## 1.2 REFERENCED DOCUMENTS

	Reference Number	Title	Issue Number	Date
[1]	MEX-MRS-IGM-IS-3016	Radio Science File naming Convention	11.0	24.10.2004
[2]	TRK-2-18	Orbit Data File Interface	change 3	15.06.2000
[3]	MEX-MRS-IGM-DS-3037	ODF Processing Software: L1a to L1b	3.4	08.11.2005

### 1.3 SOFTWARE CONFIGURATION CONTROL

This document addresses the software package

#### DSN\_ODF\_PROC\_DOP\_L1B\_TO\_L02

##### Version 1.2

After release, the software is under configuration control which will be documented in this section.

Version number	Changes/Action	New version	Release date
V1.0	First working release	V1.1	24.05.2005
V1.1	Impact Parameter for Phobos and Solar corona implemented Generation of L1B directory implemented Modification of the log file (now similar to that of the IFMS Doppler processing software)	V1.2	30.08.2005
V1.2	Automatisation of getting the names of the input files Automatic processing implemented		15.11.2005

## 2 SPECIFICATIONS FOR LEVEL 1B TO LEVEL 2 CALIBRATION

### 2.1 MAIN PROGRAM SPECIFICATIONS

#### 2.1.1 General specifications

**ODF-DOP-SPEC-2110:** This software shall

- Read Level 1b ODF data
- Apply Earth troposphere calibration to the Doppler data
- Compute differential Doppler values if two frequencies are available and the sample interval of S-band and X-band is equal
- Apply plasma calibration to the Doppler data either by using differential frequency data or by using a so called Klobucher Modell for the ionosphere of the earth (see section 3.4.4 for more information)
- Apply the Doppler predicts in order to compute residuals
- Output the results as level 2 data
- Generate PDS label files for the output files

**ODF-DOP-SPEC-2120:** the software language is FORTRAN and Perl.

**ODF-DOP-SPEC-2122:** The data processing options are

- (a) occultation
- (b) global gravity,
- (c) target gravity
- (d) solar corona
- (e) Phobos

is selected via a graphical user interface. This graphical user interface will be described in section 5.1.

## 2.1.2 Modules

The main program uses a number of modules

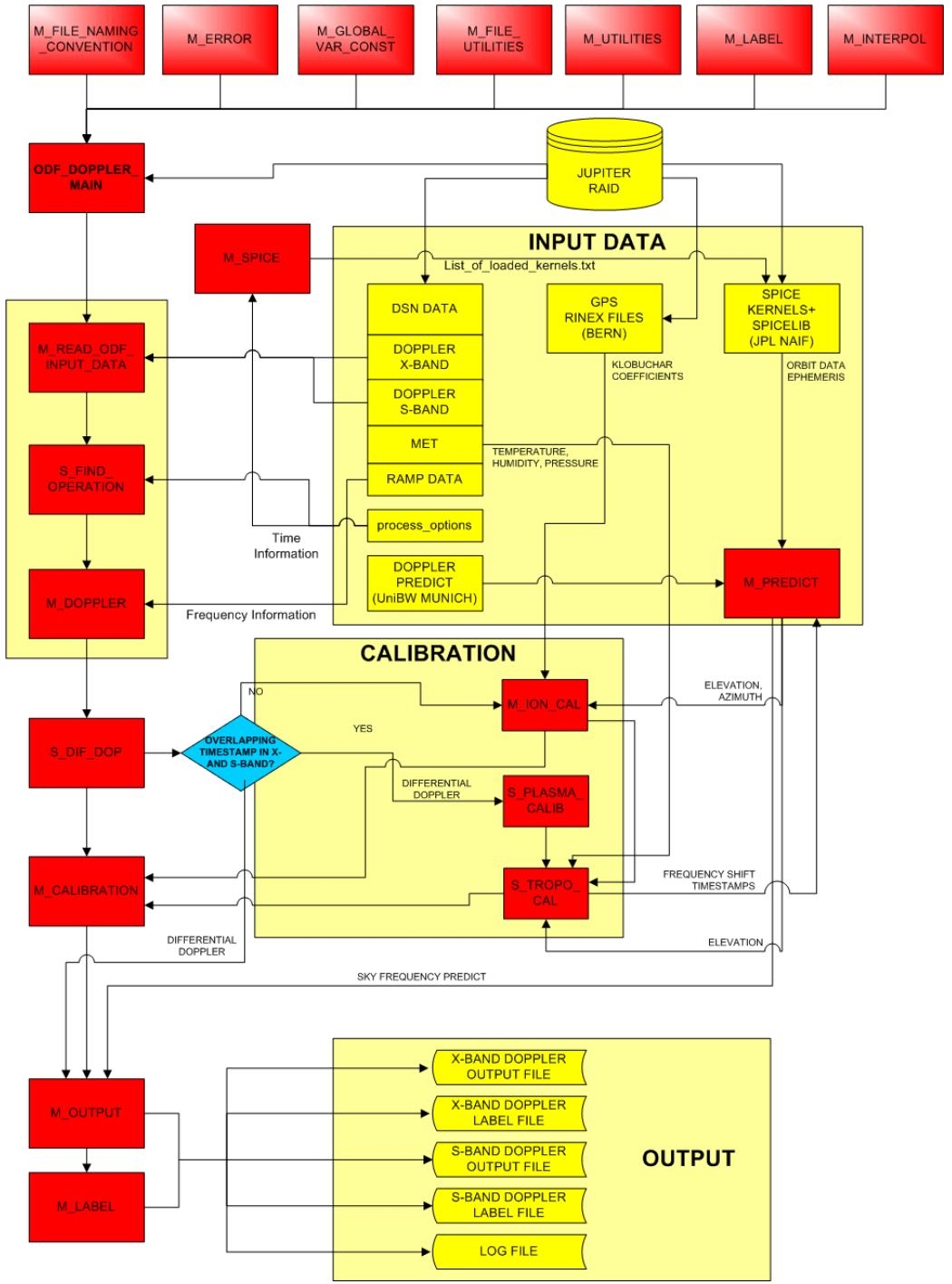
1. M\_READ\_ODF\_INPUT
2. M\_ODF\_SETTINGS
3. M\_PREDICT
4. M\_CALIBRATION
5. M\_IONO\_CALIB
6. M\_OUTPUT
7. M\_TRACKING\_TIME
8. M\_GLOBAL\_VAR
9. M\_DOPPLER

and some general modules, wherein shared subroutines and functions are provided

10. M\_FILE\_UTILITIES
11. M\_SPICE
12. M\_ERROR
13. M\_UTILITIES
14. M\_FILE\_NAMING\_CONVENTION
15. M\_LABEL
16. M\_INTERPOL
17. M\_SEARCH
18. M\_TYPE\_PARAMETER

The software describing flow diagram is shown in [Figure 2-1](#). There are only shown the internal dependencies in the stand alone Fortran software. No interdependencies between the Fortran software and the graphical user interface are shown. The only interaction between the Fortran and the Perl software is done by the `odf_process_option.txt` file which is produced by the Perl graphical user interface and contains processing information. A detailed description of the `odf_process_option.txt` file is given in section 5

Gelöscht: Figure 2-1





**Figure 2-1: Flowchart for evaluation software for the ODF Doppler data**

### 2.1.3 Input files

#### 2.1.3.1 Data file types

**ODF-DOP-SPEC-2130:** the following table defines the input file types and the logical file names used in this specification and within the software:

File Description	Logical name within program
S-band Doppler file	ODF_DOP_S
X-band Doppler file	ODF_DOP_X
Uplink ramp rate file	ODF_RAMP
DSN Media calibration	DSN_MET_MOD
Doppler predict file	PREDICT_FILE
Klobuchar coefficients for Earth ionosphere calibration	ION_COEFF
SPICE Kernels	N/A

**ODF-DOP-SPEC-2135:** input file names will be selected automatically based on information given in the graphical interface described in section 4.3.1.1 via a Windows interface if the processing should be done manually and if the processing should be done automatically via a log file described in section 4.3.1.1.

Gelöscht: 4.4

**ODF-DOP-SPEC-2140:** Klobuchar coefficients are only needed if X-band and S-band Doppler files have no overlapping timestamp or the kind of data processing is occultation

#### 2.1.3.2 File names

**ODF-DOP-SPEC-2145:** Level 1b file names are defined in [1] section 4.1

For the Doppler files:

If the sample interval is 60 seconds:

**r00ODF0L1B\_DPS\_yyddhhmm\_qq.TAB**  
**r00ODF0L1B\_DPX\_yyddhhmm\_qq.TAB**

and if it is 1 second:

**rggODFsL1B\_DPS\_yyddhhmm\_qq.TAB**  
**rggODFsL1B\_DPX\_yyddhhmm\_qq.TAB**

where s can be X for a X-band file and S for a S-band file and serves as identifier for files with 1 second sample interval.

for the ramp rate file:

**rg00DSN0\_L1B\_RMP\_yydddhmm\_qq.TAB**

for the meteorological file:

**rggDSN0L1B\_MET\_yydddhmm\_qq.TAB**

For the predict file:

**rggUNBWL02\_RTW\_yydddhmm\_qq.TAB**

### 2.1.3.3 File Formats

**ODF-DOP-SPEC-2150:** File formats are defined in [1] and [3].

## 2.1.4 Definition of constants

**ODF-DOP-SPEC-2160: ASTRONOMICAL UNIT (AU)**

$$1 \text{ AU} = 149,597,870 \text{ kilometers}$$

**ODF-DOP-SPEC-2165: SPEED OF LIGHT**

$$c = 299,792,458 \text{ m/s}$$

**ODF-DOP-SPEC-2170: RANGE UNIT (RU)**

$$1 \text{ RU} = 0.30 \text{ m}$$

**ODF-DOP-SPEC-2175: PHYSICAL CONSTANTS**

Constant		Value	SI units
Electron charge	e	1.6022 10 <sup>-19</sup>	A s
Electron mass	m <sub>e</sub>	9.1094 10 <sup>-31</sup>	kg
Electric field constant	ε <sub>0</sub>	8.8542 10 <sup>-12</sup>	s <sup>4</sup> A <sup>2</sup> m <sup>-3</sup> kg <sup>-1</sup>
Plasma constant	$\frac{1}{2} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0}$	40.30924	m <sup>3</sup> s <sup>-2</sup>

**ODF-DOP-SPEC-2180: CARRIER FREQUENCIES Mars Express**

Mars Express:

frequency band	uplink	downlink
S-band	2114.676 MHz	2296.482 MHz
X-band	7116.936 MHz	8420.432 MHz

**ODF-DOP-SPEC-2185: Transponder constants and ratios**

Mars Express:

frequency band uplink	transponder ratios downlink/uplink	
	S-band	X-band
S-band	240/211	880/211
X-band	240/749	880/749

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### 3 LEVEL 1B TO LEVEL 2 SOFTWARE SPECIFICATIONS

#### 3.1 MODULE M\_READ\_ODF\_INPUT

Module M\_READ\_ODF\_INPUT contains subroutines and functions in order to read data from ODF\_DOP\_S, ODF\_DOP\_X, ODF\_RAMP and DSN\_MET\_MOD. All data are read in from ODF\_DOP\_S, ODF\_DOP\_X, ODF\_RAMP and DSN\_MET\_MOD. ODF\_DOP\_S, ODF\_DOP\_X and ODF\_RAMP can contain data from several ground stations and timestamps with different sample intervals.

