

**TITLE:** Design of the Telescope Simulator imaging mirror

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**CHANGE RECORD**

<b>ISSUE</b>	<b>SECTION</b>	<b>REASON FOR CHANGE</b>
<b>1.0</b>		First draft of the document (23/03/01).
<b>2.0</b>		Update of $R_{SM}$ and related mirror dimensions after SPIRE Telescope Simulator internal meeting at RAL (11/05/01) and addition of section 6.
<b>3.0</b>		Addition of section 5.3 on mechanical design and references updated (27/07/01).
<b>4.0</b>		Addition of section 7 and related Appendix A (30/11/01) and B.

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**APPLICABLE AND REFERENCE DOCUMENTS**

**RD1** Duncan et al., "Long wavelength (sub-mm) telescope simulator", Infrared Phys. 34-1 (1993)

**RD2** SPIRE-Telescope simulator optical design, SPIRE-RAL-NOT 000622 (08/05/2001)

**RD3** SPIRE-Telescope simulator requirements specification.

**RD4** Martin D.H, "Long-wave optics", IEEE Trans. MTT 41-10 (1993).

**RD5** SPIRE-Feedhorn focus positions, SPIRE-RAL-NOT 000566 (10/01/2001)

**RD6** Murphy J.A., "Distortion of a simple gaussian beam on reflection from off-axis ellipsoidal mirrors", Int. J. Inf. Mm. Waves 8-9 (1987).

**RD7** Mechanical drawings: "Mirror M2, Telescope Simulator" (ref. 1-KG-0720-500-B) and "M2 mirror mount assy"(ref. KE-0104-330)

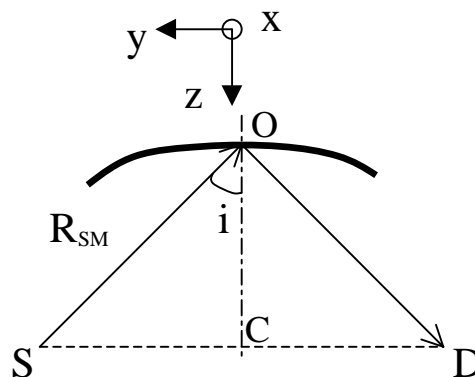
**RD8** SPIRE-Set-up and alignment procedure for the Telescope Simulator imaging mirror, SPIRE-RAL-NOT 000734 (02/07/2001)

### 1. Introduction

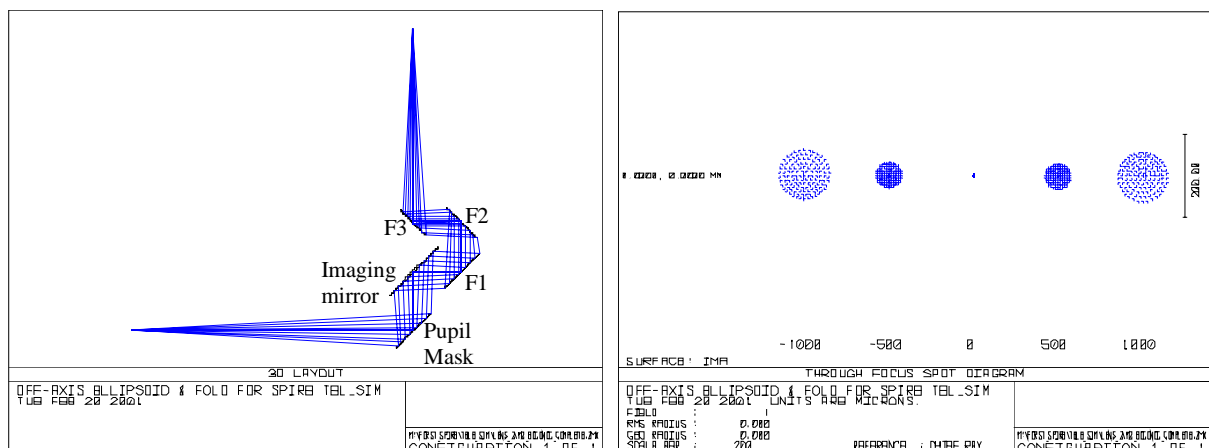
This note aims at presenting some practical solutions for the problem of 1:1 imaging between the source and the detector, giving recommendations for an adapted imaging mirror for the telescope simulator experiment, after taking into account the optical performances required as well as the manufacturing constraints. A list of specifications was derived leading to the manufacturing.

### 2. Case of a mirror with biconic (ellipsoidal/spherical) surface

In the description of a previous telescope simulator in far-infrared range (RD1), a spherical mirror was used off-axis. This leads to degradation of imaging performances as it introduces aberrations.



In order to have perfect imaging between the point source (S) and the detector (D), these points have to be located at the foci of an elliptically curved (in the plane yz) mirror. With an angle of incidence  $i$  for the incident beam at the mirror vertex O, the eccentricity  $e_y$  is given by  $e_y = \tan(i) = \sqrt{K_y}$  where  $K_y$  is the conic constant. In order to adapt the radius of curvature  $R_y$  to the wavefront radius of curvature given by the specified distance  $SO = R_{SM}$ , one needs to have  $R_y = R_{SM} * \sqrt{K_y + 1} = R_{SM} / \cos(i)$ . In the xz plane, the mirror can be just taken spherical ( $K_x = 0$ ) with its centre of curvature in C. Therefore the radius of curvature  $R_x$  is given by  $R_x = R_{SM} * \cos(i)$ . Applying these parameters at 45deg of incidence, and with  $R_{SM} = 1980.8 \text{ mm}^1$ , we obtain  $K_x = 0$ ,  $K_y = 1$ ,  $R_x = 1400.64 \text{ mm}$ ,  $R_y = 2801.29 \text{ mm}$ .



