

L.A.M. UMR 6110	HERSCHEL	Ref : SPI.STM.00.RC.01.A	Page : 1 / 12
	SPIRE SMECm	Author : P. Dargent	Date : 15 October 2001
Calculs linéaires sur l'Assemblage complet - approximé Linear FEA of approximated Assembly			

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1. Scope

This document gives preliminary results of static and resonance behavior of the SMEC mechanism Assembly.

2. Softwares

The SMECm F.E.A. has been implemented with CosmosWorks 6.0 and CAD solid model constructed in SolidWorks2000 to get estimations of eigen frequencies, characteristic deformed shapes, and highest stresses under 100G loads.

This software package is limited to linear analysis, and doesn't allow volumic and shell elements merging.

3. Definition of parts

Only main structural parts (45) have been taken into account, for a total weight of 1440 grams. Ignored parts are mainly static (preamp, harness, etc..) and have very few impact on this analysis.

Definition of every part has been slightly simplified to avoid divergence of automatic meshing process.

Main modifications between CAD model and FEA model are :

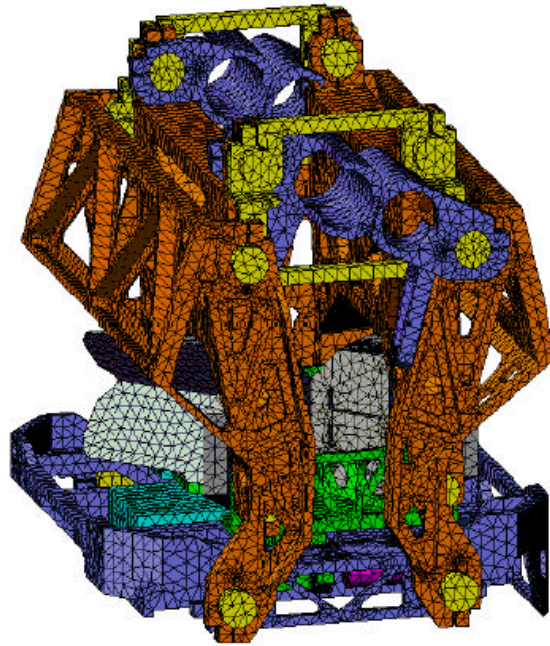
- Pivots have been replaced by homogeneous bulk parts with isotropic stiffness similar to pivot shear stiffness ***in non twisted position***. This rough simplification may affect results of calculations where torsion of pivots is solicited. Mass is preserved.
- Belts of synchronisation with 0.1 x 16 mm² of section have been modeled as 0,4 x 4 mm² short beams. Results of calculations where belts are bent will be affected.
- Latch has been simplified as plastic simple holds between actuator core and base structure. ***It doesn't take into account nonlinear behaviour of real latching*** due to few tens of microns of play in X, Y and Z directions.

BE System pivots have been modelised as simple rod of matter with $5 \cdot 10^9$ Pa modulus of elasticity, that would correspond approximately to a Nylon plastic reinforced with 30% of glass fibers. As comparison, 5016-800 TRW pivots, with lower shear stiffness for about the same torsion stiffness, would have been modelised by matter with $5 \cdot 10^8$ Pa modulus, i.e. a kind of Teflon.

All modulus are for materials at room temperature. 4K modulus are somewhat higher (about + 10% for aluminum), so frequencies will be under estimated, and static deflection over estimated.

4. Mesh

Meshing is automatic, and provides tetrahedral parabolique elements (TETRA10).
With SMECm we get 171190 elements, 52054 corner nodes, and 959115 D.O.F.



5. Latched Mechanism – Static loads

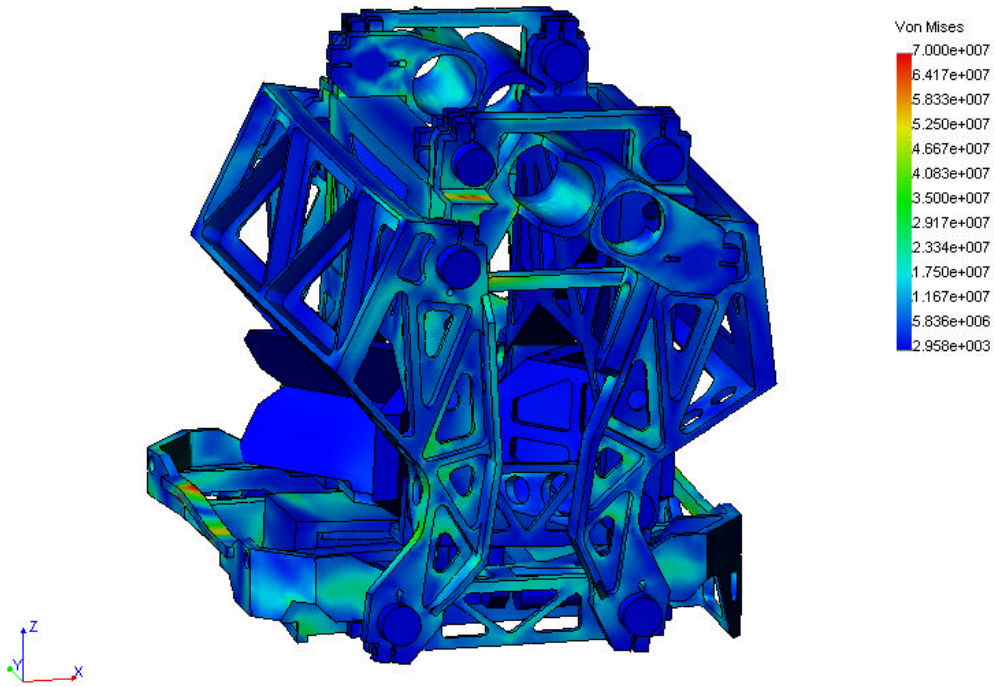
Static linear simulation have been done for 100 G acceleration in three main directions :

- X_m (parallele to mirrors movement)
- Y_m (perpendicular to the bench)
- Z_m (parallele rocking frames axis of rotation).