**ODF\_DOP\_SPEC-3100:** The data in ODF\_DOP\_S, ODF\_DOP\_X and ODF\_RAMP are stored in arrays and transferred to M\_TRACKING\_TIME to analyze it and find the appropriate data of predetermined operations.

**ODF\_DOP\_SPEC-3110:** The data in DSN\_MET\_MOD are stored in an array and transferred to M\_CALIBRATION in order to compute the tropospheric calibration

#### 3.2 MODULE M\_TRACKING\_TIME

Module M\_TRACKING\_TIME provides routines in order to analyze the data contained in ODF\_DOP\_S, ODF\_DOP\_X and ODF\_RAMP and find the appropriate data of predetermined operations.

**ODF-DOP-SPEC-3200:** Subroutine S\_FIND\_OPERATION accepts the start time, stop time and the respective ground station of one or more operations from the odf\_process\_options.txt file. For more details on the odf\_process\_options.txt file see section 5.

**ODF-DOP-SPEC-3210:** Subroutine S\_FIND\_OPERATION find in the arrays containing the data from ODF\_DOP\_S, ODF\_DOP\_X and ODF\_RAMP the respective start index and stop index of the array for the predetermined operations defined in the odf\_process\_options.txt file.

**ODF-DOP-SPEC-3220:** The start and stop index of one or more operations are stored in the self defined data type t\_log and transferred to M\_DOPPLER.

**ODF-DOP-SPEC-3230:** The arrays containing the data from ODF\_DOP\_S, ODF\_DOP\_X and ODF\_RAMP are transferred to M\_DOPPLER in order to compute the received antenna frequency and to reconstruct the uplink frequency for the specified operations.

### 3.3 MODULE M\_DOPPLER

Module M\_DOPPLER provides several subroutines and functions in order to compute the received antenna frequency and to reconstruct the uplink frequency.

**ODF-DOP-SPEC-3300:** Subroutine S\_CALC\_REC\_FREQ accepts each array containing the data from ODF\_DOP\_S or ODF\_DOP\_X and the respective start and stop index of one specified operation.

**ODF-DOP-SPEC-3310:** The received antenna frequency  $f_{antenna}$  is computed in subroutine S\_CALC\_REC\_FREQ according to the following equation and transferred to M\_OUTPUT.

$$f_{antenna}(t_i) = k \cdot f_{ref}(t_i) - f_{obs}(t_i)$$

Whereas

$k$  the transponder ratio depending on the uplink and downlink frequency (see [ODF-DOP-SPEC-2185](#) for available values),

$f_{ref}$  the reference frequency, column 13 in ODF\_DOP\_S or ODF\_DOP\_X and

$f_{obs}$  the observed Doppler, column 12 in ODF\_DOP\_S or ODF\_DOP\_X indicates.

Reconstruction of the uplink frequency

This is done for each frequency band.

**ODF-DOP-SPEC-3320:** Subroutine S\_REC\_UP\_FREQ accepts the array containing data from ODF\_RAMP and the respective start and stop index of one specified operation.

**ODF-DOP-SPEC-3330:** The uplink frequency  $f_{up}$  is computed in subroutine S\_REC\_UP\_FREQ via the following equation.

$$f_{up}(t_i) = \Delta t \cdot f_{rate}(t_i) + f_{ramp}(t_i) \quad \text{with} \quad \Delta t = t_i - t_{start, f_{ramp}}$$

Whereas

$t_i$  the current time stamp,

$f_{rate}$  the ramp rate, column 9 in ODF\_RAMP,

$f_{ramp}$  the ramp start frequency, column 10 in ODF\_RAMP, and

$t_{start, f_{ramp}}$  the ramp start time, column 2 in ODF\_RAMP, indicates.

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**Gelöscht:** ODF-DOP-SPEC-2185



### 3.4 MODULES M\_CALIBRATION AND M\_IONO\_CALIB

Module M\_CALIBRATION provides several subroutines in order to correct for the contribution by the propagation through the plasma and the neutral Earth troposphere.

Module M\_IONO\_CALIB provides several subroutines to compute the correction for the Earth ionosphere, i.e. the ionospheric delay in nanoseconds, using the so called Klobuchar model for the Earth ionosphere.

Plasma media correction can only be performed if two downlink frequencies have been recorded and is done only for gravity observations. If only one frequency is available or for occultation observations, the Earth ionosphere is corrected via the Klobuchar model.

#### 3.4.1 Tropospheric calibration

Subroutine S\_TROP\_CALIB uses the meteo data observed at the respective ground station to compute the path delay (unit is meter) of the dry and wet component of the Earth troposphere and calculate from the path delay the total correction for the Earth troposphere in Hz.

##### ODF-DOP-SPEC-3405: Tropospheric calibration

The path delay (unit is meter) of the dry and wet component of the Earth troposphere is (Hofmann-Wellenhoff et al., Global Positioning System, 4<sup>th</sup> Ed.):

$$\Delta_{dry}(E) = \frac{10^{-6}}{5} \frac{77.64 \frac{p}{T}}{\sin(\sqrt{E^2 + 6.25})} [40136 + 148.72(T - 273.16)] \quad (1.1)$$

$$\Delta_{wet}(E) = \frac{10^{-6}}{5} \frac{-12.96T + 3.718 \cdot 10^5}{\sin(\sqrt{E^2 + 2.25})} \frac{e}{T^2} 11000$$

where  $p$ ,  $T$  and  $e$  are the atmospheric pressure, Temperature and partial water vapour pressure, respectively, as observed at the ground station site.

These values are given in the ODF\_METEO file. The elevation angle  $E$  (unit in degrees) is provided by M\_PREDICT.

The following transformations have to be applied:

	equation (20)	ODF_METEO	M_PREDICT
<b>pressure <math>p</math></b>	mbar	hPascal	-
<b>Temperature <math>T</math></b>	Kelvin	°Celsius	-
<b>Water vapour partial pressure <math>e</math></b>	hPascal	-	-
<b>humidity <math>h</math></b>	-	% humidity	-
<b>elevation <math>E</math></b>	degrees	-	radian

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The relation between the water vapour partial pressure and the humidity given in ODF\_METEO is:

$$e = 6.108 \cdot 10^{-2} \cdot \text{humidity} [\%] \cdot \exp \left\{ \frac{17.393(T - 272.15)}{T - 33.95} \right\} \quad (1.2)$$

The total tropospheric calibration expressed as delay time in seconds is:

$$\tau_{\text{tropo}} = \frac{2}{c} \{ \Delta_{\text{dry}}(E) + \Delta_{\text{wet}}(E) \} \quad (1.3)$$

for the two-way radio link where  $c$  is the speed of light with definition given in [ODF-DOP-SPEC-2165](#) and

$$\tau_{\text{tropo}} = \frac{1}{c} \{ \Delta_{\text{dry}}(E) + \Delta_{\text{wet}}(E) \}$$

for the one-way radio link.

**ODF-DOP-SPEC-3410:** The correction for the Earth troposphere is then for one-way radio link:

$$m_{\text{ONE}} = \tau_{\text{tropo}} \cdot f_{\text{down}} \quad (1.4)$$

and for the two-way radio link:

$$m_{\text{TWO}} = \tau_{\text{tropo}} \cdot (f_{\text{down}} + f_{\text{up}}) \quad (1.5)$$

where  $m$  is the cycle advance and the shift in frequency is:

$$\Delta f_{\text{ONE,tropo}} = \frac{dm_{\text{ONE}}}{dt} \quad (1.6)$$

and for the two-way radio link:

$$\Delta f_{\text{TWO,tropo}} = \frac{dm_{\text{TWO}}}{dt} \quad (1.7)$$

This is done for each frequency band.

**ODF-DOP-SPEC-3415:** The result from [ODF-DOP-SPEC-3410](#) is transferred to M\_OUTPUT, added to the respective plasma correction described below and the sum is stored in column 11.

**ODF-DOP-SPEC-3420:** The result from [ODF-DOP-SPEC-3410](#) is transferred to M\_OUTPUT and added to the predicted Doppler data (see section 3.6)

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### 3.4.2 Differential Doppler

Subroutine S\_DIFF\_DOP finds out whether ODF level 1a Doppler data at X-band and S-band are overlapping in time. If this is the case and the ODF level 1a Doppler data at X-band and S-band having the same sample interval the differential Doppler is computed.

