SPIRE-ALC-MOM-00	2464
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ALC AT	E L HER	SCHEL/PLA	NCK			
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COMPTE RE	NDU DE REUI		3	LIEU / PLAC	CE : Telecon S	SPIRE/ESA/ASP
OBJET / PURPOSE : CLASSIFICATION : SPIRE Line of sight estimation accuracy telecon						
PARTICIPANTS ATTENDEES	Societe <i>FIRM</i>	SIGNATURE SIGNATURE	PARI AT	TICIPANTS TENDEES	societe <i>FIRM</i>	SIGNATURE SIGNATURE
Bruce SWINYA	RD SPIRE		D. G	UICHON	ASP	
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Eric SAWYEF	R SPIRE					
REDACTEUR / WRITTER BY :	N DG					
CONCLUSION :		Attendants				
DISTRIBUTION : PARTICIPANTS /	Pour Action : For further Action	Attendants				
ATTENDEES	B. Collaudin, G. Boubrovik, P. Rideau, Y. Roche, P. Couzin POUR INFORMATION : FOR INFORMATION					
APPROUVE PAR / APPROVED BY						
NOM / NAME						
Signature / <i>Signature</i>						

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ALCATEL HERSCHEL/PLANCK			
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COMPTE RENDU DE REUN	LIEU / PLACE : Telecon SP	IRE/ESA/ASP	
SUITE / CONTINUED :		ACTION	
1. Introduction Agenda has been distributed by B. Collaudin email	« instrument Line of		
 sight estimation accuracy » dated 30-06-05. It is : 1. Confirmation of the compliance to IID-A requies a goal : identification of conditions to achieve 0 applicability to all detector LOS verification method Available sources for calibration (results of calibration necessary to reach the mode where achieved AOB As an input for the telecon, Matt Griffin has "Astrometric accuracy achievable with SPIRE" dupdate of SPIRE-UCF-NOT-001818 (issue 2) ([R1]) UCF-NOT-002315 issue 1 ([R2]). 	irement 0.6 arcsec libration sterring ere best accuracy is distributed by mail lated 04-07-2005 an together with SPIRE-		
2. Confirmation of the compliance to IID-A requirer	nent		
- 1 arcsec maximum			
Alcaterrecali the context of the IID-A requirement.			
In order to achieve the best performances, the long the instrument reference and the scientific mode at be calibrated. The calibration phase aims mainly at bias errors. This system calibration is mandatory mainly becau will causes relative misalignment between instrumen Sight. So an extensive initial calibration (also called be necessary during the performance verification p check (also called calibration check) and cal update will be performed.	g term error between ttitude sensors has to the reduction of the use of launch which nt and ACMS Line Of main calibration) will ohase. Then periodic alibration parameter		
The calibration will be performed such as to point both the instrument and the ACMS sensors (Star Tracker) toward targets with well known directions (reference sources). The resulting ACMS attitude estimation will be compared with payload attitude data resulting in an estimation of the bias between instrument and ACMS sensors line of sight. The result of the calibration process is then used for commanding the pointing of the spacecraft (bias compensation).			

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Th re of so re	e bias will not be estimated perfectly but a calibration residual will main. This residual depends of the ACMS attitude estimation and also the instrument attitude estimation accuracy using the reference urce at the time of the calibration. This is the origin of IID-A quirement.	
<u>Di</u> : 00 ₽	<u>scussion about of SPIRE-UCF-NOT-001818 issue 2 and SPIRE-UCF-NOT-2315 issue 1:</u> Section 1 of [R1] states that the beam steering mirror which is used to create the 7-point jiggle map. SPIRE confirm that the spacecraft attitude target will remain the same during this operation (no nodding is today foreseen). The typical duration of observation will be few minutes.	
⇔	As the satellite is not moving, the SRPE contribution in section 3.4 of [R1] can be deleted.	
⊳	According to conclusion of [R2], the optimum grid spacing is 0.3 beam. ASP asks if equivalent of figure 10 and 11 are available for 0.3 beam. This should quantify $\delta \theta_{stat}$ error.	AI#1 SPIRE <mark>TBD</mark> -05
SP	IRE will assess the possibility to run this simulation.	
⊳	SPIRE indicates that there is no need to use the peak up mode for achieving the required accuracy.	
₽	The simulations reported in [R2] are considering a constant telescope error (called APE in R2). The applied error is constant all over the simulation. This doesn't simulate the real behaviour of the spacecraft.	AI#2 ASP 15-JULY -05
Al	catel will detail the frequency content of APE.	AI#3 SPIRE <mark>TBD</mark> -05
SP	IRE will update the [R2] simulations based the Alcatel input	
⇔ SP ar δθ ac	Frequency class of the error : IRE confirm that $\delta \theta_{stat}$ and $\Delta \theta_{RPE}$ are random contribution and $\delta \theta_{pos}$ and $\delta \theta_{SRPE}$ are bias contributions. pos could be very small depending of the source. SPIRE will update [R1] adding references where $\delta \theta_{pos}$ is quantified.	AI#4 SPIRE <mark>TBD</mark> -05