This analysis shows weakest sections of structural parts, and gives an estimation of maximal stress for all the assembly.

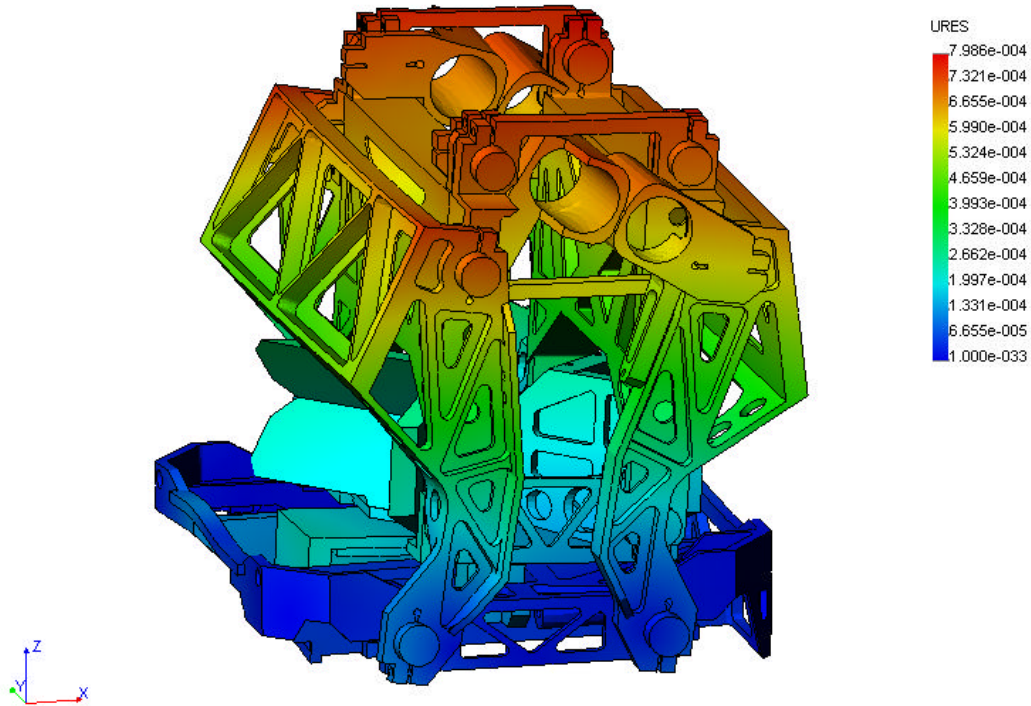
100 G in Xm :

Unités: Nm²



Highly solicited sections have been reinforced.
Maximal stress is 167 MPa

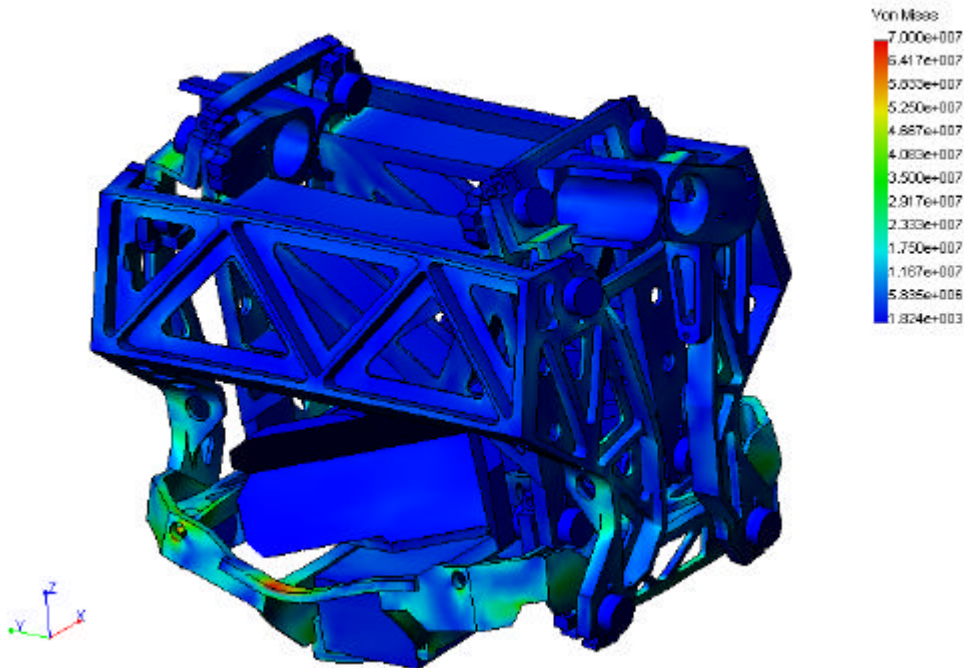
Unités: m



For 100G, maximal Longitudinal displacement is about 800 microns.