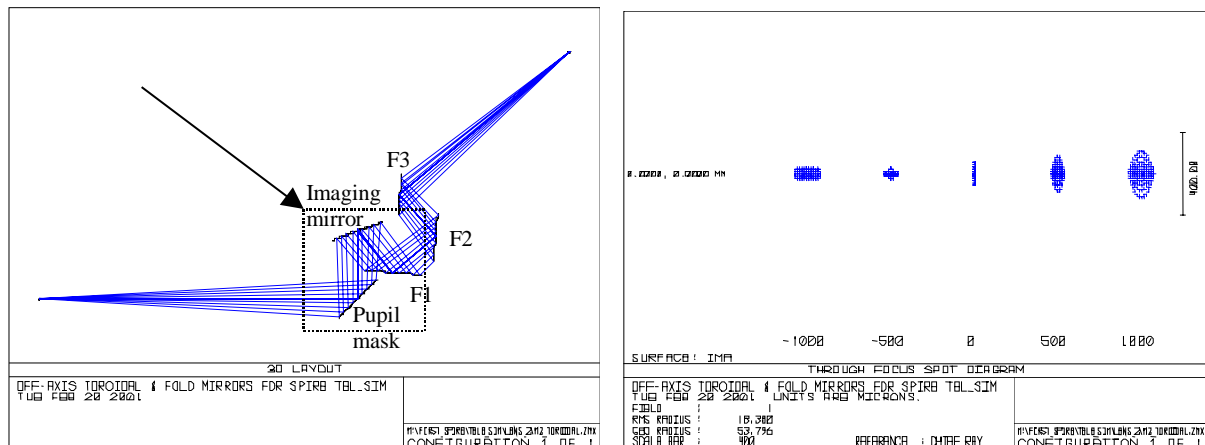
**Figure 1:** Zemax ray-tracing layout (yz plane) and spot diagrams around image surface for a test configuration

<sup>1</sup> This value for  $R_{SM}$  was determined from the study of constraints on the entire telescope simulator system layout (see RD2).

In Fig. 1, one can notice the quality of the image at the conjugate point of the point source. The displayed configuration uses the above values for the surface definition and  $R_{SM}$ . It also includes the mask (beam stop with an entrance pupil diameter set to  $\sim 181.1$  mm at 1573.5 mm from the source in order to simulate a system with  $F\# \sim 8.68$  as required in RD3), the imaging mirror under 45deg of incidence, and a set of 3 fold mirrors (F1, F2 and F3). Longitudinal defocusing by  $\pm 1$  mm gives rise to a spot size no larger than  $100 \mu\text{m}$  (rms radius).

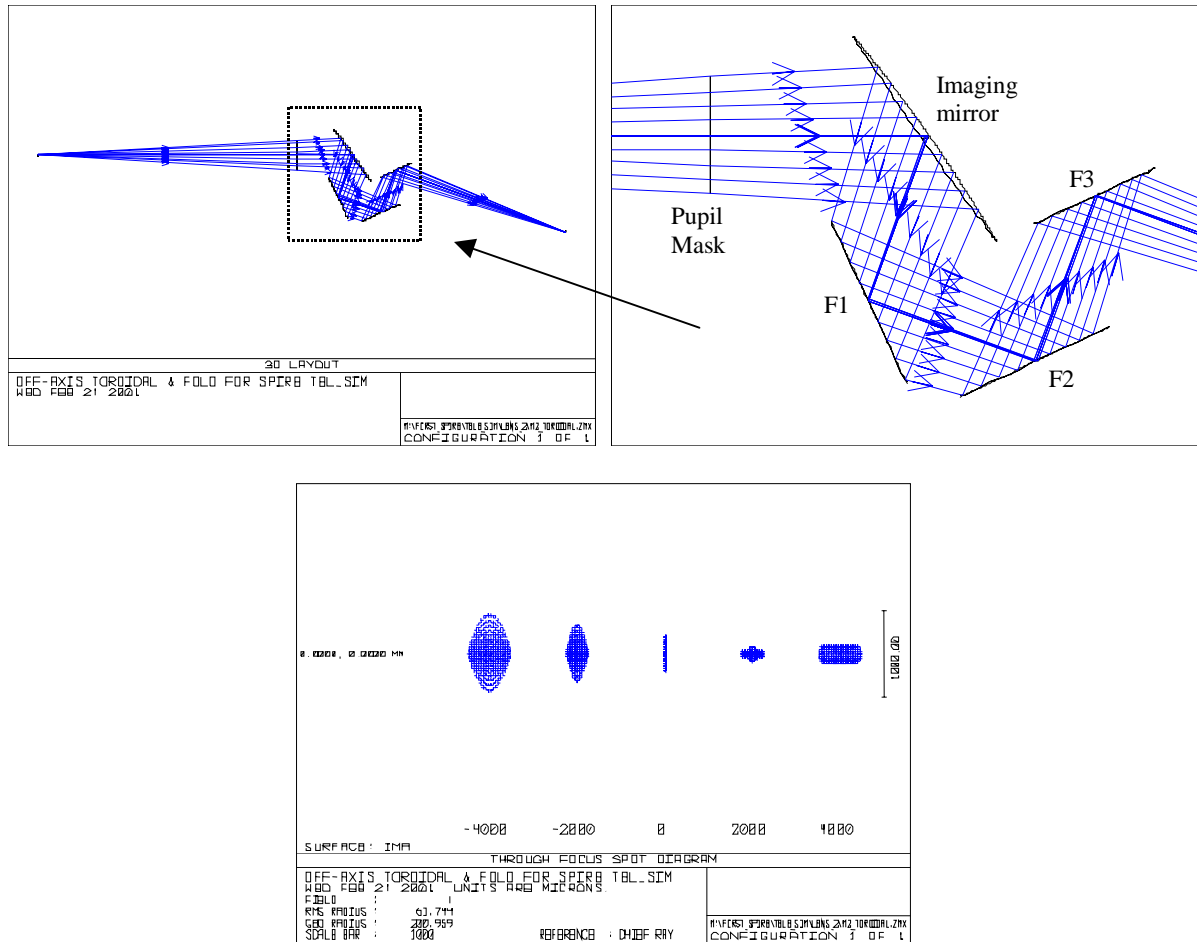
### 3. Case of a mirror with a toroidal surface

