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#### 1. Introduction

During the SPIRE SPT, a series of CS tests will be carried out to investigate a CS susceptibility seen during the PFM-5 ILT test campaign on some of the detectors on the PMW detector array.

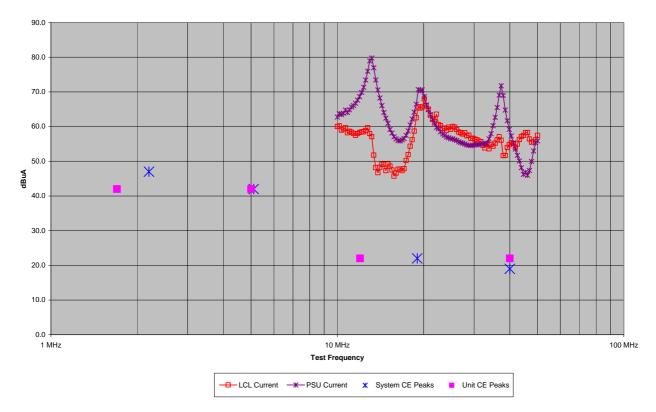
This note outlines the rationale adopted in RD\_1 for the selection of the test levels to inject during the test.

#### 2. Reference Documents

RD	Name	Doc Number
RD 1	SPIRE System Level CS Test Procedure	SPIRE-RAL-REP-003040
RD 2	Herschel S/C Level Conducted EMC Test Procedure	HP-2-ASED-TP-0155 As Run Final
RD 3	HSPSU EMC Test Report	HSPIR-PSU-REE-DA0018814-V-ASTR
RD 4	SPIRE SPT EMC CS Pre-calibration test	SPIRE-RAL-REP-003039

### 3. Results Synthesis

Figure 1 shows the results for the differential mode conducted emissions tests on the PFM PSU prior to delivery to the SPIRE project (from RD\_3), the levels measured during the system level CE test (form RD\_2) and the levels measured during a pre-calibration test (RD\_4).



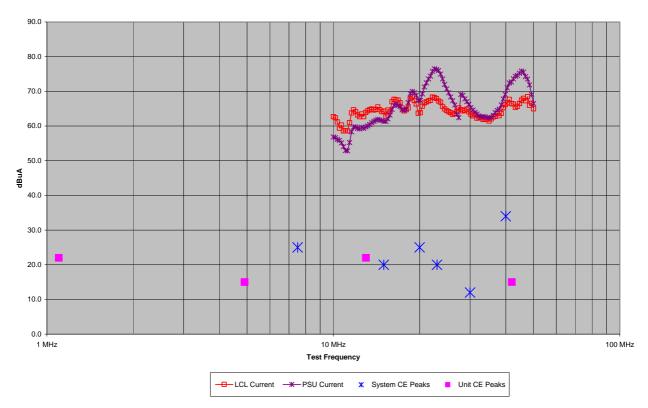
**Figure 1** – DM Current test results summary. The "LCL Current" and PSU Current" traces refer to measurements carried out during a pre-calibration test (RD\_4) where EMI was injected onto a representative primary power harness.

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Figure 2 shows the results for the common mode conducted emissions tests on the PFM PSU prior to delivery to the SPIRE project (from RD\_3), the levels measured during the system level CE test (form RD\_2) and the levels measured during a pre-calibration test (RD\_4).

In both Figure 1 and Figure 2 the points reporting the unit and spacecraft CE results represent the approximate peaks in the emissions as measured from the experimental plots.



**Figure 2** – CM Current test results summary. The "LCL Current" and PSU Current" traces refer to measurements carried out during a pre-calibration test (RD\_4) where EMI was injected onto a representative primary power harness.

#### 4. Discussion

The levels injected during the test described in RD\_4 and plotted on Figure 1 and Figure 2 was -6dB from the IID-A levels. It can be seen that there is generally more than 20dB of margin between the levels injected during the pre-calibration tests and the levels seen during unit and system level CE tests. Lowering the levels a further 6dB below the IID-A levels to -12dB from the IID-A levels will demonstrate margin with respect to the levels seen on the spacecraft bus and provide a good safety margin for the flight hardware under test.