#### **ODF-DOP-SPEC-3425: Differential Doppler**

The result from ODF-DOP-SPEC-2363 is taken to compute the differential Doppler

$$\delta f = f_{S,antenna} - \frac{3}{11} f_{X,antenna} \quad (1.8)$$

**ODF-DOP-SPEC-3426:** The result from [ODF-DOP-SPEC-3425](#) is transferred to M\_OUTPUT and stored in column 14.

**Formatiert:** Schriftart: Nicht Fett

**Gelöscht:** ODF-DOP-SPEC-3425

### 3.4.3 Plasma calibration using the differential doppler

Subroutine S\_PLASMA\_CALIB calculates the temporal change in electron content from the differential Doppler and the according frequency-shift in antenna frequency at X-band and S-band

#### **ODF-DOP-SPEC-3430: Plasma calibration**

Derive the temporal change in electron content from the differential Doppler and computes the dispersive frequency shift for each frequency band.  $f_s$  and  $f_x$  are downlink carrier frequencies and  $c$  is the speed of light, all defined in section 1.

$$\delta f = -\frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \left\{ \frac{1}{f_s^2} - \frac{1}{f_x^2} \right\} f_s \frac{dl}{dt} \quad (1.9)$$

$$\Rightarrow \frac{dl}{dt} = -\left\{ \frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \right\}^{-1} \frac{\delta f}{f_s} \left\{ \frac{1}{f_s^2} - \frac{1}{f_x^2} \right\}^{-1}$$

#### **ODF-DOP-SPEC-3431: Plasma correction**

The temporal change in electron content will be used to correct for the downlink plasma propagation for gravity observations only:

$$\Delta f_{S,plasma,cal} = \frac{40.31}{c} \frac{1}{f_s} \frac{dl}{dt} \quad (1.10)$$

$$\Delta f_{X,plasma,cal} = \frac{40.31}{c} \frac{1}{f_x} \frac{dl}{dt}$$

If equation (1.9) is applied to equation (1.10), the plasma correction is than

Gelöscht: (1.9)  
Gelöscht: (1.10)

$$\Delta f_{S,plasma,cal} = \delta f \frac{121}{112} \quad (1.11)$$

$$\Delta f_{X,plasma,cal} = \delta f \frac{33}{112}$$

For further details see [APPENDIX B](#)

Formatiert: Englisch (Großbritannien)  
Formatiert: Standard  
Gelöscht: APPENDIX B¶  
Formatiert: Schriftart: Nicht Fett  
Gelöscht: ODF-DOP-SPEC-3431  
Formatiert: Schriftart: Nicht Fett  
Gelöscht: ODF-DOP-SPEC-3431

**ODF-DOP-SPEC-3432:** The result from [ODF-DOP-SPEC-3431](#), is transferred to M\_OUTPUT added to the tropospheric correction described above and the sum is stored in column 11

**ODF-DOP-SPEC-3433:** The result from [ODF-DOP-SPEC-3431](#), is transferred to M\_OUTPUT and added to the predicted Doppler data (see section 3.6)

### 3.4.4 Plasma calibration using the Klobuchar model

If only one frequency is available or the kind of data processing is Occultation, the Earth ionosphere plasma has to be modeled. Module M\_IONO\_CAL contains subroutines in order to provide a model of the electron content of the Earth ionosphere and will be described below in detail.

#### 3.4.4.1 The Klobuchar model

Module M\_IONO\_CALIB contains several subroutines to provide a model of the electron content of the Earth ionosphere at any local time and pointing direction of the ground station antenna and determines the path delay. This is done using the Klobuchar model introducing the Klobuchar coefficients from GPS measurements of the International GPS Service (IGS). The IGS is based on about 200 globally distributed permanent GPS tracking sites. The coefficients used by module M\_IONO\_CALIB come from one of the seven IGS Analysis Center: the Center for Orbit Determination in Europe (CODE) of the Astronomical Institute of the University of Berne (AIUB), Switzerland.

**Rosetta Radio Science Investigations RSI**

**Mars Express Orbiter Radio Science Experiment MaRS**

**Venus Express Radio Science Experiment VeRa**

**DSN ODF (Orbit Data File) Calibration Software : Level 1b to Level 2**

Document number  
MEX-MRS-IGM-DS-3038

Issue: 3  
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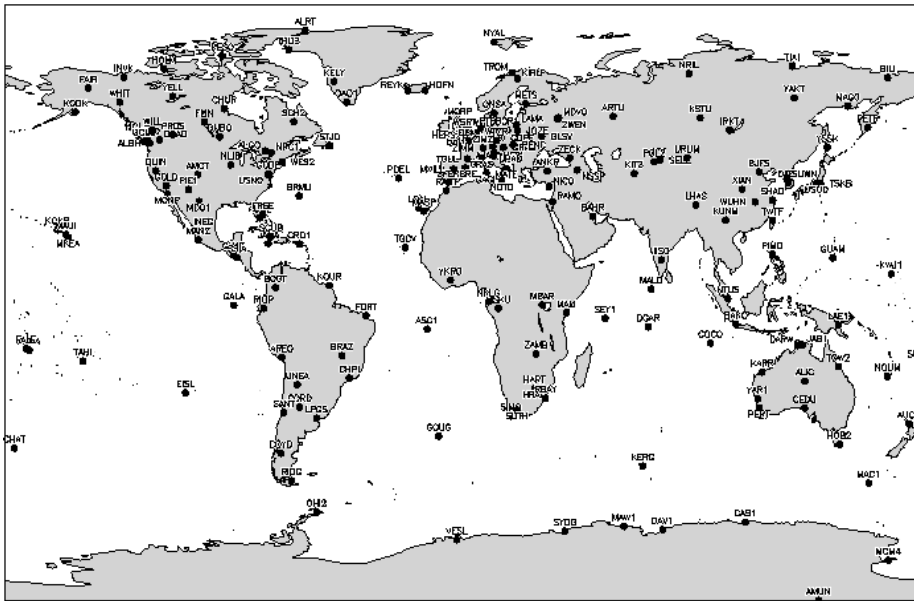
Revision: 0  
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[VEX-VRA-IGM-DS-5009](#)

CODE generates Global ionosphere maps (GIM) on a daily basis using data from about 200 GPS/GLONASS sites of the IGS and other institutions. The vertical total electron content (VTEC) is modelled in a solar-geomagnetic reference frame using a spherical harmonics expansion up to degree and order 15. Piece-wise linear functions are used for representation in the time domain. The time spacing of their vertices is 2 hours, conforming with the epochs of the VTEC maps. Instrumental biases, so-called differential P1-P2 code biases (DCB), for all GPS satellites and ground stations are estimated as constant values for each day, simultaneously with the 13 times 256, or 3328 parameters used to represent the global VTEC distribution. The DCB datum is defined by a zero-mean condition imposed on the satellite bias estimates. P1-C1 bias corrections are taken into account if needed. To convert line-of-sight TEC into vertical TEC, a modified single-layer model mapping (MSLM) mapping function approximating the JPL extended slab model mapping function is adopted. The global coverage of the GPS tracking ground stations considered at CODE is shown figure 3.5.1 including abbreviations for station identification.

GPS Tracking Ground Stations Considered at CODE



**Figure 3-1:** GPS Tracking Ground Stations

CODE computes Klobuchar-style ionospheric coefficients (alphas and betas) best fitting the IONosphere map EXchange data (IONEX) on a regular basis. The description how the Klobuchar coefficients are computed and on which ionospheric model they are based on can be found in ANNEX B.

The data files containing the Klobuchar coefficients are named `CGIMddd0.yyN`, where `ddd` and `yy` substitute doy and 2-digit year. Those coefficients derived from a final IONEX product are stored under <ftp://ftp.unibe.ch/aiub/CODE/> in `yyyy`-specific subdirectories as of 1995. For the few days where the final product is not yet available, rapid as well as predicted coefficients serving real-time applications may be found generally at <ftp://ftp.unibe.ch/aiub/CODE/>. [CGIM2410.04N\\_R](#) contains the latest set of rapid coefficients; [CGIM2420.04N\\_P](#) and [CGIM2430.04N\\_P2](#) contain the current 1-day and 2-day predicted coefficients, respectively.

Unlike the original Klobuchar ionosphere model which is based on a total of 370 possible sets of base coefficients and which is therefore of discrete nature, the model derived by CODE is not subject to a similar restriction. All the night-time TEC level of this type of ionosphere model is hard-wired to 5 nanoseconds of ionospheric delay on the first GPS frequency (corresponding to approximately 9 TECU). Because the Klobuchar-style TEC parameterization may be unpleasant at the polar caps and especially at the poles, CODE displays a corresponding warning in the RINEX navigation data files in case the TEC above a latitude of 75 degrees reaches day-time level. The format of RINEX data files is described in ANNEX C.

The module is currently only valid for the NNO ground station.

**ODF-DOP-SPEC-3440:** Module `M_IONO_CALIB` accepts the actual needed Klobuchar coefficients (described above) from input file `ION_COEFF`. The input file can be downloaded from

[ftp.unibe.ch/aiub/CODE/](ftp://ftp.unibe.ch/aiub/CODE/)

`M_IONO_CALIB` needs several input parameters, which are listed in the table below.

Parameter	Description	Unit
Phi	Geodetic latitude of receiver	Degree
Lambda	Geodetic longitude of receiver	Degree
TOW	Time of Week	Degree
Beta	The coefficients of a cubic equation representing the amplitude of the vertical delay	
Alpha	The coefficients of a cubic equation representing the period of the model	

**Table 3-1:** Input parameter of `M_IONO_CALIB`

**ODF-DOP-SPEC-3441:** The output of Module M\_IONO\_CALIB is the ionospheric slant range correction  $\tau_{iono}$ . The unit of  $\tau_{iono}$  is seconds. The calculation of  $\tau_{iono}$  is described in [ODF-DOP-SPEC-3442](#).

**Formatiert:** Schriftart: Nicht Fett

**Gelöscht:** ODF-DOP-SPEC-2482  
ODF-DOP-SPEC-2482

**ODF-DOP-SPEC-3442:** The computation of the ionospheric slant range correction  $\tau_{iono}$  depends on the local time at the ground station side. For the calculation of  $\tau_{iono}$  the following parameters are used:

**1. Local Time t:**

$$t = 4.32 \cdot \text{long}_i + \text{TOW}$$

**2. Azimuth a (in radian):**

$$a = \text{azimuth} \cdot \pi / 180$$

**3. Elevation angle e (in semicircles):**

$$e = \text{elev} \cdot 1 / 180$$

**4. Earth Centered angle psi:**

$$\text{psi} = 0.0137 / (e + 0.11) - 0.022$$

**5. Subionospheric longitude long\_i :**

$$\text{long}_i = \text{lambda} \cdot 1 / 180 + (\text{psi} \cdot \text{DSIN}(a) / \text{DCOS}(\text{lat}_i \cdot \pi))$$

**6. Subionospheric latitude lat\_i :**

$$\text{lat}_i = \text{phi} \cdot 1 / 180 + \text{psi} \cdot \text{DCOS}(a)$$

**7. Time of the Week TOW (output of the subroutine S\_GPSTIME)**

$$t = \text{DMOD}(t, 86400.)$$

**8. Slant factor sf:**

$$\text{sf} = 1. + 16. \cdot (0.53 - e)^3$$

**9. Period of model PER:**

If PER less than 72000.DO

PER = 72000.