We consider now a surface having still 2 different radii of curvature  $R_x$  and  $R_y$  but spherical in both planes with  $K_x=K_y=0$ . The radii of curvature are still defined by the same definitions as above:  $R_x=R_{SM}*\cos(i)$  and  $R_y=R_{SM}/\cos(i)$ . On-axis (normal incidence,  $i=0$ ), the surface becomes simply spherical with  $R_x=R_y=R_{SM}$  giving perfect imaging of the source. But practically, one has to use the mirror off-axis, introducing aberrations partially, not entirely attenuated by the toroidal surface shape. With  $i=20\text{deg}$ ,  $R_x=1861.35$  mm and  $R_y=2107.93$  mm. Consequently, the beam size on the mirror is then reduced (extension to  $\sim 240$  mm in diameter) when compared to the 45deg case.



**Figure 2:** Zemax ray-tracing layout (yz plane) and spot diagrams for a second configuration ( $i=20\text{deg}$ )

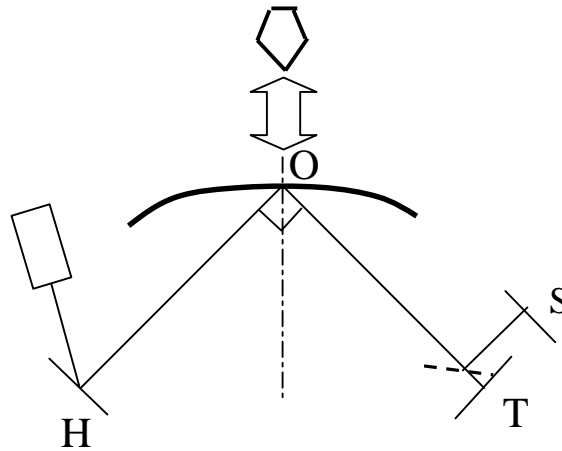
In Figure 2, the effects of aberrations are seen on the spot diagrams. At the conjugate point, the beam has a linear extension and a better focus point can be chosen within  $\sim 0.5$  mm max before the conjugate point position. Although including some aberrations, the image of the point source remains smaller than  $100 \mu\text{m}$  in that configuration. But the large radial size of the beam and the constraints on the mask-mirror and mirror-first fold mirror distances can produce practical problems such as the incident beam impinging on the mirror may be partially diffracted by the first fold mirror edge (see dashed box in Fig 2, left). Even if the mask is used in transmission, there is necessity to increase either the mirror-F1 distance (set to 300 mm here) or the rotation angle (about x) for the imaging mirror so as to separate the incident and the reflected beam of the imaging mirror. As the distance between the mirrors can not be changed by much ( $\pm 50$  mm max for the beam control), the angle  $i$  should be increased and consequently the mirror size as well. In Figure 3, a configuration with  $i=35\text{deg}$  is displayed (mask used in transmission) where it can be seen that the incident beam on the imaging mirror will still be close to the edge of the first fold mirror F1. Therefore due to beam size and system geometry constraints,  $i$  needs to be larger than  $35\text{deg}$  (and smaller than  $45\text{deg}$  to keep the whole experimental set-up in a reasonable surface area). But spot pattern with the toroidal mirror degrades rapidly as  $i$  is increased. At  $35\text{deg}$ , the best focus would be mostly located  $\sim 2\pm 1$  mm away (longitudinally shifted) along the optical path and its rms size would already reach  $\sim 200 \mu\text{m}$  (getting even larger at  $i=45\text{deg}$ ).



**Figure 3:** Zemax ray-tracing layout (yz plane) and spot diagrams for a configuration with  $i=35\text{deg}$

#### 4. Alignment consideration

The previous discussion shows the necessity to work with an off-axis angle of at least  $35\text{deg}$  and no more than  $45\text{deg}$ . An angle of  $45\text{deg}$  seems to be better suited as it can be set with the use of a pentaprism. This will replace the first concept of locating foci of the mirror illuminating it on-axis then rotating it as the mirror (any of the 2 shape considered above) demonstrates poor optical imaging properties on-axis (too “slow” optics). A more detailed (step by step approach) alignment plan is under investigation and just a first possible draft concept is described below.



The visible laser + plate with hole (H) and the flat transfer mirror (T) can be set with reference to the optical bench (main level reference for the plane of propagation) with an autocollimator.

