

# **SPIRE Critical Items List**

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**DOCUMENT No.: SPIRE-UCF-PRJ-001138**

**ISSUE: 2.0**

**Date: 6 June 2002**

## Change Record

Issue	Date	Comments
1	4 February 2002	Issued for SPIRE IBDR
2	6 June 2003	Issued for SPIRE IHDR

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## Introduction

This document provides a list of technical and/or programmatic issues which are regarded by the SPIRE Project Team as requiring action in order to maintain the project schedule and the instrument development plan and scientific capabilities. These are receiving on-going attention from the Project Team in consultation with the relevant subsystem groups, ESA, and industry. Progress is regularly tracked in regular Project Team meetings and in *ad hoc* meetings and telecons as appropriate. Each item is the responsibility of one particular individual in the Project Team.

<b>Item</b>	<b>1. SPIRE Overall Schedule</b>
<b>Status</b>	Under regular assessment
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Instrument
<b>Responsible institute</b>	RAL
<b>Nature of criticality</b>	SPIRE schedule compatibility with industry's stated instrument delivery dates
<b>Responsible persons</b>	Eric Sawyer
<b>Failure effect</b>	Late delivery or de-scoped testing
<b>Description</b>	On-time delivery of the CQM is not now feasible without untenably limited and risky programme. An alternative plan has been proposed which involves redefining the STM/CQM programme with delivery of upgraded STM (or down-graded CQM). The alternative STM/CQM/PFM development plan has been presented to ESA and Industry at the last SPIRE Interface meeting in April. The adaptability of the subsystem development plans (especially the DRCU) needs to be assessed. The purpose of this adapted plan is to protect as much as possible the on-time delivery of the PFM. It involves some compromises and added risks to the CQM programme.
<b>Due date of corrective actions</b>	A decision from ESA on whether or not SPIRE should adopt the new approach is expected by the end of June.

<b>Item</b>	<b>2. STM/CQM Programme</b>
<b>Status</b>	Currently underway
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Instrument
<b>Responsible institute</b>	RAL
<b>Nature of criticality</b>	Technical problems; schedule
<b>Responsible persons</b>	Eric Sawyer
<b>Failure effect</b>	Redesign and retest; late delivery or de-scoped testing
<b>Description</b>	The STM/CQM programme includes the first assembly and test of many subsystem and system elements. Various teething troubles and small problems continue to arise and need to be addressed, requiring much attention from the Project Team, at the expense of other areas of work. Schedule is vulnerable at this stage to any AIV facility problems.
<b>Due date of corrective actions</b>	The programme is under continual review and assessment.

<b>Item</b>	<b>3. DRCU Development Plan</b>
<b>Status</b>	Reviewed in April. Currently under discussion with CEA
<b>Last update</b>	June 9
<b>Subsystem</b>	DRCU
<b>Responsible institute</b>	CEA; RAL
<b>Nature of criticality</b>	Compatibility of DRCU schedule with instrument CQM and PFM schedules needs to be established
<b>Responsible persons</b>	CEA: Jean-Louis Augueres                      RAL: Eric Sawyer
<b>Failure effect</b>	N/A
<b>Description</b>	DRCU development plan favoured by CEA involves waiting until QM design verified by instrument CQM tests before starting PFM programme. This results in late delivery of the QM2 DRCU meaning that when QM1 is delivered to industry with the CQM, no electronics are available at RAL for instrument testing for approx. 6 months.  See DRCU Programme Review Board Report. Discussions are continuing with CEA.
<b>Due date of corrective actions</b>	Status to be reported at IHDR.

<b>Item</b>	<b>4. FDIR</b>
<b>Status</b>	In progress but slowed due to pressure of work
<b>Last update</b>	6 June
<b>Subsystem</b>	System; OBS
<b>Responsible institute</b>	RAL
<b>Nature of criticality</b>	Autonomy modes
<b>Responsible persons</b>	Ken King
<b>Failure effect</b>	Delay in OBS testing
<b>Description</b>	FDIR analysis has been slowed due to lack of resources and late availability of information from subsystems. All information is now available, and additional resources are being deployed.
<b>Due date of corrective actions</b>	Analysis needed by end of June. Update will be given at IHDR meeting.