Else

PER = beta(1) + beta(2) ·lat\_m + beta(3) ·lat\_m<sup>2</sup> +beta(4) ·lat\_m<sup>3</sup>

#### 10. Phase of the model x (Maximum at 14.00 =! 50400 sec local time):

$x = 2 \cdot \pi \cdot (t - 50400) / \text{PER}$  !

#### 11. Amplitude of the model AMP:

AMP = alpha (1) + alpha (2) ·lat\_m + alpha (3) ·lat\_m<sup>2</sup> +alpha(4) ·lat\_m<sup>3</sup>

#### 12. Ionospheric slant correction $\tau_{iono}$ :

Night (DABS(x) greater Than 1.57):

$\tau_{iono} = sf \cdot (5.D-9)$

Day:

$\tau_{iono} = sf \cdot (5.D-9 + \text{AMP} \cdot (1.D0 - x^2/2. + x^4/24.))$

at any local time and pointing direction of the ground station antenna and determines the path delay. This is done using the Klobuchar model introducing the Klobuchar coefficients from GPS measurements.

#### 3.4.4.2 Plasma calibration of the antenna frequency

Subroutine S\_PLASMA\_CALIB\_MOD corrects for the contribution by the propagation through the earth ionosphere by using the model for the earth ionosphere defined in module M\_ION\_CALIB.

**ODF-DOP-SPEC-3444:** Subroutine S\_PLASMA\_CALIB\_MOD accepts the ionospheric slant correction  $\tau_{iono}$  from module M\_IONO\_CALIB

**ODF-DOP-SPEC-3445:** The correction for the Earth ionosphere is then

$$m = \tau_{iono} \cdot f_{down}$$

where  $m$  is the cycle advance and the shift in frequency is:



$$\Delta f_{iono} = \frac{dm}{d\tau_{iono}}$$

This is done for each frequency band

**ODF-DOP-SPEC-3450:** The result from [ODF-DOP-SPEC-3445](#) is transferred to M\_OUTPUT added to the tropospheric correction described above and the sum is stored in column 11

**Formatiert:** Schriftart: Nicht Fett

**Gelöscht:** ODF-DOP-SPEC-3445

**ODF-DOP-SPEC-3455:** The result from [ODF-DOP-SPEC-3445](#) is transferred to M\_OUTPUT and added to the predicted Doppler data (see section 3.6)

**Gelöscht:** ODF-DOP-SPEC-3445

**Formatiert:** Schriftart: Nicht Fett

### 3.5 MODULE M\_PREDICT

M\_PREDICT accepts a Doppler predict file: the predict file PREDICT\_FILE considers all possible perturbing forces as the best known gravity field and solar and albedo radiation pressure. For more details about the PREDICT\_FILE see document [3]. M\_PREDICT interpolates for a given time stamp between the predicted Doppler shift for the uplink and the predicted Doppler shift for the downlink respectively for each observed time stamp. M\_PREDICT also computes with the interpolated values the predicted antenna frequency depending on the uplink frequency provided by subroutine S\_REC\_UP\_FREQ and returns an estimated parameter for each observed time stamp. This is done for each frequency band.

**ODF-DOP-SPEC-3510:** M\_PREDICT accepts input data from PREDICT\_FILE with the file name format defined in [1] section 8.1 or in [1] section 8.2 for the predicted orbit or the reconstructed orbit file, respectively. PREDICT\_FILE contains both the Doppler uplink and downlink data.

**ODF-DOP-SPEC-3515:** M\_PREDICT accepts predicted Doppler data from PREDICT\_FILE (file name specified in ODF-DOP-SPEC-2210) formatted as defined in [1] section 8.1 or in [1] section 8.2 for the predicted orbit or the reconstructed orbit file, respectively.

**ODF-DOP-SPEC-3520:** M\_PREDICT\_FILE contains predicted Doppler data with a time period that covers one entire operation.

**ODF-DOP-SPEC-3525:** M\_PREDICT accepts from subroutine S\_REC\_UP\_FREQ the reconstructed uplink frequency  $f_{up}$ .

**ODF-DOP-SPEC-3530:** M\_PREDICT accepts from S\_CALC\_REC\_FREQ the transponder ratio  $k$ .

**ODF-DOP-SPEC-3535:** Subroutine S\_DOP\_PRED\_ODF accepts from M\_READ\_ODF\_INPUT the array TIME\_DOPPLER representing the observed Doppler time stamps.

**ODF-DOP-SPEC-3540:** Subroutine S\_DOP\_PRED reads predicted Doppler data from PREDICT\_FILE and interpolates between the predicted Doppler shift for the uplink and the predicted Doppler shift for the downlink respectively for each observed time stamp given as TIME\_DOPPLER.

**ODF-DOP-SPEC-3545:** S\_DOP\_PRED computes for each frequency band the predicted antenna frequency  $f_{pred,antenna}$  received at a given ground station via

$$f_{pred,antenna}(t_i) = k \cdot f_{up}(t_i) (1 + P_{up} + P_{down} + P_{up} \cdot P_{down})$$

where  $P_{up} = \frac{\Delta f_{up}}{f_{up}} = \frac{v_{r,up}}{c}$  and  $P_{down}$  is the predicted Doppler shifts of the uplink and the

downlink path, respectively. The result is stored in the array DOPPLER\_PREDICT\_SKY. For more details about the computation see Appendix A.

**ODF-DOP-SPEC-3550:** The array DOPPLER\_PREDICT\_SKY is transferred to the module M\_OUTPUT. This is done for each frequency band.

**ODF-DOP-SPEC-3555:** M Subroutine S\_DOP\_PRED reads time values of the two way light time from PREDICT\_FILE and interpolates between each value of the two way light time for each observed time stamp given as TIME\_DOPPLER. This is done for each frequency band.

**ODF-DOP-SPEC-3560:** The resulting values are subtracted from TIME\_DOPPLER at each time stamp in order to compute the transmit frequency ramp reference time.

**ODF-DOP-SPEC-3565:** The result from [ODF-DOP-SPEC-3560](#) is transferred to M\_OUTPUT and stored in column 6.

**Formatiert:** Schriftart: Nicht Fett

**Gelöscht:** ODF-DOP-SPEC-3560

### 3.6 MODULE M\_OUTPUT

Module M\_OUTPUT provides routines to create the output files for each frequency band and the log file containing processing information.

**ODF-DOP-SPEC-3610:** As many output files are created as many individual tracking time ranges have been selected. The start date and time of the input data are used as the time in the output file name defined in [ODF-DOP-SPEC-4110](#). The individual tracking time ranges are selected in Modul M\_TRACKING\_TIME.

**Gelöscht:** ODF-DOP-SPEC-4110

**Formatiert:** Schriftart: Nicht Fett

**ODF-DOP-SPEC-3615:** Subroutine S\_DOPPLER\_OUTPUT accepts the interpolated values for the received antenna frequency  $f_{pred}$  from M\_PREDICT.

**ODF-DOP-SPEC-3620:** Subroutine S\_DOPPLER OUTPUT adds the tropospheric calibration and the plasma correction to the interpolated predicts and stores the result in column 10.

$$f_{pred,calib} = f_{pred} + \Delta f_{iono} + \Delta f_{tropo}$$

This is done for each frequency band

**ODF-DOP-SPEC-3625:** Subroutine S\_DOPPLER OUTPUT computes for each frequency band the frequency residuals  $\Delta f_{res}$  by subtracting the interpolated and corrected, predicted antenna frequency  $\Delta f_{pred,calib}$  from the received antenna frequency  $f_{antenna}$  and stores the result in column 12.

$$\Delta f_{res} = f_{antenna} - f_{pred,calib}$$

This is done for each frequency band.

#### **ODF-DOP-SPEC-3630:** output files

For each selected individual activity an output file is created (for each downlink frequency). The format of the output files is specified in [ODF-DOP-SPEC-4210](#) and [ODF-DOP-SPEC-4230](#).

**Gelöscht:** ODF-DOP-SPEC-4210

**Formatiert:** Schriftart: Nicht Fett

**Formatiert:** Schriftart: Nicht Fett

**Gelöscht:** ODF-DOP-SPEC-4230

#### **ODF-DOP-SPEC-3635:** log file

Subroutine S\_WRITE\_OPT\_FILE creates a log file that contains information about the processing operation e.g. file names of the files used during the processing start and stop time of all processed operations, which calibration was applied, standard deviation of the residuals and so on. A detailed description of the format of the log file is given in section 4.3.1.1.



## 4 OUTPUT FILES

### 4.1 FILE NAMES

**ODF-DOP-SPEC-4110:** The DOPPLER\_OUTPUT file names are defined as

**rggODF0L02\_sss\_yyddhhmm\_qq.TAB**

The definitions are given in Table 2-1.

**Table 4-1:** DOPPLER\_OUTPUT file name Definition

Acronym	Description	Examples
r	Spacecraft (Raumsonde) name R = Rosetta M = Mars Express V = Venus Express	M
gg	Ground station ID: <u>DSN complex Canberra</u> 34 = 34 m BWG 43 = 70 m 45 = 34 m HEF <u>DSN complex Goldstone:</u> 14 = 70 m 15 = 34 m HEF 24 = 34 m BWG 25 = 34 m BWG 26 = 34 m BWG 27 = 34 m HSBWG <u>DSN complex Madrid:</u> 54 = 34 m BWG 55 = 34 m BWG 63 = 70 m 65 = 34 m HEF	43
tttt	data source identifier <u>Level 2</u> ODF0 = DSN ODF closed-loop file with 60 seconds sample interval ODFX = ODFS =	ODF0
lll	Data archiving level L02 = Level 2	L2
sss	data type	DPS

	<u>data level 2:</u> DPS S-band Doppler DPX X-band Doppler	
yy	Year	04
ddd	Day of the year	153
hhmm	Sample hour , minute (Start time)	1135
qq	Sequence or version number	01
eee	.TAB ASCII data files	

**4.2 FILE FORMATS**

**ODF-DOP-SPEC-4210:** The format of the DOPPLER\_OUTPUT\_X file is defined in Table 4-2

**ODF-DOP-SPEC-4220:** if only X-band Doppler data exist, the differential Doppler cannot be computed and is set to “-9999.999”.