The pentaprism is used to set a 90deg angle between the incident and reflected beam. Then the pentaprism is replaced by the mirror. Assuming existence of a mark locating the centre O of the mirror, the 3 points O, H and impact on T should be in the same plane. Verification could be made by use of a beam splitter + image surface to locate the second conjugate focus S (and tilt angle errors) after moving H along the path OH (measured distance; no need to it for OT) to the required  $R_{SM}$  (first focus point) distance (use of slide radius with expected accuracy smaller than 1mm). This required the visible laser signal to be strong enough (not too much scattered by mirror surface roughness and consequently there is a need for extra polishing around the centre of the mirror). Fine-tuning for the focus at detector plane could be made afterwards when inserting the 3 fold mirrors system that would allow final correction/adaptation of the second focus to detector. All this is actually placed under constrain of space made available (mainly determined by the chosen  $R_{SM}$  distance compared to the size of possible optical bench). Update and details of the procedure are in RD

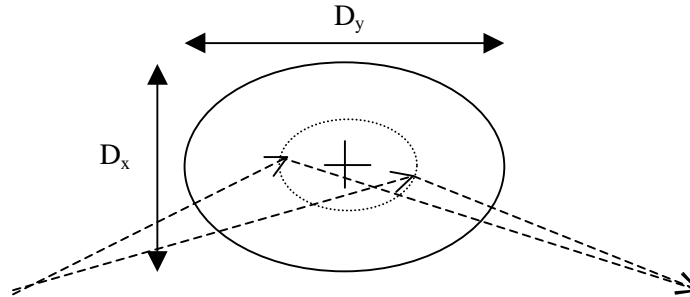
## 5. Specification of the mirror

From the considerations developed in the above paragraph, the case of an ellipsoidal mirror with a biconic surface seems to present to best optical properties with respect to the requirements and considering manufacturing difficulties which would be the same in both cases.

### 5.1. Mirror dimensions:

Mirror needs to be oversized with respect to the beam size on its surface to avoid loss by edge diffraction and spillover. A factor 1.2 (20% oversized) is applied as a compromise between a too large size difficult and costly to manufacture and minimising the introduction of diffraction losses.

On-axis the beam impact on the mirror would be circular with a diameter  $\sim 230$  mm (also given by  $\sim R_{SM}/F\#$ ). Therefore in the case presented in section 1, we have  $D_x=276$  mm and  $D_y=390$  mm (the dimension is longer along y due to the angle of incidence  $i=45$ deg in yz plane).



High precision is not particularly requested on these dimensions but would be driven by the manufacturing constraints<sup>2</sup>. The mirror thickness should be at least 10 mm (due to the surface definition, see below, and dimensions) at the edge but 10% of the mirror size is commonly used which leads in this case to ~30 to ~40 mm of thickness at the mirror aperture edge.

In order to reduce the mirror dimensions without changing the system parameters ( $R_{SM}$ ,  $i$ ), a 10% oversize factor only could be applied instead of 20% because the propagation is relatively well “beamed”. A few percent loss may be expected through beam spreading due to diffraction over the free space optical path, which would affect mainly the longest working wavelength, as the diffraction coefficients varies with  $\lambda^{1/2}$ . This may need to be confirmed by further specific coherent beam pattern calculations. With this oversize smaller factor, the dimensions become  $D_x=252$  mm and  $D_y=356$  mm. These two sets of values would represent limits on the mirror dimensions for this value of  $R_{SM}$ . For lower values of  $R_{SM}$ , the mirror dimensions would decrease also as a good first-order estimation of the beam diameter of the mirror is given by  $R_{SM}/F\#$ .

### 5.2. Surface definition:

The sag  $z$  of the biconic surface is given the following equation:

$$z = \frac{c_x x^2 + c_y y^2}{1 + \sqrt{1 - (1 + K_x)c_x^2 x^2 - (1 + K_y)c_y^2 y^2}}$$

with  $c_x=1/R_x$  and  $c_y=1/R_y$  and  $K_x=0$ ,  $K_y=1$ ,  $R_x=1400.64$  mm and  $R_y=2801.29$  mm. Tolerancing with Zemax showed that  $R_x$  and  $R_y$  should vary by no more than a few mm from these nominal values in order to keep a rms spot size radius smaller than 0.1 mm at the image point (particularly  $R_y$ ). For the surface roughness, a max rms value taken as  $\lambda/20$  can be used. With the smallest working wavelength at ~200  $\mu\text{m}$ , that leads to ~10  $\mu\text{m}$  rms which is well achievable even over the entire large surface area. For alignment purposes, the central part of the mirror (disk of a few cm in diameter around the middle) needs to be polished afterwards and should have a mark indicating the centre position. Cracks and surface defects may still remain after the polishing phase, and their sizes can cause some degradation (by surface scattering) of an alignment laser beam in the visible.

Different manufacturers have been approached and among them, Thomas Keating Ltd would be able to machine such a large optical component with the above specification. An expected delivery time was estimated to 8 to 10 weeks.