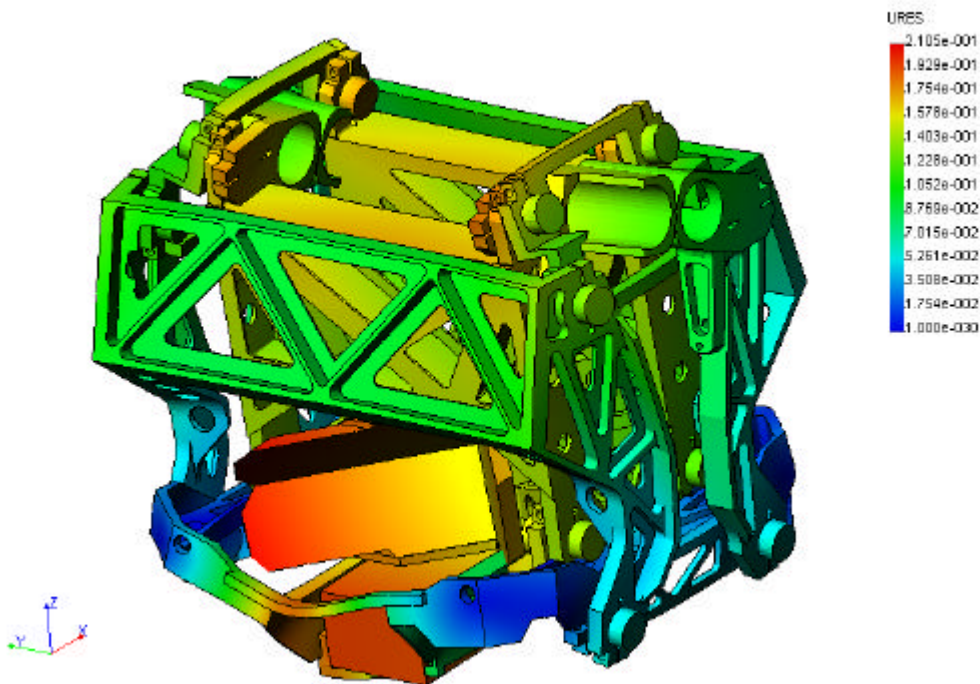
100 G in Ym :

Unités: Nmm²



Highly solicited sections have been reinforced.
Maximal stress is 96 MPa

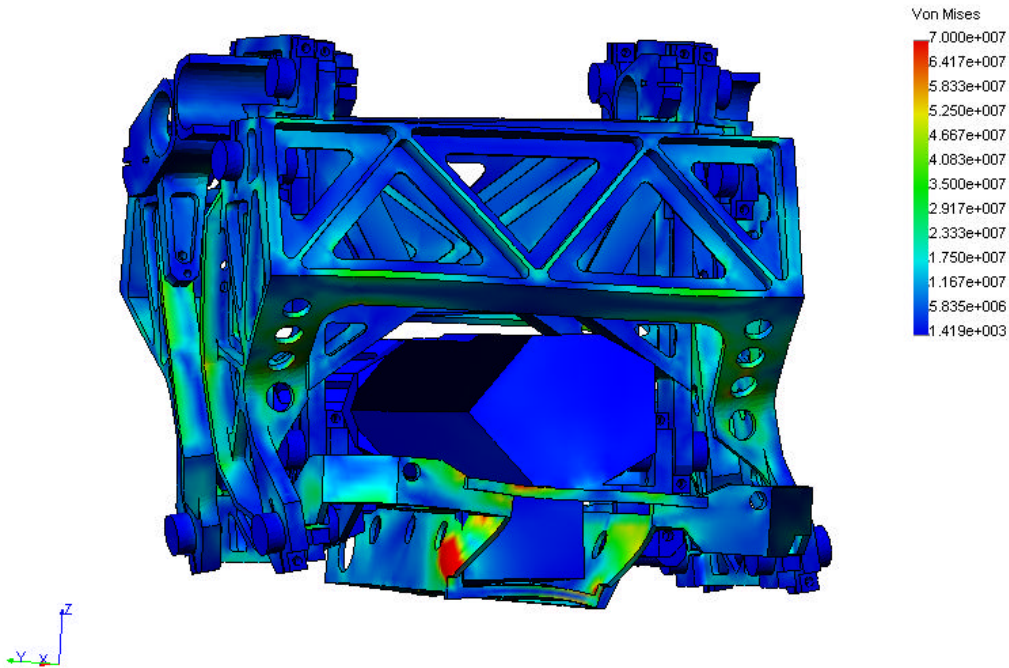
Unités: mm



For 100G, maximal vertical displacement is about 210 microns.

