

Project Document

SPIRE SAFETY SUBMISSION

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Issue	4.0
Date	18 ^h June 2004
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SPIRE

SUBJECT:

SPIRE SAFETY SUBMISSION

PREPARED BY:

E.A.CLARK

DOCUMENT No:

SPIRE-RAL-DOC-001293

ISSUE

4

APPROVED BY:

Name

Date:

Signature

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Principle Investigator

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Project Manager K.J. King

Instrument Development Manager E. Sawyer

PA Manager E. Clark

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Distribution

Live Link

Change Record

ISSUE	DATE	
1.0	22 Mar 02	1 st Issue of Hazard check list
2.0	10 Jul 02	Combine Safety checklist & Submission into one document.
3.0	15 th May 03	SBT Report added.
4.0	18 th June 04	Residual Hazard report & SBT report Issue 1.2 added.

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1 Introduction

This document details the analysis of the SPIRE safety submission

2 SCOPE

To investigate and identify any Safety hazards, with respect to ground and launch operations, for the SPIRE instrument.

3 DOCUMENTS

3.1 Applicable Documents

AD1 SPIRE Product Assurance Plan (SPIRE-PROJECT-PRJ-000017)

AD2 Spire IBDR 5th & 6th March 2002.

3.2 Reference Documents

PT-RQ-04410 PA requirements for FIRST/PLANCK
CSG-RS-10A-CN Safety Regulations General Rules V1
CSG-RS-21A-CN Safety Regulations Specific Rules Ground installations
V2/P1
CSG-RS-22A-CN Safety Regulations Specific Rules Spacecraft V2/P2
HSO-SBT-SP-001 Sorption Cooler Specification
HSO-SBT-PI-013 Sorption Cooler AIV Plan

4 SAFETY ASSESSMENT

The SPIRE Instrument is completely integrated into the satellite prior to launch site operations, and is not accessible. When cold there are no safety hazards the only potential problem at normal temperature is the Sorption Cooler but the design is such that no hazard is envisaged, a report on the testing of the Sorption Cooler is attached.

Note: - The Sorption Cooler is identical to the PACS device.

A description of the SPIRE instrument can be found on LiveLink IBDR $5^{\rm th}$ & $6^{\rm th}$ March 02. (AD2)

All materials and parts currently used on SPIRE are detailed on the relevant declared lists, and have no known hazards other than has already been indicated in this document.



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5 HAZARD SOURCE CHECK LIST

Hazard Source (Potential)	Applicable To Payload Element (S) (E.G. Sensor, Boom, Electronics Box)	GES
Hazardous Electrical Systems (E.G. High Voltage)	NONE	NONE
Electroexplosive Devices (Pyrotechnics)	NONE	NONE
Propellants, Solid, Liquid	NONE	NONE
Pressurised Items	Sorption Cooler	Sorption Cooler
	80 Bar at room Temperature None when Cold	80 Bar at room Temperature None when Cold
Chemical Products - Corrosive (E.G. Battery) - Toxic Or Asphyxiating - Explosive (Also Pyros) - With Biological Effect	NONE	NONE
Radiation - Non-lonising - Ionising - Visible, Ir, Uv - Acoustic / Vibration Emission	NONE	NONE
High / Low Temperature	Yes.	Yes.
(E.G. Cryogenic)	Not controlled by SPIRE	Not controlled by SPIRE
Deploying Mechanism	NONE	NONE
Other Hazard Sources	NONE	NONE

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6 APPENDIX A REPORT

6.1 HSO-SBT-TN-076 Sorption Cooler Safety Issues 1.2



SPIRE & PACS Sorption Coolers

Safety issues

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SPIRE & PACS Sorption Coolers SAFETY ISSUES

SBT internal ref: SBT/CT/2003-37

	Name & Function	Date	Signature
Prepared	L. Duband - Cooler project manager		
SBT PA Check	M. Dubois – Cooler PA manager		
SPIRE Approval			
PACS Approval			
PA Approval		N/A	
Project Approval	J.L Augueres - SAp HSO project manager	N/A	
Project Approval	L. Duband - Cooler project manager		

Service des Basses Températures (SBT) Département de Recherche Fondamentale sur la Matière Condensée (DRFMC) COMMISSARIAT A L'ENERGIE ATOMIQUE - GRENOBLE (CEA-Grenoble) 17, rue des Martyrs 38054 GRENOBLE Cédex 9, France.