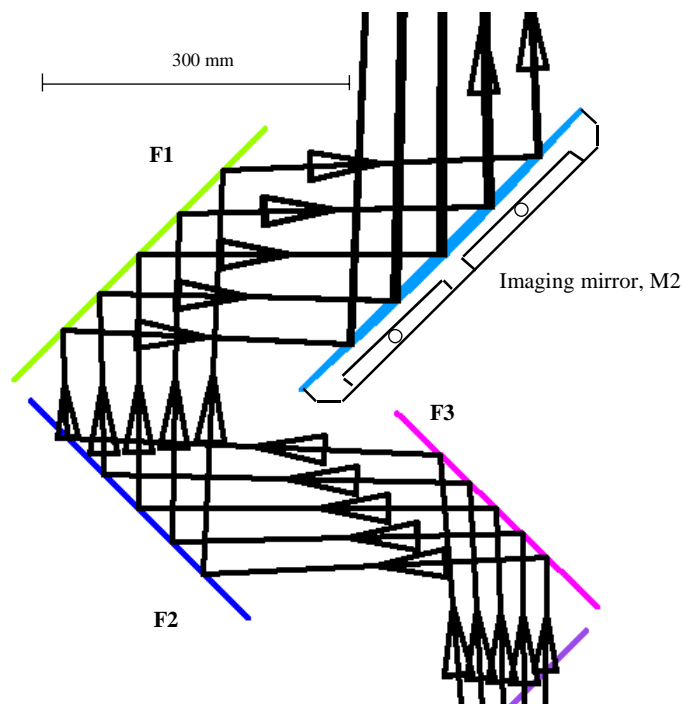
<sup>2</sup> The material is likely to be aluminium (lightweight, no corrosion) which leads to an approximate maximum weight of 6-7 kg for the mirror (excluding any support/mounting). With this metallic material (aluminium), a fractional power loss per reflection would be around 0.3% at  $\lambda \sim 300 \mu\text{m}$ .

### 5.3 Mechanical design:

From the above requirements and specifications, a mechanical drawing of the mirror was made. A PDF file of it can be found under the ref. RD7. A brief review of the drawing features is done here.

Apart from the front surface of the mirror based on the optical requirements, the back of the imaging mirror is designed from the following ideas (illustrated in Figure 4 below):

- the thickness should be at least ~10% of the longest dimension but should avoid interaction with rotating mirror F3,
- shape of the back should leave a clear zone for free-space beam propagation between F2 and F3 (inducing removal of the edge corners),
- a thick base (lowering the centre of gravity) is needed for stability in vertical position,



**Figure 4:** Zoom into the Telescope Simulator (ASAP model, optical layout plan view) with reverse (i.e. from simulated TFP) ray tracing in SPIRE FOV scan mode (large beam). M2 as from mechanical design is shown.

Extrusion of material in the back keeps the total weight less than about 15 kg. Bottom horizontal surface and lateral vertical ones are set to be reference for parallelism with the optical axis. They are designed within a tolerance limit obtained from alignment consideration (see RD8).

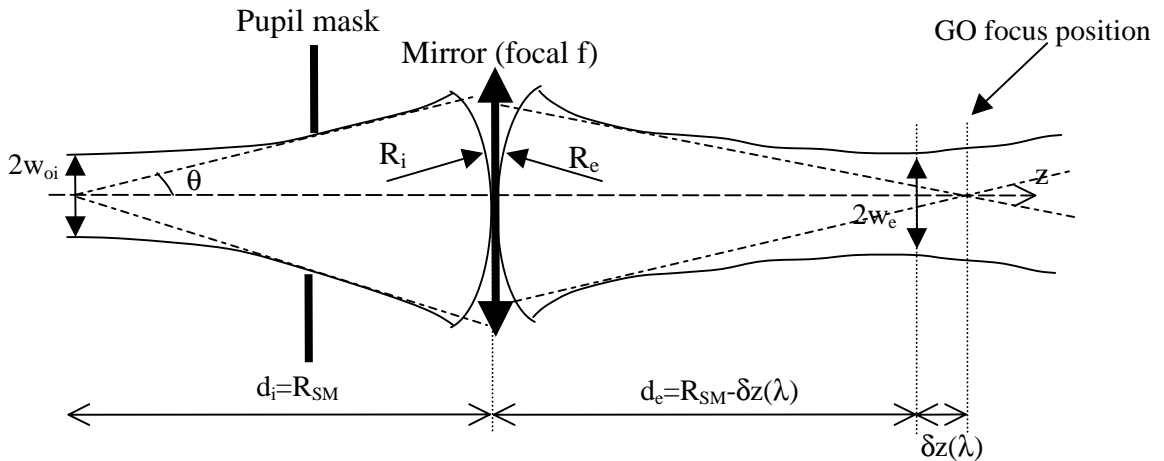
The elliptical aperture of the optical surface is enclosed in a larger (by ~4-5mm at the ellipse extremities) rectangular face (with final dimensions 285mm×400mm, 50mm maximum in thickness). This extra material is meant to be a margin allowing eventual re-machining of the surface (removing more material than expected) if not acceptable at delivery. Verification of the surface shape can be performed by metrology measurements of the surface profile along X=0 and Y=0 axis, using bottom and lateral surfaces as references. These measurements will be limited in precision by the surface roughness (~10 μm) but should lead to the best-fitted profiles of the final shape along both planes of symmetries to a good accuracy. Parameters like the radii of curvature  $R_x$  and  $R_y$  can then be retrieved as well as the real position of the centre of the surface (possible deviation from the fiducial mark location) and the optical axis direction with respect to reference surfaces.

A dedicated mounting system has been designed for this mirror (see Appendix B and RD8).



### 6. First-order (i.e. fundamental mode) Gaussian beam analysis

One of the foreseen system sources positioned at the front focus is a FIR laser. Such a source could be modelled as a provider of a near-gaussian beam with a finite size at the focus. Following long-wavelength (gaussian beam) optics formulation in RD4, one can define a gaussian beam waist located at the front focus of the ellipsoidal mirror. From GO consideration (relatively large F# so slow beam), the mirror can be modelled in a first order approximation as an on-axis lens-type perfect phase transformer (see figure below).