<b>Item</b>	<b>5. Spacecraft thermal interfaces</b>
<b>Status</b>	ECR09 has not yet been accepted by ESA/industry as a statement of SPIRE's requirement
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	SPIRE FPU: thermal behaviour
<b>Responsible institute</b>	RAL
<b>Nature of criticality</b>	Thermal characteristics of SPIRE and its environment are critical to its scientific performance.
<b>Responsible persons</b>	John Delderfield
<b>Failure effect</b>	Cooler hold time less than 46-hour spec.; loss of sensitivity due to thermal drifts and/or detector temperature running too high.
<b>Description</b>	See ECR09 and minutes of recent interface and dedicated thermal meetings. SPIRE is concerned that optimisation of the overall thermal system is critically important for instrument performance. The iterative approach currently being pursued by ESA/industry is the most constructive way forward in practical terms; SPIRE would like to see this expedited as much as possible.
<b>Due date of corrective actions</b>	SPIRE is in any case planning to implement all necessary measures in order to achieve the best thermal performance of the instrument (see items 7 and 8)



<b>Item</b>	<b>9. FTS mechanism vibration qualification</b>
<b>Status</b>	Open
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	SMECM
<b>Responsible institute</b>	LAM
<b>Nature of criticality</b>	Vibration qualification.
<b>Responsible persons</b>	LAM: Dominique Pouliquen                      RAL: Bruce Swinyard
<b>Failure effect</b>	Mechanism redesign; possible failure to meet spectral resolution goal
<b>Description</b>	SMECM design was not compatible with vibration levels predicted by FEA model (by a factor of four). Following instrument level vibration tests, levels have been lowered by more than the required factor.
<b>Due date of corrective actions</b>	Qualification tests will take place at unit level before delivery of the PFM SMECM in November 2003.

<b>Item</b>	<b>10. BDA qualification</b>
<b>Status</b>	Under regular review
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Detectors
<b>Responsible institute</b>	JPL
<b>Nature of criticality</b>	BDA qualification and test programme have been severely delayed due to problems with Kevlar supports for 300-mK stage.
<b>Responsible persons</b>	JPL: Marty Herman      Cardiff: Matt Griffin      RAL: Bruce Swinyard
<b>Failure effect</b>	Schedule; detector performance; risk of single-point failure
<b>Description</b>	Mechanical qualification not yet achieved; BDA fabrication and test has been delayed; minimal time now available to build, test, deliver JPL hardware. CQM arrays not yet tested FS deliverability may be at risk due to JPL budget constraints.
<b>Due date of corrective actions</b>	Programme is under regular review in weekly telecons with the Project Team.

<b>Item</b>	<b>11. JFET power dissipation</b>
<b>Status</b>	Under regular review
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Detectors
<b>Responsible institute</b>	JPL
<b>Nature of criticality</b>	JFET Total power dissipation when optimised for low noise operation is too high in photometer mode.
<b>Responsible persons</b>	JPL: Jamie Bock RAL: Bruce Swinyard
<b>Failure effect</b>	Loss of scientific performance due to adverse thermal effects, operation at reduced power/higher noise or switching off of selected channels to reduce overall dissipation.
<b>Description</b>	For best noise performance, the JFETs need to be operated at about 120 K. The overall dissipation needed to achieve this in photometer mode (worst case) is approx. 70 mW (allocation is 42 mW). Power can be reduced at the expense of higher noise, or some FETs can be switched off. The thermal impact will not be fully understood until CQM tests at system level have been carried out.
<b>Due date of corrective actions</b>	CQM tests will clarify power dissipation. JPL are investigating change to JFET support membrane design as alternative design to reduce heating power needed (not a SPIRE-funded programme; best efforts only). RAL are investigating increasing thermal isolation of JFET rack.

<b>Item</b>	<b>12. 300-mK thermal strap supports</b>
<b>Status</b>	Mechanical failure in instrument STM; recovery plan formulated and underway
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Thermal straps
<b>Responsible institute</b>	Cardiff
<b>Nature of criticality</b>	Potential single-point failure for SPIRE
<b>Responsible persons</b>	Cardiff: Peter Hargrave RAL: Doug Griffin
<b>Failure effect</b>	Loss of SPIRE due to thermal short between 300-mK and 2 K.
<b>Description</b>	A 300-mK thermal strap system light-baffle and thermal strap support unit failed under warm vibration of the instrument STM. The MRB established the cause as a workmanship problem that led to Kevlar breaking due to friction, but concluded that the design should also be improved. The new design provides larger radii at the places where the Kevlar is wrapped around the metal support unit.
<b>Due date of corrective actions</b>	New design will be built, and tested at unit level before installation into the CQM for cooldown (scheduled for mid-August).