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Document Status

Issue	Revision	Date	Nb of pages	Modifications
 0	0	04/04/2003		First draft
 1	0	15/04/2003		First issue
1	1	10/07/2003		Modification of table on & 4.2.2
				following ESA's comments after IHDR
1	2	15/06/2004		3D views of cooler updated



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List of Acronyms

AD	Applicable Document		
CEA	Commissariat à l' Energie Atomique		
CDR	Critical Design Review	Revue de conception détaillée	RCD
CQM	Cryogenic Qualification Model		
ECSS	European Cooperation for Space Standardisation		
FIRST	Far Infrared and SubmillimetreTelescope		
FS	Flight spare		
HSO	Herschel Space Observatory		
N/A	Not Applicable		
PACS	Photoconductor. Array Camera and Spectrometer		
PFM	ProtoFlight Model		
PSS	Product Assurance Specification System		
RD	Reference Document		
SAp	Service d'Astrophysique		
SBT	Service des Basses Températures		
SCO	Sorption Cooler (full unit)		
SPIRE	Spectral & Photometric Imaging Receiver		



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SERVICE DES BASSES TEMPERATURES [CEA/DSM/DRFMC/SBT]

1 Scope of the document

This note discuss the safety issues of the SPIRE/PACS sorption coolers. It deals with the potential damages from the cooler to the user, as well as from the user to the cooler. Although the stored energy remains low, these systems are under 8 MPa when at room temperature. Safety issues related to the high pressure are discussed.



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Documents

Applicable documents

HSO-SBT-PR-025: Test en pression

HSO-SBT-PR-027: Handling, packing, transportation and storage manual



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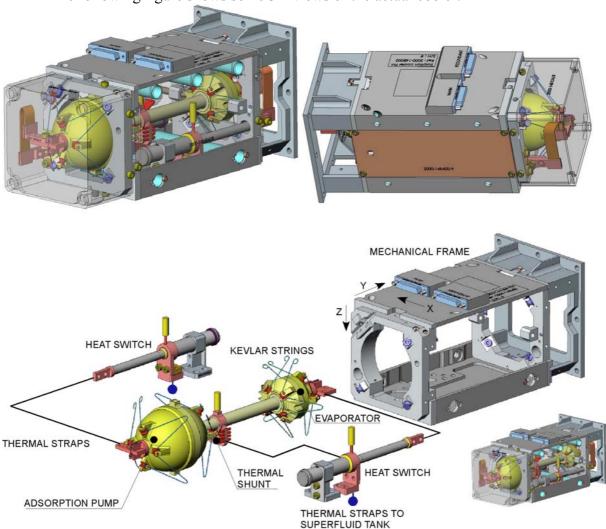
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3 <u>Description of sorption cooler</u>

The cooling of the SPIRE and PACS detectors down to 300 mK will be effected by a helium three sorption cooler. This sub-Kelvin sorption cooler provides a wide range of heat lift capability at temperature below 400 mK. It relies on the capability of porous materials to adsorb or release a gas when cyclically cooled or heated. Using this physical process one can design a compressor/pump which by managing the gas pressure in a closed system, can condense liquid at some appropriate location and then perform an evaporative pumping on the liquid bath to reduce its temperature. Helium sorption refrigerators have no moving parts, are vibrationless and can be designed to be self contained and compact with a high duty cycle efficiency.

The following figure shows some 3D views of the actual cooler.





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4 Potential damages to the user

4.1 Cooler manipulation

The cooler must be handled with care and using gloves to avoid any contamination to it. As described further all sensitive elements are enclosed in protective covers.

Although the cooler can be manipulated without any major risk, the external mechanical frame features some sharp edges.

All materials used are harmless. The mechanical frame is mostly made out of Titanium Ta6V and aluminum, and include a couple of copper interfaces.