From GO analysis, a point source located at one of the ellipsoidal mirror foci (at  $R_{SM}$  from the mirror centre in this case) would be (perfectly) imaged at the conjugate focus, located at  $R_{SM}$  after the mirror. Extension of the analysis into Gaussian optics allows taking into account the effect of finite (long) wavelength. An incoming beam with a waist  $w_{oi}$  located at  $d_i$  from the mirror would give rise to a wavefront radius of curvature (ROC)  $R_i$  when reaching the mirror, with  $R_i$  given by:

$$R_i = d_i \left( 1 + \left( \frac{z_0}{d_i} \right)^2 \right) = d_i + \left( \frac{\lambda^2}{\pi^2 \theta^4 d_i} \right) \text{ where the confocal distance } z_0 \text{ is: } z_0 = \frac{\pi w_{oi}^2}{\lambda} = \frac{\lambda}{\pi \theta^2}$$

One can see that the ROC from a diffractive gaussian beam would differ by an additive term increasing with the wavelength from the expected GO spherical wavefront with ROC  $d_i$ . Applying the lens equation  $1/f = (1/R_i) - (1/R_e)$  to obtain the emerging wavefront ROC from the mirror, and developing  $R_e$  in the same as above as a function of  $d_e$ , would lead to a second-order equation in  $d_e$ . This result can be directly obtained from the transfer matrix (see RD4) between the initial beam size (radius  $w_{oi}$ ) and divergence  $\theta$  at  $d_i$  before the mirror to a plane at  $d_e$  after the mirror. In this case, one can directly get:

$$d_e = f + \frac{f^2 (d_i - f)}{(f - d_i)^2 + \left( \frac{\pi w_{oi}^2}{\lambda} \right)^2}$$

Using the system parameters  $d_i = R_{SM}$  and  $f = R_{SM}/2$ ,  $d_e$  can be expressed in the following way:

$$d_e = \frac{R_{SM}}{2} \left( 1 + \frac{1}{1 + \left( \frac{2\lambda}{\pi \theta^2 R_{SM}} \right)^2} \right) \approx R_{SM} - \delta z(\lambda)$$

where from first-order series development the wavelength-dependent defocus term is given by:

$$\delta z(\lambda) = \frac{2\lambda^2}{\pi^2 \theta^4 R_{SM}}$$

With  $R_{SM}=1980.81\text{mm}$  (i.e. the design value) and an initial beam divergence<sup>3</sup>  $\theta \sim 1/2F\#$  (with  $F\#=8.68 \Rightarrow \theta=0.0572\text{rad}$ ), this longitudinal shift will be:

$\lambda$ ( $\mu\text{m}$ )	$\delta z$ (mm) with $\theta=0.0572\text{rad}$	$\delta z$ (mm) with $\theta=0.051\text{rad}$
250	0.57	0.91
500	2.30	3.70

It should be noted that this effect would occur for the Herschel telescope and lead to focus re-positioning of the SPIRE detector feedhorns for optimum coupling (see details in RD5). In order to compensate for this effect at the telescope simulator level, the system source could be shifted longitudinally depending on the working frequency. Practically, the compensation can be achieved after the imaging mirror by moving together the mirrors F1 and F2. Such changes in the optical path length have already been studied in the focus control of the telescope simulator control laws where corrections of the order of a millimetre should be made during scan of the field-of-view (see RD2).

The large off-axis angle ( $i=45\text{deg}$ ) could be a source of beam distortion as studied in RD6. From an incident fundamental-mode-gaussian-beam (FMGB), the emergent distorted beam is given in a first-order approximation by the following correction from an ideal emergent gaussian beam:

$$E_e(x_e, y_e, z_e = d_e) \propto \left( 1 - \frac{\tan(i)w_e}{f} \left( \frac{x_e}{w_e} - \left( \frac{x_e}{w_e} \right)^3 - \left( \frac{x_e}{w_e} \right) \left( \frac{y_e}{w_e} \right)^2 \right) \right) \cdot E_{e\_ideal}(x_e, y_e, z_e = d_e)$$

with the beam radius at  $d_e$  given by:

$$w_e \approx w_{oi} \left( 1 - \frac{2\lambda^2}{\pi^2 \theta^4 R_{SM}^2} \right) = \frac{\lambda}{\pi\theta} \left( 1 - \frac{\delta z(\lambda)}{R_{SM}} \right),$$

which is here very close to the input waist size. A graphical illustration of it is displayed on the figure below.

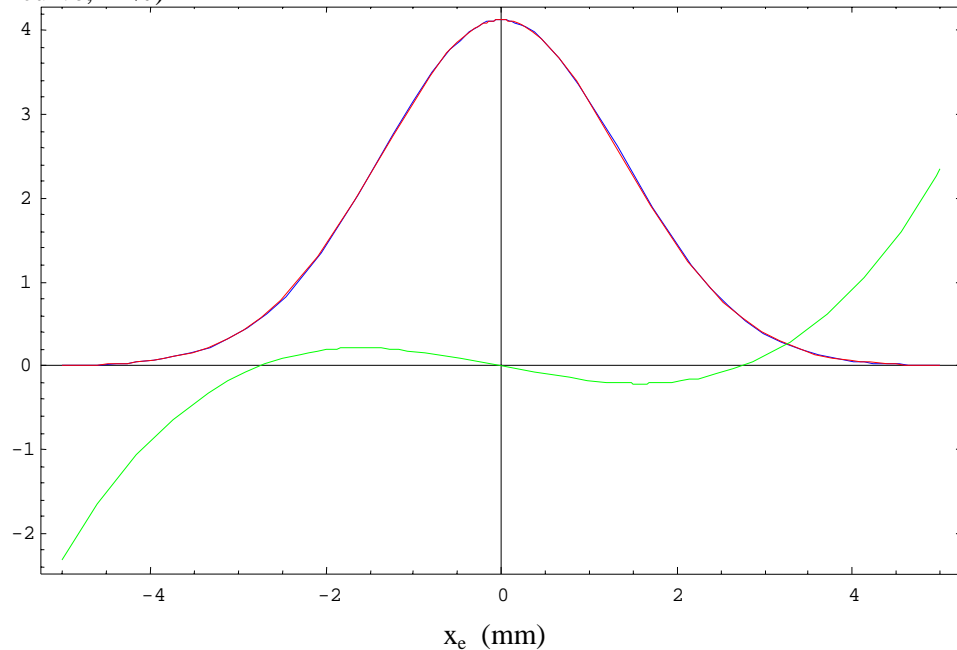
The loss of power from the fundamental is thought to transfer into higher-order modes and is estimated