**ODF-DOP-SPEC-4230:** The format of the DOPPLER\_OUTPUT\_S file is defined in Table 4-3.

**ODF-DOP-SPEC-4240:** if only S-band Doppler data exist, the differential Doppler cannot be computed and is set to “-9999.999”.

**ODF-DOP-SPEC-4250:** All data that are not available in the data file are set to a default value corresponding to their format description. For example data with format F10.3 are set to -99999.999. This default value indicates that the data is not a valid number and can not be used for further computations. For details see Table 4-2 and Table 4-3.

**ODF-DOP-SPEC-4260:** The first and the last value of column 11 of the DOPPLER\_OUTPUT\_X file and DOPPLER\_OUTPUT\_S\_file (calibration) is set to his default value due to the way of computation.

**Table 4-2** Definition of DOPPLER\_OUTPUT\_X file format

column	description	unit	resolution
1	Sample number		
2	Ground received time <i>as UTC in ISO format</i>		
3	Ground received time <i>as UTC in fractions of day of year starting with the first day of the year the data was recorded at 00:00.000</i>	day	10 <sup>-10</sup> day
4	Ground received time <i>as elapsed terrestrial barycentric dynamic time (TDB) time since noon of the first calendar day of year 2000 (12:00 1 January 2000 TDB)</i>	second	10 <sup>-6</sup> sec
5	Distance <i>Propagation experiments: approximate value of the closest approach of a downlink geometric ray path to the center of the reference body (Sun, planet, minor object). When two-way, the value is approximate average of uplink and downlink rays</i> <i>Gravity observations: geometric distance of the s/c from the center of mass of referenced body</i>	kilometer	10 <sup>-3</sup> m
6	<p>Transmit frequency ramp reference time <i>UTC in ISO format</i></p> <p><i>The time (t<sub>0</sub>) at which the transmitted frequency would have been f<sub>0</sub> using the coefficients f<sub>0</sub> (column 7) and df (column 8). At any time t within the interval when those coefficients are valid, the transmitted frequency f<sub>t</sub> may be calculated from</i></p> $f_t = f_0 + df \cdot (t - t_0)$ <p><i>For DSN two-way measurements:</i>  <i>f<sub>t</sub> is the uplink frequency of the ground transmitter; the f<sub>t</sub> photon will reach the receiver one RTLT later.</i></p> <p><i>For DSN one-way measurements:</i>  <i>f<sub>t</sub> is the downlink frequency of the spacecraft transmitter; the f<sub>t</sub> photon will reach the receiver OWLT later. In both cases, f<sub>0</sub> and df may change; but f<sub>t</sub> is always continuous, and changes in the coefficients occur only on integer seconds.</i></p> <p><i>For ODF measurements:</i></p> $f_t = f_0$		



	<i>because <math>df=0</math>.</i>		
7	Transmit frequency corresponding to time in column 6 <u>Two-way coherent modes:</u> Uplink frequency of ground station S-band order of 2100 MHz X-band order of 7100 MHz <u>One-way mode:</u> S/C transmission frequency X-band order of 8400 MHz S-band order of 2300 MHz	Hertz	$10^{-6}$ Hz
8	Uplink frequency ramp rate <u>DSN two-way coherent:</u> Time derivative of uplink frequency in column 7 <u>DSN one-way downlink mode:</u> Value of spacecraft frequency drift, if known and/or meaningful; -99999.999999 <u>ODF measurements:</u> Ramp rate is always zero; $df=0$	Hertz/sec	$10^{-6}$ Hz/sec
9	Observed X-band antenna frequency Frequency of the signal at the terminals of the receiving antenna structure at UTC TIME columns 2 to 4 ( $t_r$ ). Set to -9999999999.999999 for missing or corrupted data.	Hertz	$10^{-6}$ Hz
10	Predicted X-band antenna frequency Based on the ESOC reconstructed orbit file or SPICE kernels Expected frequency of the signal at the terminals of the receiving antenna structure at UTC TIME in columns 2 to 4 ( $t_r$ ). The calculation includes geometrical effects (relative positions and motions of ground station and spacecraft, including Earth rotation and light time adjustments), tuning of both the transmitter and receiver and a model-based correction for one- or two-way (as appropriate) propagation through the Earth's atmosphere.	Hertz	$10^{-6}$ Hz
11	Correction of Earth atmosphere propagation Correction term for the propagation of the signal in the Earth atmosphere, based on meteorological data observed at the ground station site (MET-files)	Hertz	$10^{-6}$ Hz
12	Residual calibrated X-band frequency shift column 9 minus 10	Hertz	$10^{-6}$ Hz
13	Received signal level	dBm / dB	0.1 dB

	<p><u>Closed-loop data:</u> Signal level from AGC in decibels relative to one milliwatt (dBm).</p> <p><u>Open-loop (RSR):</u> Signal level in decibels (dB) relative to an arbitrary reference.</p>		
14	<p>Differential Doppler</p> $f_s - \frac{3}{11} f_x$ <p>Where <math>f_s</math> and <math>f_x</math> are the received S-band and X-band frequencies            If BAND_NAME = X (from the label file), <math>f_x</math> comes from column 9 in this table and <math>f_s</math> comes from column 9 in the file identified by SOURCE_ID (from the label file).             If BAND_NAME = S (from the label file), <math>f_s</math> comes from column 9 in this table and <math>f_x</math> comes from column 9 in the file identified by SOURCE_ID (from the label file).            if either band is not available, this column is set "-99999.999"</p>	Hertz	$10^{-6}$ Hz
15	<p>standard deviation of the observed antenna frequency X-band in column 9 (open-loop only)            for closed-loop this value is set "-99999.999"</p>	Hertz	$10^{-6}$ Hz
16	<p>Received X-band signal quality (open-loop only)            Ratio of observed received signal strength to the statistical standard deviation of the measurement, column 15 divided by column 19            For closed-loop this is value is set "-999.9"</p>	dB	0.1 dB
17	<p>standard deviation of received signal level at X-band (open-loop)            A statistical measure of the error in determining SIGNAL LEVEL (column 15) based on fit of a data spectrum to a sinc function. Uses the same arbitrary scale factor as column 15; units of dB.            for closed-loop this is set "-999.9"</p>	dB	0.1 dB

**Table 4-3:** Definition of DOPPLER\_OUTPUT\_S file format

column	description	unit	resolution
1	Sample number		
2	Ground received time <i>as UTC in ISO format</i>		
3	Ground received time <i>as UTC in fractions of day of year starting with the first day of the year the data was recorded at 00:00.000</i>	day	10 <sup>-10</sup> day
4	Ground received time <i>as elapsed terrestrial barycentric dynamic time (TDB) time since noon of the first calendar day of year 2000 (12:00 1 January 2000 TDB)</i>	second	10 <sup>-6</sup> sec
5	Distance <i>Propagation experiments: approximate value of the closest approach of a downlink geometric ray path to the center of the reference body (Sun, planet, minor object). When two-way, the value is approximate average of uplink and downlink rays</i> <i>Gravity observations: geometric distance of the s/c from the center of mass of referenced body</i>	kilometer	10 <sup>-3</sup> m
6	Transmit frequency ramp reference time <i>UTC in ISO format</i> <i>The time (t<sub>0</sub>) at which the transmitted frequency would have been f<sub>0</sub> using the coefficients f<sub>0</sub> (column 7) and df (column 8). At any time t within the interval when those coefficients are valid, the transmitted frequency f<sub>t</sub> may be calculated from</i> $f_t = f_0 + df \cdot (t - t_0)$ <i>For DSN two-way measurements:</i> <i>f<sub>t</sub> is the uplink frequency of the ground transmitter; the f<sub>t</sub> photon will reach the receiver one RTLT later.</i> <i>For DSN one-way measurements:</i> <i>f<sub>t</sub> is the downlink frequency of the spacecraft transmitter; the f<sub>t</sub> photon will reach the receiver OWLT later. In both cases, f<sub>0</sub> and df may change; but f<sub>t</sub> is always continuous, and changes in the coefficients occur only on integer seconds.</i> <i>For ODF measurements:</i>		

	$f_t = f_0$		
	<i>because df=0.</i>		
7	Transmitted frequency corresponding to time in column 6 <u>Two-way coherent modes:</u> Uplink frequency of ground station S-band order of 2100 MHz X-band order of 7100 MHz <u>One-way mode:</u> S/C transmission frequency X-band order of 8400 MHz S-band order of 2300 MHz	Hertz	$10^{-6}$ Hz
8	Uplink frequency ramp rate <u>DSN two-way coherent:</u> Time derivative of uplink frequency in column 7 <u>DSN one-way downlink mode:</u> Value of spacecraft frequency drift, if known and/or meaningful; -99999.999999 <u>ODF measurements:</u> Ramp rate is always zero; df=0	Hertz/sec	$10^{-6}$ Hz/sec
9	Observed S-band antenna frequency Frequency of the signal at the terminals of the receiving antenna structure at UTC TIME columns 2 to 4 ( $t_r$ ). Set to -9999999999.999999 for missing or corrupted data.	Hertz	$10^{-6}$ Hz
10	Predicted S-band antenna frequency Based on the ESOC reconstructed orbit file or SPICE kernels Expected frequency of the signal at the terminals of the receiving antenna structure at UTC TIME in columns 2 to 4 ( $t_r$ ). The calculation includes geometrical effects (relative positions and motions of ground station and spacecraft, including Earth rotation and light time adjustments), tuning of both the transmitter and receiver and a model-based correction for one- or two-way (as appropriate) propagation through the Earth's atmosphere.	Hertz	$10^{-6}$ Hz
11	Correction of Earth atmosphere propagation Correction term for the propagation of the signal in the Earth atmosphere and ionosphere, based on meteorological data observed at the ground station site (MET-files)	Hertz	$10^{-6}$ Hz