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In synthesis, SPIRE agree that 1 arcsec is acheivable. Better performance will be difficult to achieve.				
 goal : identification of conditions to achieve 0.6 arcsec The highest quality of measurement of the position is 0.5 arcsec (see section 3.5 of [R1]). This is in line with gaol requirement. The associated conditions are among other things : ⇒ Negligible telescope beam asymmetry ⇒ Negligible RPE 				
SPIRE to identify in detail the conditions (instrument external and internal conditions) for achieving 0.5 arcsec accuracy. SPIRE will elaborate a budget to analyse different contributor weight and to be able to identify different conditions to achieve the goal.	AI#5 SPIRE <mark>TBD</mark> -05			
- applicability to all detector LOS The note mentions «the boresight of the SPIRE instrument ». There are several detector LOSs in the SPIRE instrument. The alignment between the different line of sight will be measured on ground (including before and after vibration testing). Even if the launch effect is expected to be small, The 3 detector lines of sight will be estimated in flight using the same philosophy.				
Nevertheless ,the accuracy will be different (typically 1", 1.5" and 2). SPIRE will update [R1] note including other detector accuracy.	AI#6 SPIRE <mark>IBD</mark> -05			
- verification method				
In addition to the simulation campaign, SPIRE Line of sight estimation accuracy will be verified by test during Optical alignment test.				
3. Available sources for calibration (results of calibration sterring group -				
Al of pointing meeting in Estec (4 dec 2003))				
Section 2 of [R1] provides a list of suitable objects for system calibration. Alcatel asks if a catalog of these objects is existing (ex : LOS given in J2000 or inertial frame) and which objects have been used for the simulations reported by SPIRE-UCF-NOT-002315 issue 1.				
SPIRE will provide a catalog suitable objects for system calibration. SPIRE need for this the visibility tool to be provided by ESOC FD.	AI#7 SPIRE <mark>TBD</mark> -05			
4. Operation necessary to reach the mode where best accuracy is achieved				

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The instrument operation and spacecraft operation are independent since the attitude of the spacecraft remains constant during the 7-point map.	
5. AOB Values of Ky and Kz :	
Values have been provided in AVM Test Specification SPIRE Peak-up Mode Test reference SPIRE-RAL-NOT-002372 issue 1. They have been commented by Alcatel saying that this gives a possible correction of 327.67 arcsec (4.8481368110954e ^{-8*215}). System Requirement Specification says that maximum allowed correction is 10 arcsec (upper correction will be discarded by ACMS). If 10 arcsec is confirmed, Ky and Kz could be optimised. SPIRE will confirm the ky and Kz values of SPIRE-RAL-NOT-002372 issue 1.	AI#8 SPIRE <mark>TBD</mark> -05
Ground peak up testing	
 The <u>AVM case</u> has been analysed via AVM Test Specification SPIRE Peak-up Mode Test reference SPIRE-RAL-NOT-002372 issue 1. Associated questions are : ⇒ What is the procedure to reach the REDY mode (mentioned in section 2.3) ? 	
 ⇒ Is the associated command list send by ground TC ? Where are they described (HPSDB) ? ⇒ Is the AVM procedure applicable to The <u>Spacecraft</u> case ? 	
SPIRE will answer the above questions about ground peak up testing	AI#9 SPIRE <mark>TBD</mark> -05

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	ACTION				DATE
INITIATOR Firm / person	N°	DESCRIPTION		ACTIONEE Firm / person	DUE
ALCATEL	01	SPIRE will assess the possibility to run this simula 0.3 beam).	RE will assess the possibility to run this simulation (grid spacing is beam).		TBD
SPIRE	02	Alcatel will detail the frequency content of APE	catel will detail the frequency content of APE. ALCATEL		15-JULY-05
ALCATEL	03	SPIRE will update the [R2] simulations based the	IRE will update the [R2] simulations based the Alcatel input		TBD
ALCATEL	04	IRE will update [R1] adding references where $\delta \theta_{\text{pos}}$ is quantified.		SPIRE	TBD
ALCATEL	05	SPIRE to identify in detail the conditions (instrument external and internal conditions) for achieving 0.5arcsec accuracy. SPIRE will elaborate a budget to analyse different contributor weight and be able to identify different conditions to achieve the goal.SPIRETBD			TBD
ALCATEL	06	SPIRE will update [R1] note including other dete	SPIRE will update [R1] note including other detector accuracy.		TBD
ALCATEL	07	PIRE will provide a catalog suitable objects for system calibration		SPIRE	TBD
ALCATEL	08	PIRE will confirm the ky and Kz values of SPIRE-RAL-NOT-002372 issue		SPIRE	TBD
ALCATEL	09	SPIRE will answer the above questions about gr	PIRE will answer the above questions about ground peak up testing		TBD

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