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### 5. Appendix - Spacecraft CE Test results (SPIRE summary)

#### 5.1 CM Reference

 $PLOT\ NO: 01\underline{6};\ PROJECT:\ Herschel\ Co\underline{n}ducted;\ OP\ MODE:ref;\ COMMENT:\ Ambient,\ \ DATE:\ 07/11/30$ 

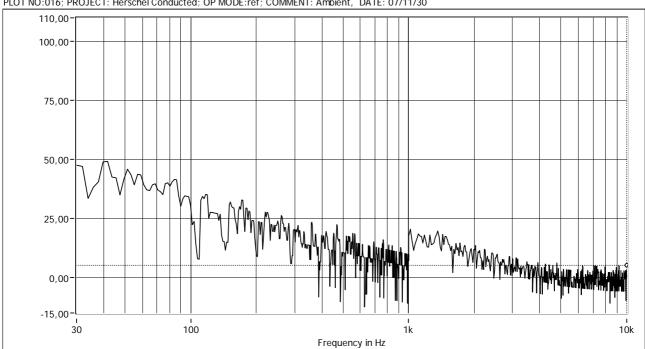


Figure 3 - CM reference (30Hz-10kHz)

PLOT NO: 016; PROJECT: Herschel Conducted; OP MODE: ref; COMMENT: Ambient; DATE: 07/11/30

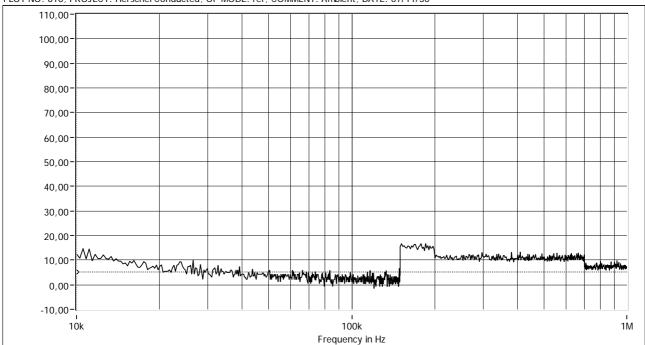


Figure 4 - CM reference (10kHz-1MHz)

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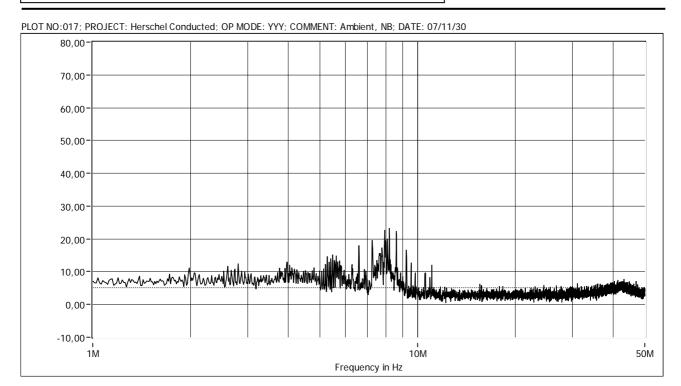


Figure 5- CM reference (1MHz-50MHz)

#### 5.2 DM Reference

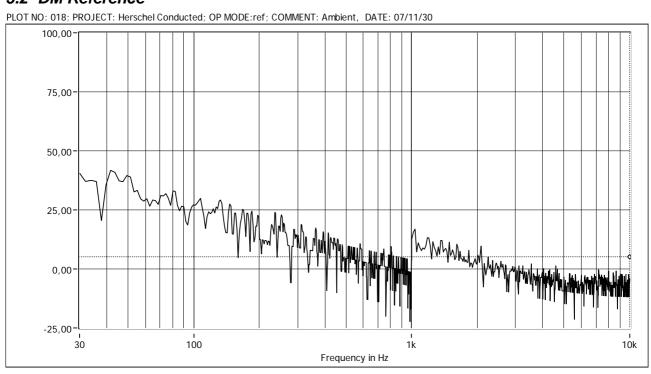


Figure 6 - DM reference (30Hz-10kHz)