The cooler internal contains helium gas (³He). Helium is an inert gas and, in case of a gas leak, the amount stored (6 liters NTP) is not significant enough to create any problem.

4.2 Pressure related aspects

4.2.1 <u>Leak before burst</u>

The large internal pressure required for efficient operation of the cooler, as well as for size purpose, certainly calls for an appropriate design. The design has been performed so that the pumping line, ie the thin walled tubing, is the weakest mechanical part of the cooler with respect to the internal pressure. The margin of safety on all other components are significantly higher than that of the tube.

Experimental pressure tests performed with water on pieces of tubing show that when the internal pressure reaches the bursting pressure the tube opens up along its length, and the internal fluid or gas quickly leaks to the outside, leading to leak before burst type behavior.

In a previous ESA contract a test piece representative of the pumping line has been manufactured and tested to demonstrate this behavior. A schematic of this piece is shown below.



The calculated burst pressure of this piece was 35 MPa (350 bars) and consequently the pressure test could only be safely performed with a water pump. Thus prior to testing this piece, a couple of stainless steel samples (smaller diameter and wall thicknesses) were also manufactured to check for any difference in behavior between water and gaz pressure test and additionnally any difference in behavior between stainless steel and titanium.

The results are displayed on the following figures.



SPIRE & PACS Sorption Coolers

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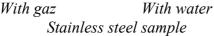
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With water *Titanium sample*

For all the tests the failure appears as expected in the middle of the tube, along its length; with gaz, as gas is exhausted and because of the remaining internal pressure the crack propagates toward the copper brazed end, which in the case of the small stainless tube is then teared up. The tube does not break apart and no "flying" pieces are seen – with water the failure is "clean": basically the same (middle of tube, along its length), but since the pressure is almost instantly released, the crack does not propagate.

The experimental bursting pressure for the titanium sample was found to be 36 MPa, consistent with the calculated 35 MPa.

In this case the visual aspect is quite different from stainless steel and is probably due to the difference in ductility: the breaking elongation for stainless steel can reach values higher than 60% as for Ta6V it remains below 15%. In any case the failure appears in the middle of the tube and then propagate along its length.

We believe these tests have demonstrated the tube failure at the burst pressure is of the "leak before burst" type. Consequently if the tube is designed to be the weakest point in the system with regards to internal pressure the cooler can be qualified as a leak before burst device.

In addition as described further, the cooler internal (components under pressure) are fully enclosed to further increase the safety aspects.

4.2.2 Proof pressure testing

Obviously the real coolers are not tested to failure. However to provide confidence on the quality of the tubes and on all of the soldering, each cooler is proof pressure tested under helium at 20 MPa (200 bars) – twice the maximum operating pressure (MOP).



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The maximum operating pressure is defined as the maximum pressure the cooler will ever reach in its lifetime. This maximum pressure occurs during the bake out process at 80°C. Consequently the cooler has been designed with an operating pressure of 8 MPa at ambient (20°C), leading to 10 MPa at 80°C. all structural elements have been designed accordingly. The table below displays the stress at the MOP and the margin of safety (MOS). The MOS is defined as:

$$MOS = \frac{max \ allowable \ stress}{applied \ stress} - 1$$

The maximum allowable stress is taken as the elastic limit.

	Stress at MOP (MPa)	MOS
Pumping line	250	2.2
Evaporator	77	9.4
Sorption pump	125	5.4

(Note: titanium Ta6V, actual measurements – elastic limit: 798 MPa – ultimate strength: 875 MPa)

The table hereafter displays the burst pressure and margins with regards to the maximum operating pressure.

	Burst Pressure (MPa)	MOP (MPa)	Margin
Pumping line	35	10	3.5
Evaporator	113	10	11.3
Sorption pump	70	10	7

During the proof pressure test, the helium leak rate is recorded. The results of each proof pressure test are reported on an inspection sheet (FI) as shown hereafter (FI for SPIRE and PACS COM units).