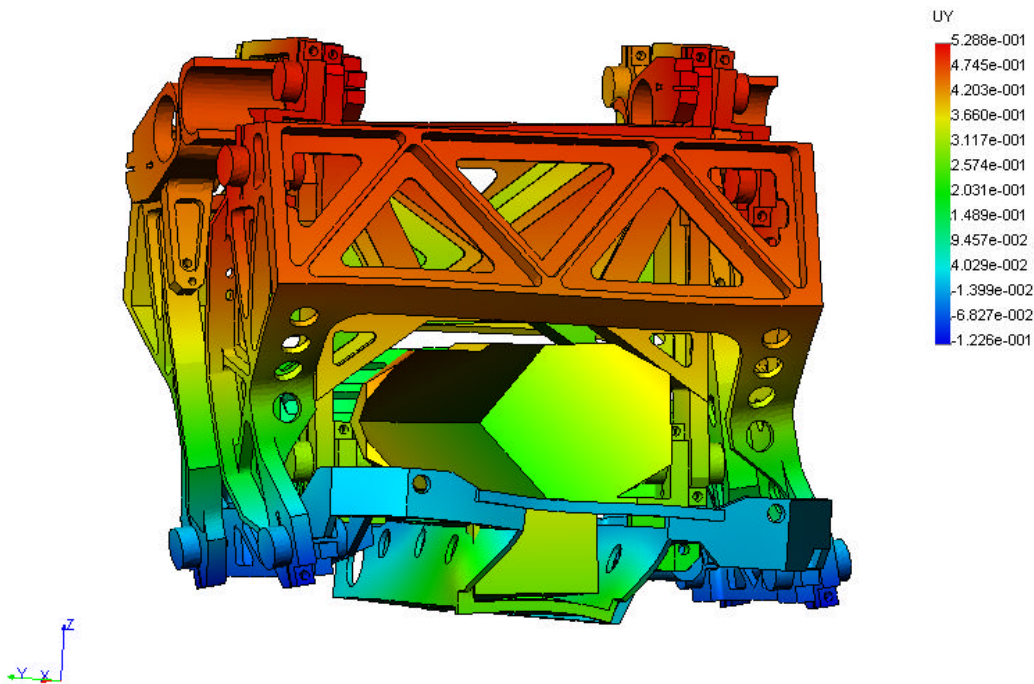
100 G in Zm :

Unités: N/m²



Highly solicited sections have been reinforced.
Maximal stress is 170 MPa

Unités: mm

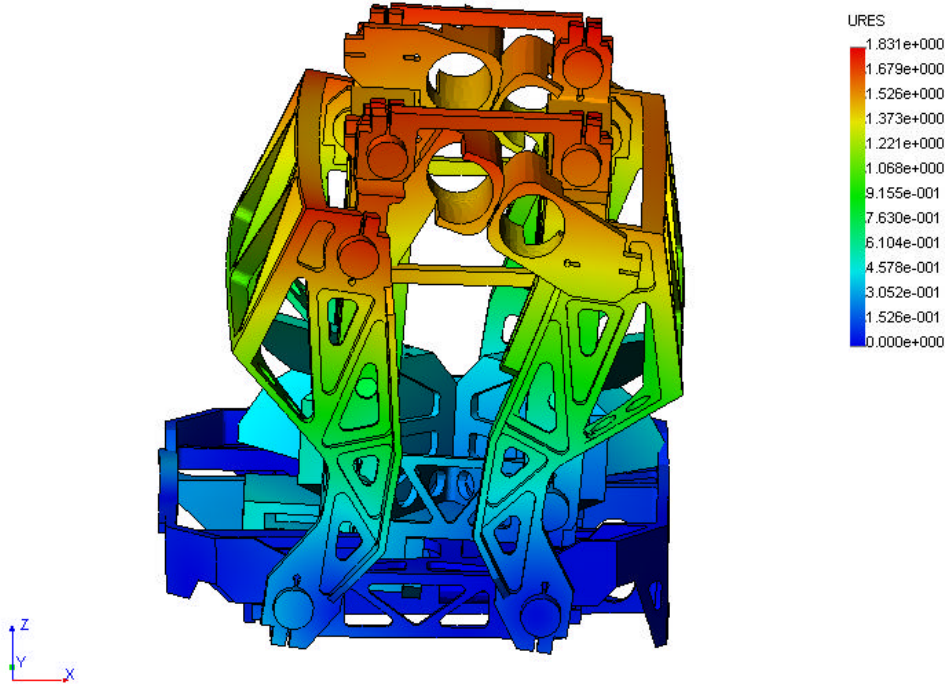


For 100G, maximal transversal displacement is about 530 microns.

6. Latched Mechanism – Eigen Frequencies

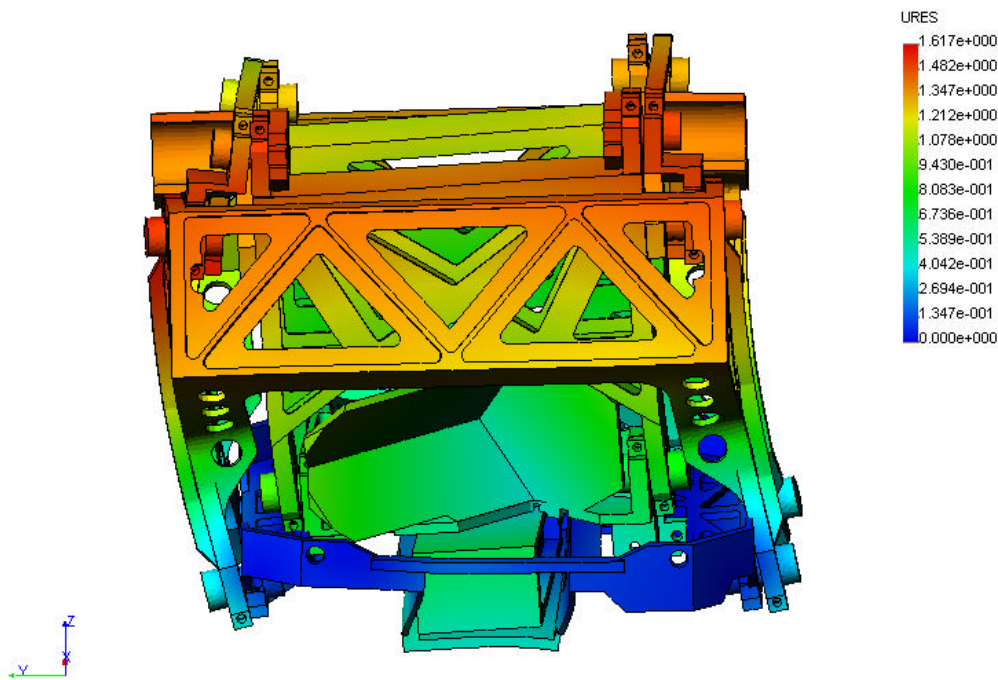
Number of modes was limited to 5.