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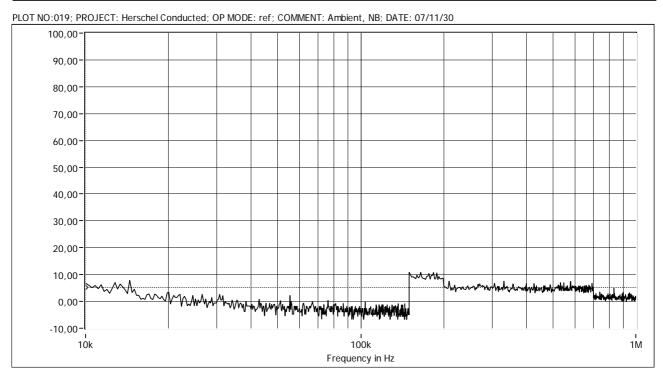


Figure 7 - DM reference (10kHz-1MHz)

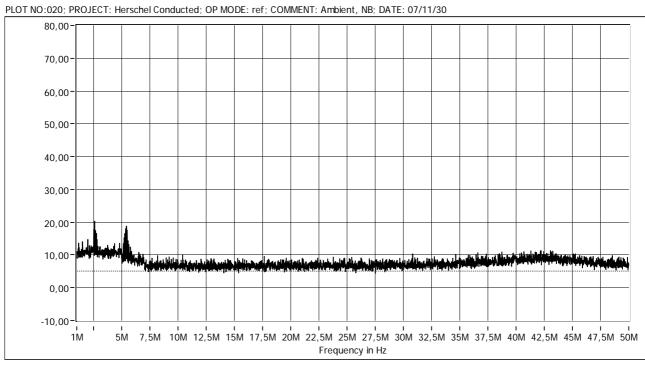


Figure 8- DM reference (1MHz-50MHz)

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#### 5.3 CM Test

PLOT NO: 57; PROJECT: Herschel Conducted; OP MODE: noisiest; COMMENT: DUT,cm; DATE: 07/12/03

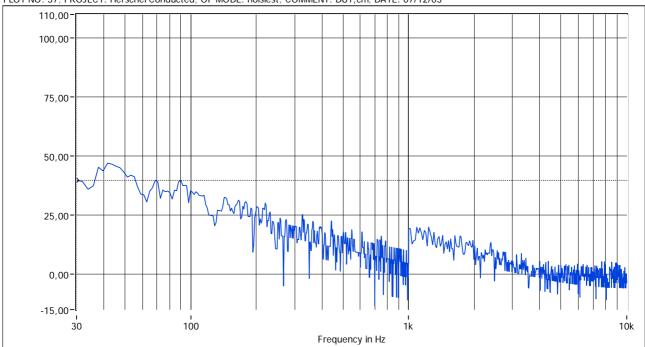


Figure 9 - CM Test (30Hz-10kHz)

PLOT NO: 58; PROJECT: Herschel Conducted; OP MODE: noisiest; COMMENT: DUT, CM; DATE: 07/12/03

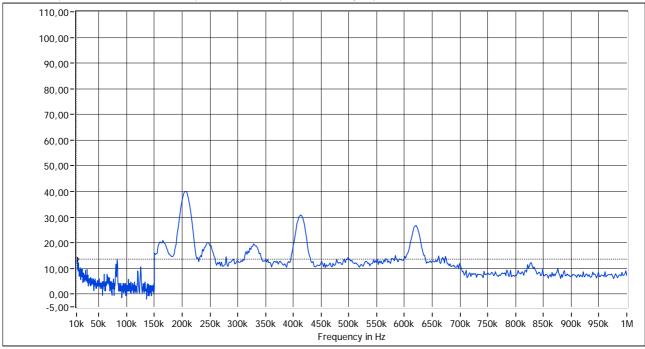


Figure 10 - CM Test (10kHz-1MHz)