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Grenable	SPIRE & PACS Sorption Coolers FICHE D'INSPECTION (FI)	Référence : HSO-SBT-FI2.1.
SERVICE DES BAS	SSES TEMPERATURES	
Type d'inspection :		Date: 09 04 02
1. Approvisionnement	3. Réception pièces	Nom:
2. Equipement	(4) Autre leure en pression	- Clinc
Type de contrôle :		Signature :
1. Visuel	3. Fonctionnel	Gel
2. Dimensionnel	4. Autre melanique	(10)
	cernant l'équipement à inspecter :	
Modèle concerné : +	tigo He3 HSO 2000-14 - C	NHG
Nom et réf. item : B2	00/1	
Fabriquant :	Sous traitant : Of	16
Renseignements qua	lité :	
PV contrôle:	Référence spéc. :	
	100: Test à 20 MPa (200	
Presnou finale	100: Test à 20 MPa (200 193 tars. Temps, 2mn. 100 - 1.109 mtart/s	



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Gren ble	SPIRE & PACS Sorption Coolers FICHE D'INSPECTION (FI)	Référence : HSO-SBT-FI22.		
SERVICE DES BAS	SSES TEMPERATURES			
Type d'inspection :	•	Date: 10 04 02		
1. Approvisionnement	3. Réception pièces	Nom: (1 FDC		
2. Equipement	4) Autre Tenne en pression	and the same		
Type de contrôle :		Signature :		
1. Visuel	3. Fonctionnel,	Core		
2. Dimensionnel	(4) Autre melamque	100		
	ncernant l'équipement à inspecter :	· ·		
Modèle concerné : +	rigo He3 HSO 2000-14 - CQ1	12		
Nom et réf. item : 320				
Fabriquant :		16		
Renseignements quo	alité :			
PV contrôle :	Référence spéc. :			
Rapport: Task Coloo: Test à 200 tars.				
Task coloo: Test à 200 tars. Pression finale 200 tars, temps 2mm. Nivean réviduel Het ent spectro en 3.100 mearl s.				
Anomalie(s): Visa Projet - Nom, da Visa Qualité - Nom,		* Ambai		



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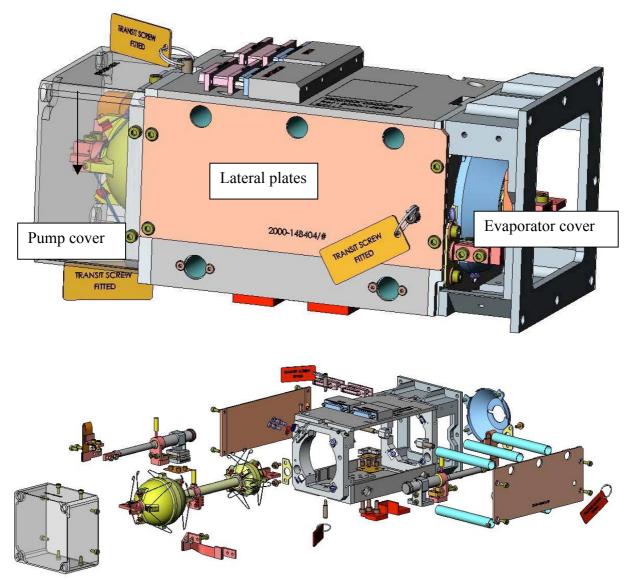
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5 Potential damages to the cooler

The cooler sensitive parts are mainly the thin wall titanium tubes as well as an original suspension system using Kevlar strings. Whenever possible the cooler internal area is protected by covers to prevent any damages. Seen from the outside the cooler is pretty much a closed box with limited access to any internal parts and no access to sensitive parts (see figure below). The dismounting of any covers or plates is prohibited by non authorized staff.