12	Residual calibrated X-band frequency shift <i>column 9 minus 10</i>	Hertz	10 <sup>-6</sup> Hz
13	Received S-band signal level <u>Closed-loop data:</u> <i>Signal level from AGC in decibels relative to one milliwatt (dBm).</i> <u>Open-loop (RSR):</u> <i>Signal level in decibels (dB) relative to an arbitrary reference.</i>	dBm / dB	0.1 dB
14	Differential Doppler $f_s - \frac{3}{11} f_x$ <i>Where <math>f_s</math> and <math>f_x</math> are the received S-band and X-band frequencies</i> <i>If BAND_NAME = X (from the label file), <math>f_x</math> comes from column 9 in this table and <math>f_s</math> comes from column 9 in the file identified by SOURCE_ID (from the label file).</i>  <i>If BAND_NAME = S (from the label file), <math>f_s</math> comes from column 9 in this table and <math>f_x</math> comes from column 9 in the file identified by SOURCE_ID (from the label file).</i> <i>if either band is not available, this column is set "-99999.999"</i>	Hertz	10 <sup>-6</sup> Hz
15	standard deviation of the observed antenna frequency S-band in column 9 (open-loop only) <i>for closed-loop this value is set "-99999.999"</i>	Hertz	10 <sup>-6</sup> Hz
16	Received S-band signal quality (open-loop only) <i>Ratio of observed received signal strength to the statistical standard deviation of the measurement, column 15 divided by column 19</i> <i>For closed-loop this is value is set "-999.9"</i>	dB	0.1 dB
17	standard deviation of received signal level at S-band (open-loop) <i>A statistical measure of the error in determining SIGNAL LEVEL (column 15) based on fit of a data spectrum to a sinc function. Uses the same arbitrary scale factor as column 15; units of dB.</i> <i>for closed-loop this is set "-999.9"</i>	dB	0.1 dB

## 4.3 ADDITIONAL OUTPUT FILES

### 4.3.1.1 Log file

The Module M\_OUTPUT generates an additional output file a so called log file. This file contains the processing mode, the whole path of all input files, additional information like downlink and uplink frequency in Hz, the sample rate in samples per seconds, statistical data about the processed data like average value and standard deviation, version of the processing software and error messages.

The log file will not be distributed and is only intended for internal use. Therefore the filename of the log file is not complying with [1]. But in order to relate the log file with the corresponding data files the log file gets the file name of the corresponding DOPPLER\_OUTPUT\_X\_file but instead of the ending .tab the ending .log is used. If a log file is already existing in the processing folder and the date are not automatically processed the log file gets the file name of the corresponding DOPPLER\_OUTPUT\_S\_file with ending .log. An example of a log file is shown in

[Figure 4-1](#).

Gelöscht: Figure 4-1

**ODF-DOP-SPEC-4310:** The average values of the residuals of S-Band data and X-Band data are computed only for the first 40% of the data. The computation is done via the following formulation

$$\bar{f}_{res} = \frac{1}{N} \sum_{i=1}^N f_{res,i}$$

**ODF-DOP-SPEC-4320:** The standard deviation of the residuals of S-Band data and X-Band data are computed only for the first 40% of the data. The computation is done via the following formulation

$$f_{res,std} = \sqrt{\frac{1}{N} \sum_{i=1}^N (f_{res,i} - \bar{f}_{res})^2}$$

```
MEX

OC

OCCULTATION

FLAGS FROM PROCESS_OPTIONS FILE:
-----
T      Processing with Predict
F      Processing with AGC
063    Groundstation
2004-08-29T13:58:00.000      Start time
2004-08-29T17:56:00.000      Stop time

NUMBER OF INPUT FILES:
-----
01     Number of doppler S-band files
01     Number of doppler X-band files
01     Number of Meteo files
00     Number of AGC S-Band files
00     Number of AGC X-Band files

FILES USED FOR PROCESSING:
-----
Z:\ddswork\DSN_data\odf_processed\MEX\2004\242\L1B\M00ODF0L1B_DPS_042421
400_02.TAB
Z:\ddswork\DSN_data\odf_processed\MEX\2004\242\L1B\M00ODF0L1B_DPX_042421
400_01.TAB
Z:\ddswork\DSN_data\odf_processed\WEA\M60ODF0L1B_MET_040010000_00.TAB
Z:\ddswork\DSN_data\odf_processed\MEX\2004\242\L1B\M00ODF0L1B_RMP_042421
400_02.TAB
Z:\unibw\Predicts\MarsExpress\2004\Predicts_GS63_SCO_242\M63UNBWL02_PTW_
042421335_00.TAB
```

```

FILES CREATED DURING PROCESSING:
-----
Z:/Processed_temp/MEX/Orbit/2005/DOY_002_1_MEX/NN13_NN11/D1/M32ICL3L02
_D1S_050020542_00.TAB
Z:/Processed_temp/MEX/Orbit/2005/DOY_002_1_MEX/NN13_NN11/D1/M32ICL1L02
_D1X_050020542_00.TAB
Z:/Processed_temp/MEX/Orbit/2005/DOY_002_1_MEX/NN13_NN11/D1/M32ICL3L02
_D1S_050020542_00.LBL
Z:/Processed_temp/MEX/Orbit/2005/DOY_002_1_MEX/NN13_NN11/D1/M32ICL1L02
_D1X_050020542_00.LBL

CONFIGURATION INFO:
-----
UPLINK-FREQUENCY X-BAND: 7166619369.9976720809936523
DOWNLINK-FREQUENCY X-BAND: 8420060140.9852495193481445
SAMPLE-INTERVAL X-BAND: 1.0000000000000000
TRANSPONDER-RATIO X-BAND:880/749
UPLINK-FREQUENCY S-BAND: 7166619369.9976720809936523
DOWNLINK-FREQUENCY S-BAND: 2296380038.4505224227905273
SAMPLE-INTERVAL S-BAND: 1.0000000000000000
TRANSPONDER-RATIO S-BAND:240/749

PROCESSING INFO
-----
AVERAGE S-BAND RESIDUALS IN mHZ: -6.94218
STANDARD DEVIATION S-BAND RESIDUALS IN mHZ: 4.39143
AVERAGE X-BAND RESIDUALS IN mHZ: 9.68471
STANDARD DEVIATION X-BAND RESIDUALS IN mHZ: 14.90616
PLASMA-CORRECTION DONE WITH DIFFERENTIAL DOPPLER
FILES OVERLAPPING IN TIME
X-BAND-MODE: TWO-WAY
S-BAND-MODE: TWO-WAY

SOFTWARE INFO:
-----
SOFTWARE NAME: DSN_ODF_PROC_DOP_L1A_TO_L2_V1.2
CREATION TIME: 2005-11-07T16:24:09.000
PROCESSED BY: andert
ERRORS:
-----

```

Figure 4-1: Example of a log file



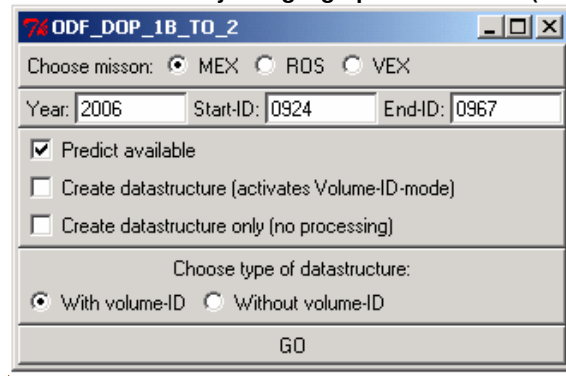


## 5 USAGE OF THE SOFTWARE

The above described software is embedded in a PERL script that calls the Fortran software.

One ODF data file of level 1b can contain more than one operation i.e. data from more than one ground station and over a long time period. Therefore before processing the level 1b data file start time, stop time and the respective ground station for each operation have to be selected.

This can be done by using a graphical interface (see



Formatiert: Zentriert

Formatiert: Schriftart: 12 pt, Nicht Fett, Englisch (USA)

Figure 5-4, and section Fehler! Verweisquelle konnte nicht gefunden werden, for a detailed description).

Gelöscht: Figure 5-3

Gelöscht: 5.1.1

The processing options like mission, observation type or availability of files can be adjusted by means of another graphical interface (see and section 0, for a detailed description).

Gelöscht: 5.1.2

If all this information are filled in the graphical interfaces and the software starts to work the Perl script creates a file named odf\_process\_options.txt and writes all necessary information into this file. This odf\_processing\_options.txt file is read in from the Fortran software and serves as a data interface between the graphical interfaces and the Fortran software.

Therefore the Fortran software can be used by editing the odf\_process\_options.txt and filling the necessary information by hand in. An example of a odf\_process\_options.txt file is shown in Figure 5-1.

Gelöscht: Figure 5-1

Figure 5-1: Example of a odf\_process\_options.txt file

```

MEX      ! Mission
NO       ! Autoprocessing
GR       ! Processing Mode
YES      ! Processing with predict
YES      ! Processing with agc
1        ! Number of X-band files
1        ! Number of S-band files
1        ! Number of meteo files
1        ! Number of AGC X-band files
1        ! Number of AGC S-band files
SOLAR    ! Observation Type
Andert   ! login
1        ! Number of operations
63       ! Station ID
2004-242::14:45:00 ! Start time of the operation
2004-242::16:00:00 ! Stop time of the operation
    
```

Usually the Fortran software finds the files to process automatically based on the information in the `odf_process_options.txt` file but if the files can not be found the files must be selected by hand. The selection of the respective files for processing is done via another graphical interface shown below in [Figure 5-2](#). The simultaneously arising DOS window (see [Figure 5-3](#)) indicates what kind of file is needed and shows subsequently the processing status.

Gelöscht: Figure 5-2  
 Gelöscht: Figure 5-3

In addition there is a possibility provided to process an amount of data automatically. But for this a log file (see section 4.3.1.1) must exist, i.e. the files have to be processed one time before by hand.