$$\text{by } (1 - \eta) \approx \frac{1}{8} \tan^2(i) \left( \frac{w_m}{f} \right)^2 \sim \frac{1}{8F\#^2}$$

leading to value  $<0.2\%$  for the present case (and  $w_m$  being the beam radius on the mirror). For the case of an incident polarised beam, twice this amount is thought to be transferred into cross-polar component after reflection on the off-axis mirror.

<sup>3</sup> One can see that  $\delta z$  is strongly dependent on  $\theta$ . Here the value comes from GO consideration of rays limitation by the pupil mask dimensions and leads to an edge taper of  $-8.68\text{dB}$  (intensity level at  $1/e^2$ ) at the pupil mask aperture edge. In RD1, a value of  $0.051\text{rad}$  was used from Herschel telescope geometry and gaussian beam optics which would lead here to a larger magnitude of the edge taper (actually  $\sim -10.91\text{dB}$ ) and  $\sim 60\%$  longer defocus  $\delta z$ . This value of  $\theta=0.051\text{rad}$  is derived from a model of optimum coupling (overlap integral calculation) between a FMGB and an uniformly distributed field ("top-hat") across a circular aperture (the system pupil, i.e. Herschel secondary, or equivalently, in the Telescope Simulator case, the pupil mask). If the ratio "aperture radius/waist" is about 1.121, an optimum power coupling of 81.5% can be achieved in theory. Over-illumination of the pupil mask ( $\theta > 1/2F\#$ ) would bring the opposite effect. But these above values in  $\delta z$  are expected to be a worst-case estimation, as beam truncation by the pupil mask aperture would bring the global system behaviour closer to GO results.

Beam intensities (arbitrary units, blue and red curves) and relative difference (green curve, in %)



**Figure 5:** Intensity for an ideal gaussian beam (blue), a real distorted beam (red) and their relative difference in % (green) as function of  $x_e$  (in mm) in the plane  $y_e=0, z_e=d_e$ .

All these quantitative estimations show effects with quite small magnitude. Deviations, other than the ones due to above approximation, may arise from beam clipping effect at the pupil mask and real field distribution at the source (higher-order gaussian modes for example).

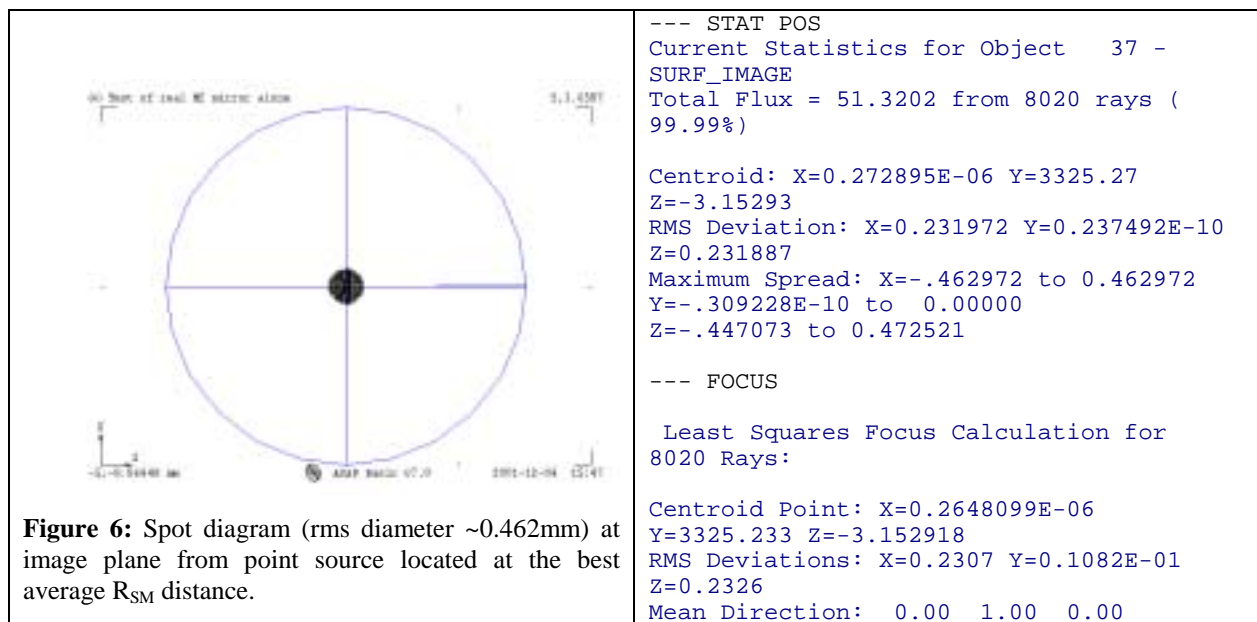
## 7. Metrology measurements on the manufactured mirror