<b>Item</b>	<b>13. FPU internal black coating</b>
<b>Status</b>	Open
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	FPU
<b>Responsible institute</b>	Cardiff; MSSSL; RAL
<b>Nature of criticality</b>	Black coating is needed in certain places inside the SPIRE FPU to suppress stray light.
<b>Responsible persons</b>	Cardiff: Peter Hargrave MSSSL: Chris Brockley-Blatt RAL: Bruce Swinyard
<b>Failure effect</b>	CQM will have limited black coating installed for qualification purposes only.
<b>Description</b>	Qualification of suitable black compound is not yet complete, and the design and manufacture of the black tiles to be installed in the FPU has yet to be carried out. This programme has been slowed down due to the need to solve the problems that have arisen with the thermal strap supports. A work-around will need to be formulated to qualify for use on PFM.
<b>Due date of corrective actions</b>	Samples for CQM testing needed beginning of August. PFM blackening is needed in early October.

<b>Item</b>	<b>14. Correction of mirrors</b>
<b>Status</b>	Open
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Optics
<b>Responsible institute</b>	LAM
<b>Nature of criticality</b>	STM optics verification revealed misalignments
<b>Responsible persons</b>	LAM: Dominique Pouliquen RAL: Bruce Swinyard
<b>Failure effect</b>	Loss of efficiency due to misalignment
<b>Description</b>	CQM CM3 mirror is misaligned by 0.5° and two others have small misalignments. LAM are having CM3 remade - may be available for retro-fit in the CQM. In the worst case, CQM programme will use existing mirrors with compromised testing capabilities.
<b>Due date of corrective actions</b>	Problems to be corrected for PFM mirrors due in early October.

<b>Item</b>	<b>15. Filter availability</b>
<b>Status</b>	Open
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Filters
<b>Responsible institute</b>	LAM
<b>Nature of criticality</b>	Schedule; technical (filter delamination)
<b>Responsible persons</b>	Cardiff: Ian Walker RAL: Eric Sawyer
<b>Failure effect</b>	Late delivery of filters - compromised CQM schedule
<b>Description</b>	Production problems with large area filters need to be solved before filters for the CQM instrument and the AIV facility can be delivered. Work continues at Cardiff to solve technical problems.
<b>Due date of corrective actions</b>	In order to avoid the CQM schedule being driven by filter availability, CQM instrument and AIV filters needed by July 1. Status report will be given at the IHDR.

<b>Item</b>	<b>16. Microvibrations</b>
<b>Status</b>	Microphonics test plan has been formulated and issued.
<b>Last update</b>	6 June 2003
<b>Subsystem</b>	Bolometers; SMECM
<b>Responsible institute</b>	RAL
<b>Nature of criticality</b>	Shielding of the SPIRE detectors from EMI is critical to their scientific performance.
<b>Responsible persons</b>	Doug Griffin
<b>Failure effect</b>	Excess bolometer noise or non-functioning bolometers
<b>Description</b>	The SPIRE bolometers and FTS mechanism position sensor are potentially vulnerable to microvibration noise (for instance from the spacecraft gyros). Analysis of susceptibility by SPIRE and of the expected vibration environment by Alcatel indicates that the levels of vibration due to the spacecraft should not pose a significant problem. However, the uncertainties are considered to be large, and continuing analysis, measurement, and review will be important. However, it has been agreed that the system design of the spacecraft need not be driven by this issue. The operating modes of the SPIRE FTS, which is most sensitive to vibration, have been defined such that a step-and-integrate mode of data-taking can be implemented. This may involve some loss of basic sensitivity, but would be far less sensitive to microphonic disturbance.
<b>Due date of corrective actions</b>	Semi-quantitative tests will be carried out during CQM and PFM ILT, as described in the microphonics test plan.