Mode: 1



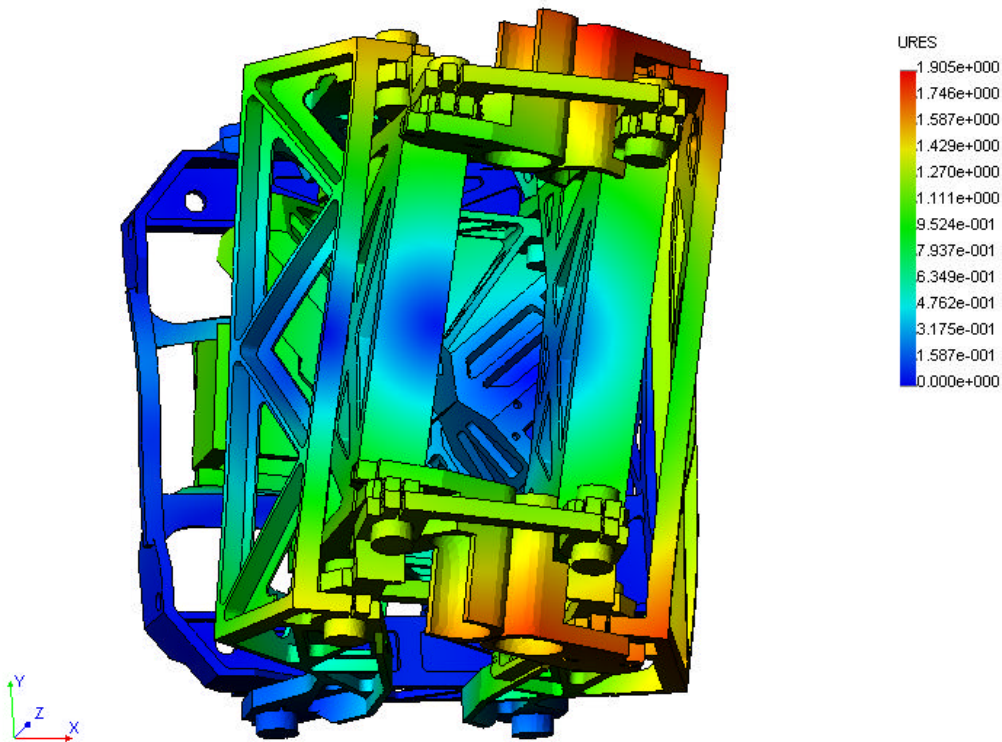
First Mode : 201,6 Hz – Longitudinal bending

Mode: 2



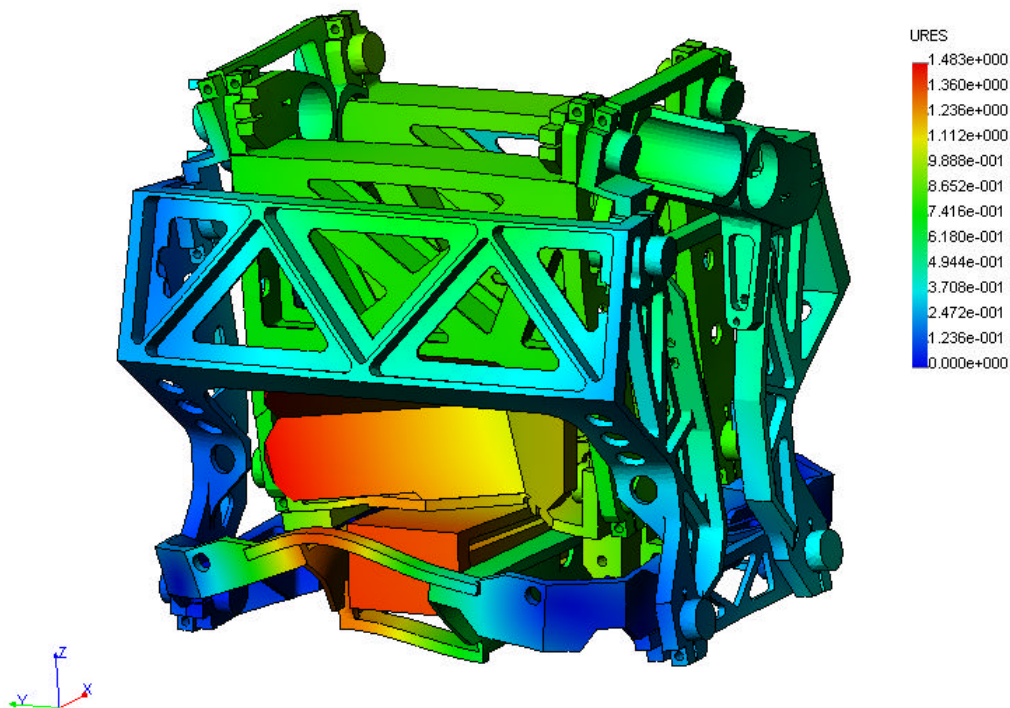
Second Mode : 243,8 Hz – Transversal bending