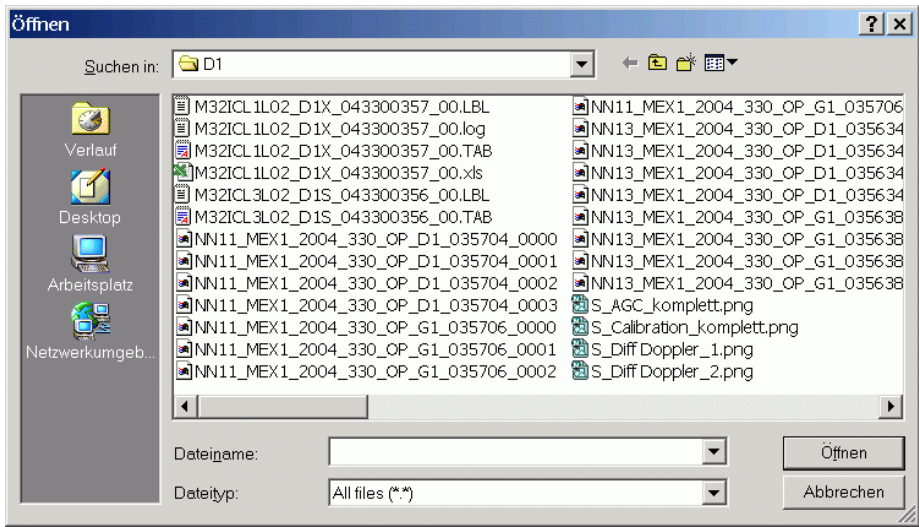


Figure 5-2: Graphical interface for selection of the input files.

```

Auswählen D:\Coding\Perl\586\bin\perl.exe
STARTS TO WORK
$-BAND-FILE
X-BAND-FILE
RAMP-FILE
METEO-FILE
PREDICT-FILE
X-BAND DOPPLER FILE READ IN
X-BAND DOPPLER FILE READ IN
RAMP FILE READ IN
METEO FILE READ IN
TROPOSPHERIC CALIBRATION DONE
DIFFERENTIAL DOPPLER DONE
PLASMA CALIBRATION DONE
PREDICT DONE
WRITE DATA DONE
Yes, ODF_DOP_1B_TO_2 did work.
-----
    
```

Figure 5-3: DOS window showing information about the kind of file needed to be inputted and processing status.

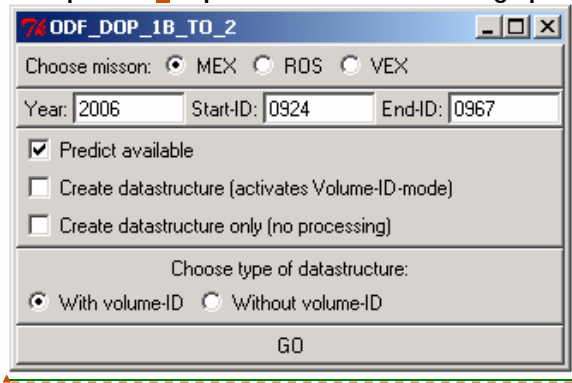
## 5.1 GRAPHICAL INTERFACE

In order to process the data some information are needed. For this job an graphical user interface is developed and will be described in detail below.

- Gelöscht: several
- Gelöscht: s
- Gelöscht: are

### 5.1.1 Selection of operations

The selection of the operations s to process can be done via a graphical interface (see



- Formatiert: Zentriert
- Formatiert: Schriftart: 12 pt, Nicht Fett, Englisch (USA)

Figure 5-4). The mission and the year have to be filled in the graphical interface. Pressing return starts to search for all DSN operations between the selected volume-IDs in the current version of the logbook. There are several options to be selected. Processing with or without a predict file is possible. Creating of the data structure is

- Gelöscht: Figure 5-1
- Gelöscht: day of year
- Gelöscht: of the
- Gelöscht: day of year

possible to be switched off. Also its possible to let the script create these data structures without L2-processing. Two types of data structures are available. Pressing the GO-button starts processing. All information necessary for processing are read from the current version of the logbook by the PERL-script.

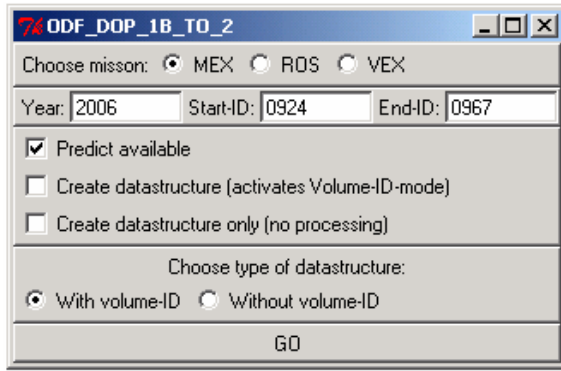


Figure 5-4: Selection of operations to be processed

## 5.2 ADDITIONAL OUTPUT FILES

Several files with additional information about the processed data are produced during the processing operation by means of a PERL script which is called by the main script.

### 5.2.1 Data validation Excel sheet

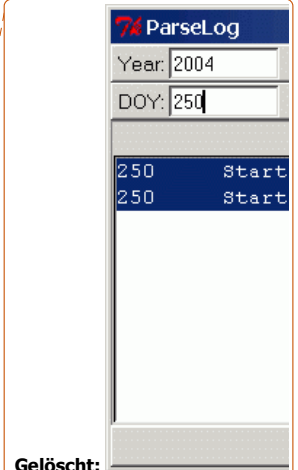
An Excel sheet is automatically generated during the processing operation. The information contained in the Excel sheet is read in from the above described log file. This Excel sheet is used for data validation aims and is complemented with additional information during data validation and can accordingly be copied into a log book comprising processing information about all level 2 data.

The excel sheet will not be distributed and is only intended for internal use. Therefore the filename of the Excel sheet is not complying with [1]. But in order to relate the Excel sheet with the corresponding data files the Excel sheet gets the file name of the corresponding DOPPLER\_OUTPUT\_X\_file but instead of the ending .tab the ending .xls is used.

### 5.2.2 Data illustration

During the processing operation a number of plots illustrating the processed data are automatically generated.

**Gelöscht:** and shows the results in the lower area of the graphical interface. The operations can be selected by clicking on it. There can be selected more than one operation. By pressing the go button a new graphical interface will appear. ¶



**Gelöscht:**

**Formatiert:** Schriftart: 12 pt, Nicht Fett, Englisch (USA)

**Formatiert:** Zentriert

**Gelöscht:** 5

**Gelöscht:** start time, stop time and ground station for each

**Gelöscht:** ¶  
 <#>Selection of processing options¶

¶  
 In the new graphical interface (see Figure 5-5) the before selected operations are filled in and the remaining options for processing can be completed there. The graphical interface is divided in several adjustment parts for processing and the constituent parts will be described in the following. ¶  
 ¶ ... [1]

**Formatiert:** Englisch (Großbritannien)

**Formatiert:** Englisch (Großbritannien)

**Gelöscht:** 5. Additional processing information¶

- ¶  
 <#>Autoprocessing: The selected operation will be processed automatically u ... [2]

- **Correction of the earth atmosphere propagation** (column 11 in the data file of level 2) in Hz is plotted over the entire time period. This is done for S-band and X-band Doppler data.
- **Residual calibrated data** (column 12 in the data file of level 2) in Hz is plotted over the entire time period for S-band and X-Band Doppler data. In addition partial plots are generated. If the total number of sample points is bigger than 3600 the data to illustrate is divided into subintervals with 3600 data points or less for the remaining data points and plotted. If the total number of sample points is smaller than 3600 the data to illustrate is divided into subintervals with 600 data points or less for the remaining data points and plotted.
- **Received signal level** (column 13 in the data file of level 2) in dBm is plotted over the entire time period for S-band and X-Band Doppler data.
- **Differential Doppler** (column 14 in the data file of level 2) is plotted over the entire time period for S-band and X-Band Doppler data if it is available. In addition partial plots are generated. If the total number of sample points is bigger than 3600 the data to illustrate is divided into subintervals with 3600 data points or less for the remaining data points and plotted. If the total number of sample points is smaller than 3600 the data to illustrate is divided into subintervals with 600 data points or less for the remaining data points and plotted.

### 5.3 ERROR MESSAGES

The following describes some errors that maybe occur during the processing operation.

- A kernel defined in the list of loaded kernels is not available in the folder where all kernels for processing are stored. Therefore the missing kernel has to be copied into the kernel folder. An example of the error message is shown in [Figure 5-5](#).

Gelöscht: Figure 5-6

```

D:\Coding\Perl\586\bin\perl.exe
Start:365 Ende:365

DOY_365_MEX
find_log done!!
STARTS TO WORK

=====

Toolkit version: N0053

SPICE(NOSUCHFILE) --

The tenth file
'Z:\ddswork\process_data\Soft_Doppler_L2\Kernels\ORMM__041201000000_00105.BSP'
specified by KERNELS_TO_LOAD in the file
Z:\ddswork\process_data\Soft_Doppler_L2\Kernels\List_of_loaded_kernels.txt
could not be located.

A traceback follows. The name of the highest level module is first.
FURNISH --> ZZLDKER

Oh, by the way: The SPICELIB error handling actions are USER-TAILORABLE. You
can choose whether the Toolkit aborts or continues when errors occur, which
error messages to output, and where to send the output. Please read the ERROR
"Required Reading" file, or see the routines ERRACT, ERRDEU, and ERRPRT.

=====
Return code 1
Sorry, an error occurred! No output produced!
-----
    
```

Figure 5-5: Example of an error message if a kernel file is missing.

Gelöscht: 6

- A wrong kernel file containing ephemeris data with inappropriate time stamps is loaded. This has to be corrected in the module M\_SPICE. An example of the error message is shown in [Figure 5-6](#).

Gelöscht: Figure 5-7

```

D:\Coding\Perl\586\bin\perl.exe
DOY_365_MEX
find_log done!!
STARTS TO WORK

=====

Toolkit version: N0053

SPICE(SPKINSUFFDATA) --

Insufficient ephemeris data has been loaded to compute the state of -41 (MARS EXPRESS) relative to 0 (SOLAR SYSTEM BARYCENTER) at the ephemeris epoch 2004 DEC 30 05:34:15.683.

A traceback follows. The name of the highest level module is first.
SPKEZR --> SPKEZ --> SPKAPP --> SPKSSB --> SPKGEO

Oh, by the way: The SPICELIB error handling actions are USER-TAILORABLE. You can choose whether the Toolkit aborts or continues when errors occur, which error messages to output, and where to send the output. Please read the ERROR "Required Reading" file, or see the routines ERRACT, ERRDEU, and ERRPRT.

=====
Return code 1
Sorry, an error occurred! No output produced!
    
```

Figure 5-6: Example of an error message if a wrong kernel file is loaded.

Gelöscht: 7

- Two or more identical lines in the data file are existing and therefore the interpolation routine is not working. Consequently the redundant information has to be erased. This can happen in the meteo file of level 1a and the predict file. An example of the error message is shown in [Figure 5-7](#). If the meteo file contains redundant data the terminal error arises after READ DOPPLER DONE.