3D exploded view showing the protective covers, lateral plates and guiding tubes for the screws (PACS interface)

5.1 <u>Internal access to the cooler</u>



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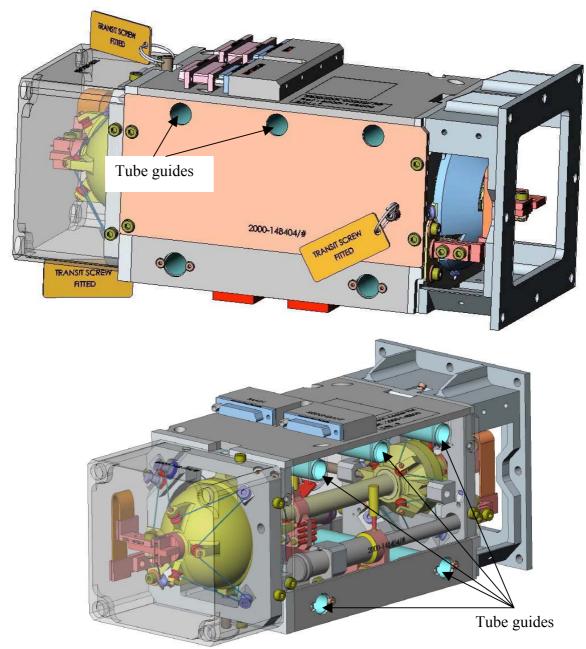
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As much as possible the internal access to the cooler has been restrained and is limited to the mounting interfaces operation, which only concerns the mounting within the PACS instrument.

Within the PACS instrument, the cooler is mounted on the side via 5 screws. These screws are inserted inside the structural box, and to prevent any damage to any internal component as well as to ease the screw mounting, aluminum tubes have been included in the design as guides for the screws and tool (see previous figure and below).



Side view of unit – 5 through holes for PACS mounting (Lateral plate removed – tube guides are visible)



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6 On orbit configuration

Once in orbit the cooler is thermally connected to a superfluid helium tank. The internal pressure of the cooler is then limited to values lower than typically 0.1 MPa (1 bar). Note that this is also the case whenever the cooler is cooled to below roughly 10 K. Consequently the cooler is not anymore a pressure vessel once cold.

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7 APPENDIX B RESIDUAL HAZARD REPORT.



RESIDUAL HAZARD REPORT

PRODUCT ASSURANCE Space Science and Technology Department

Spacecraft / Project	HERSCHEL / SPIRE	Originator's Name	Eric Clark
System / Experiment / Model	CQM/PFM	Signature	
Sub-System		Date	18 th June 2004
Sub-Assembly	Sorption Cooler	Document No/Issue	SPIRE-RAL-DOC-001293 is 4.0
Item		HAZARD CATEGORY	Insignificant
Serial Number			

Title	SPIRE Safety Submission
	o <u> </u>

Description of Hazard Source and Potential Effects of Hazard:

The Sorption Cooler is a closed volume containing He³, and the maximum pressure occurs during the bake out process at 80°C. Consequently the cooler has been designed with an operating pressure of 8 MPa at ambient (20°C), leading to maximum Operating pressure 10 MPa at 80°C.

The potential hazard is from this volume fracturing.

Applicable Safety Requirements

AD10-1 - CSG-RS-10A-CN 5 3 - CSG Safety regulation Vol 1-General Rules.

AD10-3 - CSG-RS-22A-CN_5_4 - CSG Safety regulation Vol 2 Part 2 - Specific rules Spacecraft.

Hazard Controls and Acceptance Rationale:

- 1. The Sorption cooler only presents this hazard potential when warm. Herschel will spend most of its time cold, in particular all of its time cold when near the launch vehicle. Under these conditions pressure in the cooler is very much less than 1 atmosphere.
- 2. The amount of He³ is 6 litres at STP and even its instantaneous release into a warm Herschel cryostat will not raise the cryostat's internal pressure by 1 atmosphere.

 The demonstrated fracture characteristic is the titanium tube splitting which only occurs at exceptional over proof pressure and even then it is a gradual leak, not an explosive discharge, (Calculated burst pressure of this piece was 35MPa (350 bars) tested was 36MPa) and for SPIRE the cooler is in a substantial vented Aluminium box.
- 3. Each Cooler is proof pressure tested under helium at 20 MPa (200 bars) twice the maximum operating pressure (MOP).
- 4. The Coolers are run during instrument tests where a very small He³ leak (indicative of any problems) would be very easily detected.

Attachments				
None				
	PA Manager (Or Deputy)	Project Manager		
Required Signatures & Date	Trimanager (er zepaty)	1 Tojost manager		