Mode: 3



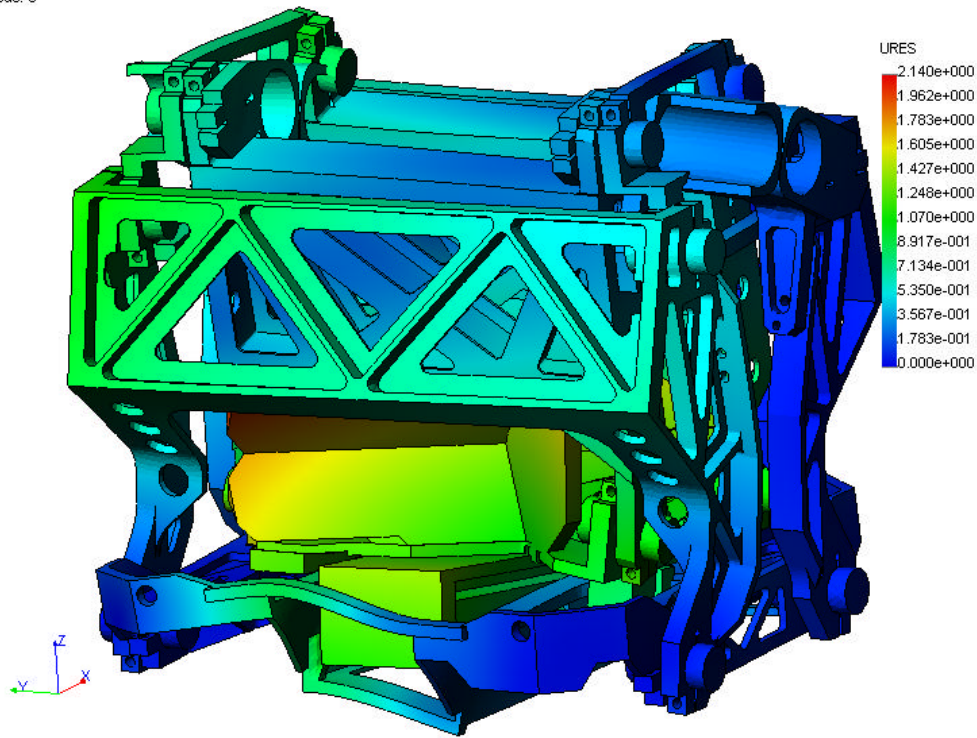
Third Mode : 296,2 Hz – Vertical torsion

Mode: 4



Fourth Mode : 431,3 Hz – Base plate oscillation

Mode: 5



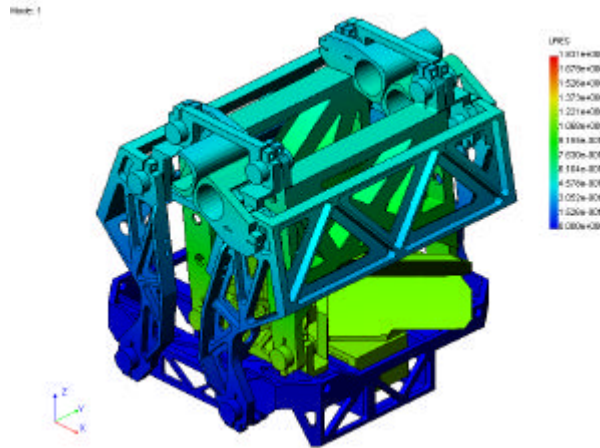
Fifth Mode : 499 Hz – Torsion + Bending

7. Unlatched Mechanism – Eigen Frequencies

Calculations have been also performed with same model as before, but without plastic holds on the actuator (simulating launch latch).

Purpose of this analysis is to know :