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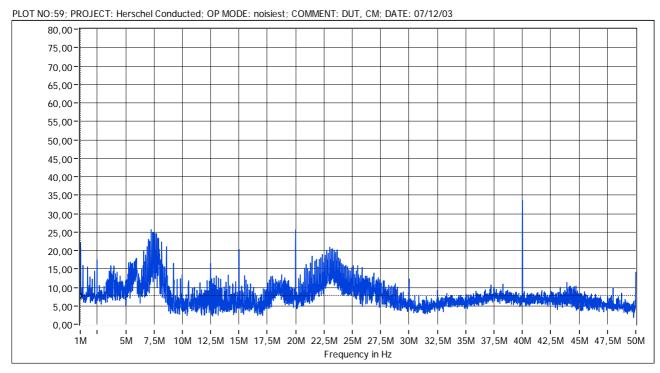


Figure 11 - CM Test (1MHz - 50MHz)

#### 5.4 DM Test

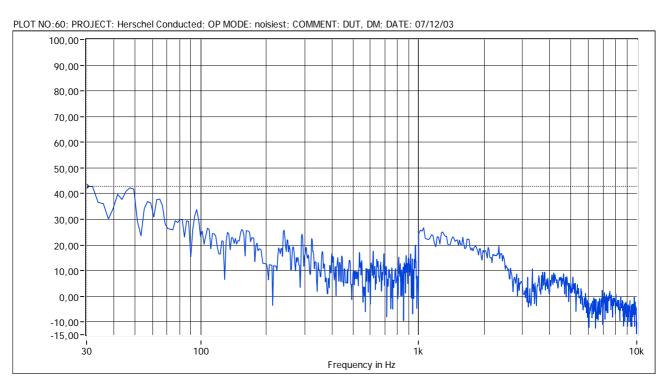


Figure 12 – DM Test (30Hz-10kHz)

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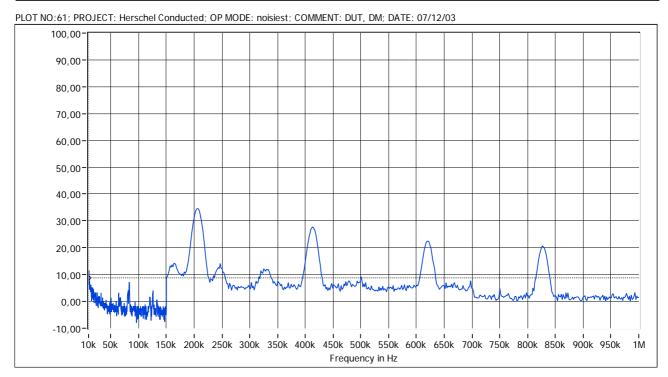


Figure 13 - DM Test (10 kHz - 1 MHz)

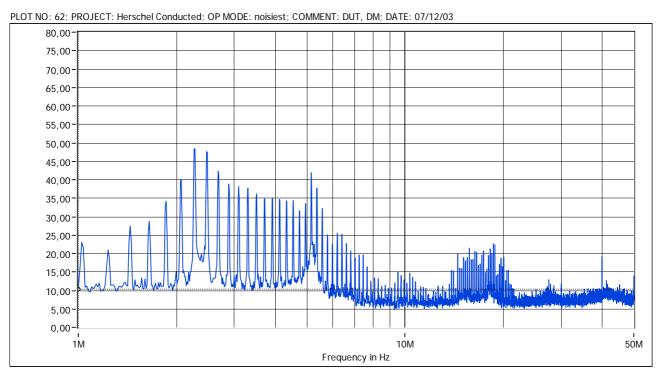


Figure 14 - DM Test (1MHz-50MHz)