Gelöscht: Figure 5-8



Rosetta Radio Science Investigations RSI

Mars Express Orbiter Radio Science Experiment MaRS

Venus Express Radio Science Experiment VeRa

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[VEX-VRA-IGM-DS-5009](#)

```
D:\Coding\Perl\586\bin\perl.exe
Start:365 Ende:365
DOY_365_MEX
find_log done!!
STARTS TO WORK
READ DOPPLER DONE
READ METEO DONE
TROPO CALIBRATION DONE
DIFFERENTIAL DOPPLER DONE
PLASMA CALIBRATION DONE

*** TERMINAL ERROR 2 from DC1SOR. Points in the data point abscissas array,
*** XDATA, must be distinct, but XDATA(2) = XDATA(3) =
*** 1.576555624123737D+08.
Here is a traceback of subprogram calls in reverse order:
Routine name      Error type  Error code
-----
DC1SOR            5           2      (Called internally)
DC2DEC            0           0      (Called internally)
DC2INT            0           0      (Called internally)
DC2IEZ            0           0      (Called internally)
DCSIEZ            0           0
USER              0           0
Kein Logfile
```

Figure 5-7: Example of an error message if redundant data is contained in the predict file.

Gelöscht: 8



## APPENDIX A

### Computation of the sky frequency received at ground station from doppler predicts

#### Acronyms:

$f_{gs}$  = frequency emitted from ground station

$f_{sc}$  = frequency emitted from spacecraft

$f_{r_{sc}}$  = frequency received at spacecraft

$f_{r_{gs}}$  = frequency received at ground station

$\Delta f_{sc}$  = frequency shift received at spacecraft in the uplink signal emitted from groundstation

$\Delta f_{gs}$  = frequency shift received at groundstation in the downlink signal emitted from the spacecraft

K = transponder conversion ratio

$P_{UL}$  = doppler predict of the uplink signal independent from frequency

$P_{DL}$  = doppler predict of the downlink signal independent from frequency

#### General relations:

$$P_{UL} = \frac{\Delta f_{sc}}{f_{gs}}$$

$$P_{DL} = \frac{\Delta f_{gs}}{f_{sc}}$$

One-way case

$$\Delta f_{gs} = f_{s_{sc}} \cdot P_{DL}$$

it is needed

$$f_{r_{gs}} = \Delta f_{gs} + f_{s_{sc}}$$

therefore the sky frequency is

$$f_{r_{gs}} = f_{s_{sc}} \cdot P_{DL} + f_{s_{sc}}$$

or

$$f_{r_{gs}} = f_{s_{sc}} \cdot (P_{DL} + 1)$$

**Rosetta Radio Science Investigations RSI****Mars Express Orbiter Radio Science Experiment MaRS****Venus Express Radio Science Experiment VeRa****DSN ODF (Orbit Data File) Calibration Software : Level 1b to Level 2**

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[ROS-RSI-IGM-DS-3128](#)[VEX-VRA-IGM-DS-5009](#)Two-way case:

$$\Delta f_{sc} = f_{s_{gs}} \cdot P_{UL}$$

$$\Delta f_{gs} = f_{s_{sc}} \cdot P_{DL}$$

needed is

$$f_{r_{gs}} = \Delta f_{gs} + f_{s_{sc}}$$

therefore

$$f_{r_{gs}} = f_{s_{sc}} \cdot P_{DL} + f_{s_{sc}}$$

or

$$f_{r_{gs}} = f_{s_{sc}} \cdot (P_{DL} + 1)$$

with

$$f_{s_{sc}} = K \cdot f_{r_{sc}}$$

$$\Rightarrow f_{r_{gs}} = K \cdot f_{r_{sc}} \cdot (P_{DL} + 1)$$

$$\Rightarrow f_{r_{gs}} = K \cdot (f_{s_{gs}} + \Delta f_{sc}) \cdot (P_{DL} + 1)$$

$$\Rightarrow f_{r_{gs}} = K \cdot f_{s_{gs}} \left( 1 + \frac{\Delta f_{sc}}{f_{s_{gs}}} \right) \cdot (P_{DL} + 1)$$

$$\Rightarrow f_{r_{gs}} = K \cdot f_{s_{gs}} (1 + P_{UL}) \cdot (1 + P_{DL})$$

and therefore the sky frequency is

$$f_{r_{gs}} = K \cdot f_{s_{gs}} (1 + P_{UL} + P_{DL} + P_{UL} \cdot P_{DL})$$



## APPENDIX B

### Computation of the plasma correction using the differential doppler

The differential doppler is computed via

$$\delta f = f_{S,antenna}|_{tropo\_corrected} - \frac{3}{11} f_{X,antenna}|_{tropo\_corrected} \quad (1.12)$$

or

$$\delta f = -\frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \left\{ \frac{1}{f_S^2} - \frac{1}{f_X^2} \right\} f_S \frac{dl}{dt} \quad (1.13)$$

therefore the temporal change in electron content is

$$\frac{dl}{dt} = -\left\{ \frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \right\}^{-1} \frac{\delta f}{f_S} \left\{ \frac{1}{f_S^2} - \frac{1}{f_X^2} \right\}^{-1} \quad (1.14)$$

the plasma correction for S-Band is

$$f_{S,antenna,cal} = f_{S,antenna}|_{tropo\_corrected} + \frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \frac{1}{f_S} \frac{dl}{dt} \quad (1.15)$$

and for X-Band

$$f_{X,antenna,cal} = f_{X,antenna}|_{tropo\_corrected} + \frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \frac{1}{f_X} \frac{dl}{dt} \quad (1.16)$$

If equation (1.14) is inserted into (1.15)

Gelöscht: (1.14)

Gelöscht: (1.15)

$$f_{S,antenna,cal} = f_{S,antenna}|_{tropo\_corrected} + \frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \frac{1}{f_S} \left( -\frac{1}{2c} \frac{1}{4\pi^2} \frac{e^2}{m_e \epsilon_0} \right)^{-1} \frac{\delta f}{f_S} \left\{ \frac{1}{f_S^2} - \frac{1}{f_X^2} \right\}^{-1}$$

$$\Rightarrow f_{S,antenna,cal} = f_{S,antenna}|_{tropo\_corrected} - \delta f \left\{ \frac{f_S^2}{f_S^2} - \frac{f_S^2}{f_X^2} \right\}^{-1}$$

and with the general relations

$$f_X = \frac{11}{3} f_S \Leftrightarrow \frac{f_X}{f_S} = \frac{11}{3} \Leftrightarrow \frac{f_S}{f_X} = \frac{3}{11} \quad (1.17)$$

follows than

$$f_{S,antenna,cal} = f_{S,antenna}|_{tropo\_corrected} - \delta f \left\{ 1 - \frac{9}{121} \right\}^{-1}.$$

Therefore equation (1.15) can be written as

Gelöscht: (1.15)

$$f_{S,antenna,cal} = f_{S,antenna}|_{tropo\_corrected} - \delta f \frac{121}{112} \quad (1.18).$$

A similar computation can be done for equation (1.16).

Gelöscht: (1.16)

$$\Rightarrow f_{X,antenna,cal} = f_{X,antenna}|_{tropo\_corrected} - \delta f \left\{ \frac{f_X f_S}{f_S^2} - \frac{f_X f_S}{f_X^2} \right\}^{-1}$$

Using equation (1.17).

Gelöscht: (1.17)

$$\Rightarrow f_{X,antenna,cal} = f_{X,antenna}|_{tropo\_corrected} - \delta f \left\{ \frac{11}{3} - \frac{3}{11} \right\}^{-1},$$

therefore equation (1.16) can be written as

Gelöscht: (1.16)

$$f_{X,antenna,cal} = f_{X,antenna}|_{tropo\_corrected} - \delta f \frac{33}{112} \quad (1.19)$$



## **APPENDIX C**

Earth Klobuchar Ionosphere Model (see attached document CGIM\_ANNEX\_C.pdf)

Page left free

## **APPENDIX D**

Klobuchar File Format Description (see attached document CGIM\_ANNEX\_D.pdf)

## Selection of processing options

In the new graphical interface (see Figure 5-5) the before selected operations are filled in and the remaining options for processing can be completed there. The graphical interface is divided in several adjustment parts for processing and the constituent parts will be described in the following.

### 1. Mission

**MEX:** Mars-Express

**ROS:** Rosetta

**VEX:** Venus-Express

### 2. Observation type

**Commissioning:** Part of the mission where the retrieved data are only used for calibration aims.

**Occultation:** Occultation measurements are performed.

**Target Gravity:** A specified target is chosen for gravity measurements.

**Global Gravity:** Global measurements are performed.

**Phobos:** Gravity measurements at the Mars moon Phobos are performed (only for Mars-Express applicable).

**Solar Corona:** Measurements of the Solar Corona are performed.

### 3. Procession mode:

**Occultation:** Tropospheric calibration (see section 3.4.1) and plasma calibration via the klobuchar model (see section 3.4.4) is applied.

**Gravity:** Tropospheric and plasma calibration is applied. The plasma calibration is done via the differential Doppler (see section 3.4.2 and 3.4.3 for more details). If no differential Doppler is available plasma calibration via the klobuchar model (see section 3.4.4) is applied.

### 4. Operations:

Here are the start time, stop time and the ground station identifier automatically filled in, but can be adjusted by hand.

## 5. Additional processing information

**Autoprocessing:** The selected operation will be processed automatically using the log file (see section 4.3.1.1)

**One-Way:** The selected operations will be processed as one-way measurements.

## 6. Availability of input files

This part defines the availability of files to process. Both X-band and S-band files and the meteo file are required for processing. The processing can be done without a predict file and AGC file but not all columns of the output file will get a valid value. The Klobuchar file is only required for plasma correction (see section 3.4.4 for more details) for Occultation and for Gravity measurements if no differential Doppler is available.

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Your login: andert

Please choose Mission:  MEX  ROS  VEX

Select Observation Type:

COMMISSIONING

OCCULTATION

TARGET GRAVITY

GLOBAL GRAVITY

PHOBOS

SOLAR CORONA

Select Processing Mode:

OCCULTATION

GRAVITY

Specify operations details:

#OPS	DSN	START TIME	STOP TIME
1	63	2004-250::15:26:00.000	2004-250::17:50:00.000

Autoprocessing: ON

One-Way Operation: YES

DPX File available 1

DPS File available 1

AGX File available 0

AGS File available 0

MET File available 1

UBW File available

KLO File available

Ramp File available

Quit

Process ODF DOPPLER

Figure 5-5: Graphical interface for ODF Doppler data processing