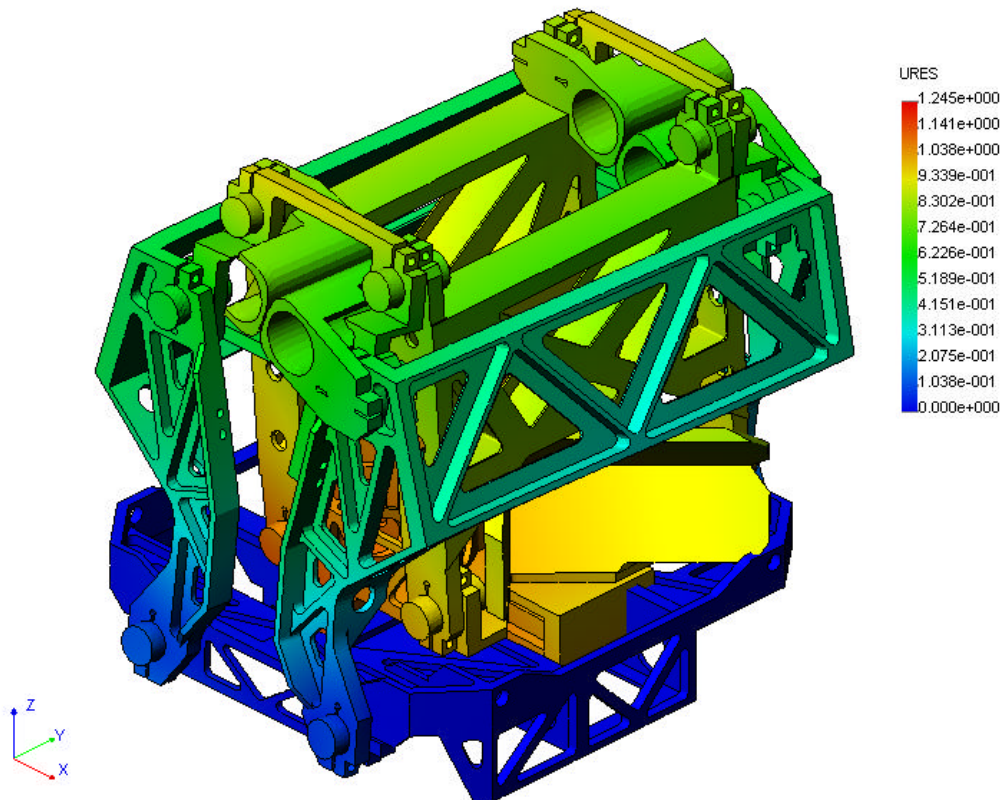
- what kind of spurious modes could affect PID control loop (at – 8mm absolute position of the carriage)
- what basic modes could be excited and nonlinearly limited by launch latch.



First Mode : 47,3Hz – « on axis » resonance ..

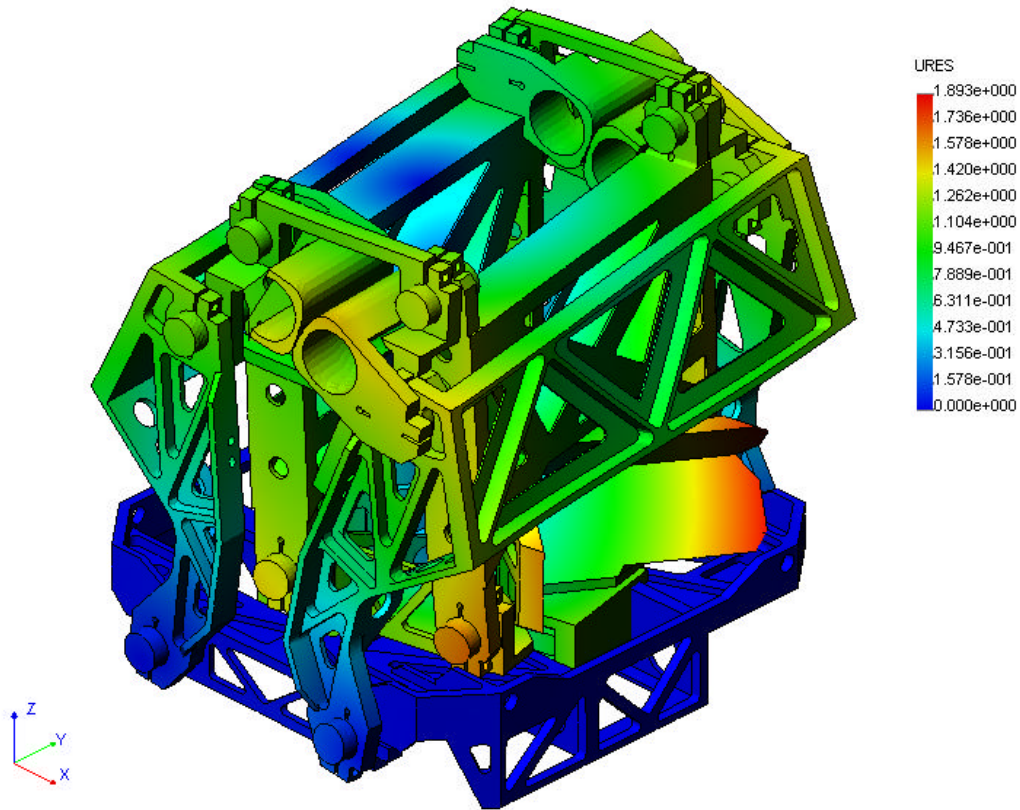
This Mode calculation is definitely very much affected by approximations of pivots and belts. Real value for X axis resonance is about 1 Hz.

Mode: 2



Second Mode : 94,8 Hz – Transversal deflection

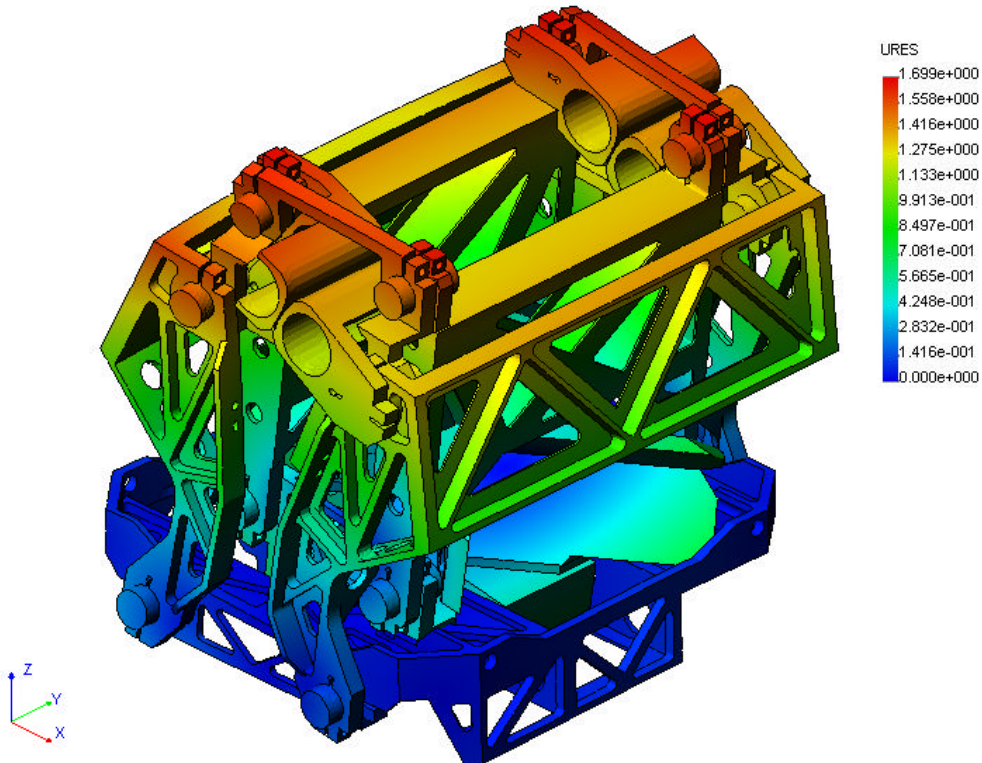
Mode: 3



Third Mode : 173,1 Hz – Vertical Torsion

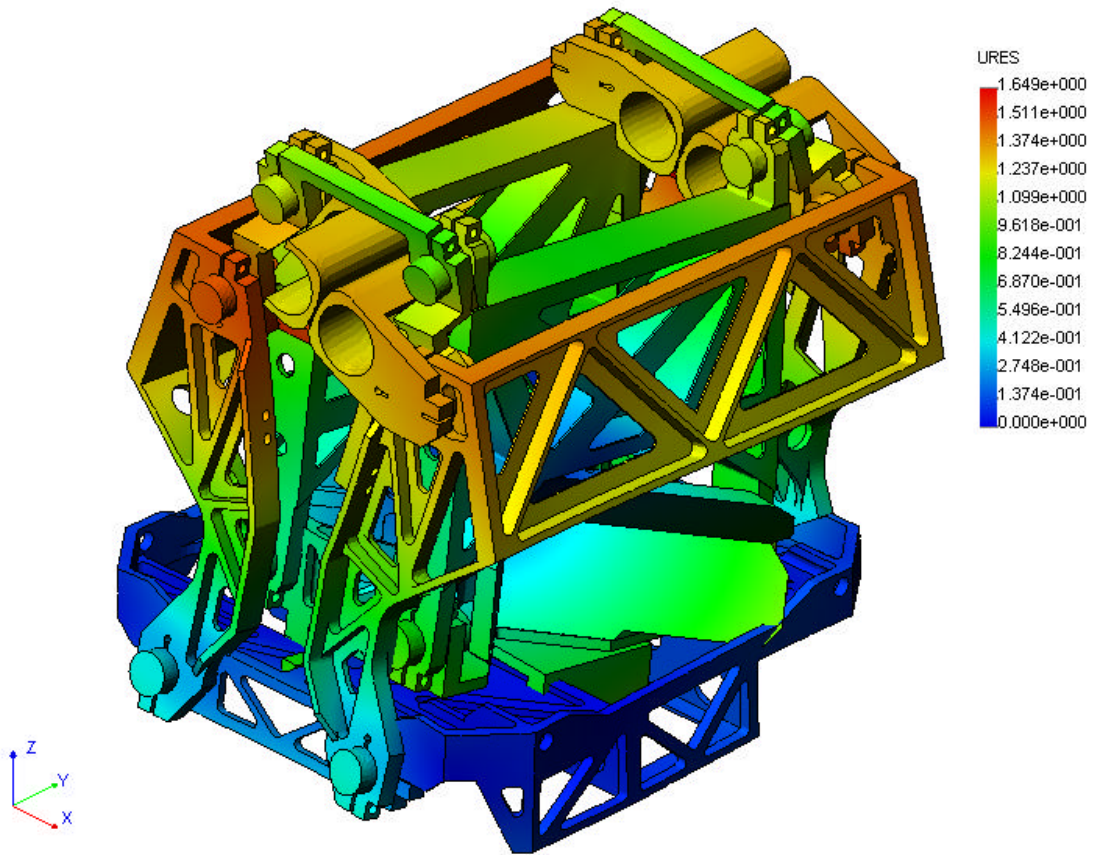
This Mode would be very sensitive for control, as it leads to Actuator / Optical sensor misadjustment (movements of the actuator could be unseen by sensor)

Mode: 4



Fourth Mode : 186,7 Hz -

Mode: 5



Fifth Mode : 281 Hz -