After manufacturing, the imaging mirror M2 has been delivered to RAL. This was immediately followed by metrology measurement of the optical surface to the Inspection Facility (RAL/ Eng. Dpt.), between 27/11/2001 and 29/11/2001. From the measurements, and following a previously tested procedure<sup>4</sup> (see Appendix A), the optical parameters (radii-of-curvature  $R_x$  and  $R_y$ ) of the surface can be accurately retrieved by using a numerical non-linear fitting scheme (with or without a priori knowledge of the conic constants  $K_x$  and  $K_y$ ) on the collected data. The results of the metrology data reduction for the different probes sizes are summarised in a table in Appendix A4. Below is a summary of the results:

- The general dimensions meet the specifications of the CAD drawing in RD7;

<sup>4</sup> Preliminary tests were first performed, in order to assess the precision of the metrology measurements and the numerical parameter retrieval method, on a small shaped metallic mirror borrowed from the RAL/MMT group.

- The surface treatment (polishing) over the optical surface has been found well uniform although further qualitative tests in the visible range may be conducted in order to evaluate the scattering loss on a visible laser point source located at the  $R_{SM}$  distance, away from the mirror;
- The fiducial mark is precisely cut and has been found to be located in the right position on the surface (within  $\sim 0.2\text{-}0.3$  mm max of the ideal surface centre);
- The biconic surface was designed with 2 conic constants along the 2 main axis direction (Y and X) that can not be retrieved with high precision (Table 1 in Appendix A4) but the associated radii of curvature were retrieved with accuracy better than 0.1%. Assuming the conic constants to follow the design (Table 2 in Appendix A4), it was found that these 2 radii (and the  $R_{SM}$  mirror-to-foci distance as well) were out-of-spec. This brings a relative difference between the previously specified (see RD2) and the best average  $R_{SM}$  value of  $\sim 0.55\%$  (i.e.,  $\sim 10.9$  mm). Further optical analysis of the impact of this difference on M2 imaging performances (see below) shows that a point source located at the modified distance from the mirror gives rise to a spot diagram with rms diameter still between 1/10 and 1/5 of the Airy disk diameter at  $\lambda=200\mu\text{m}$  (with a geometric Strehl ratio still maintained  $>0.95$  for  $\pm 1$ mm tolerances field positions).



A NCR document, with minor level, has been set for the reason stated above (focal length slightly out-of spec).

- The angular orientation of the surface (surface principal axis) remains well within the specified  $\pm 1$  arcmin with respect to lateral references surfaces.

### Appendix A. Procedure for surface quality control via metrology measurements on the optical surface of the M2 imaging mirror

#### A1. Introduction

For the purpose of the test and calibration, at RAL, of the SPIRE instrument (for the future Herschel Space Observatory), a Telescope Simulator was designed (see RD2). Among the quasi-optical elements composing this long-wavelength instrument, there is a large metallic (aluminium) ellipsoidal mirror designed for point-to-point imaging (see previous section of this note).

The shape of the mirror optical surface determines its function and this one has been specified to follow the sag equation below:

$$z(x, y) = \frac{c_x x^2 + c_y y^2}{1 + \sqrt{1 - (1 + K_x)c_x^2 x^2 - (1 + K_y)c_y^2 y^2}}$$

where  $c_x$  and  $c_y$  are the surface curvatures in each

respective transverse co-ordinate direction, and  $K_x$  and  $K_y$  are the respective associated conic constants. From the design, we have:  $K_x=0$ ,  $K_y=1$ ,  $c_x=1/1400.64$  and  $c_y=1/2801.29$  (all lengths are expressed in mm.)

Other specifications for the optical surface include:

- surface roughness, form factor: ~ 5-10  $\mu\text{m}$  rms,
- extra polishing of the central region,
- fiducial mark (small cross) at the expected centre of the mirror surface.

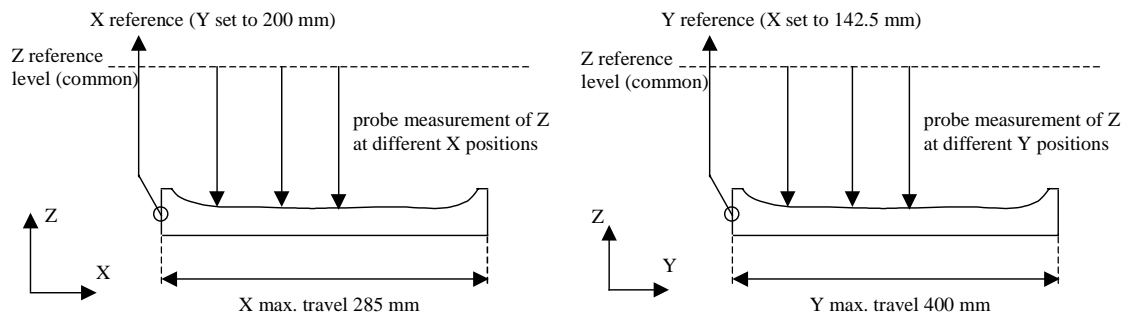
The mirror lateral surfaces, horizontal and vertical (see CAD drawing in RD7), are specified as references for the surface optical axis to within  $\pm 1$  arcmin.

The mirror M2 fits into a volume 400mm $\times$ 285mm $\times$ 50mm and will be manufactured by an external company.

#### A2. Metrology measurements

In order to check the quality of the manufacturing, some measurements would need to be performed on the mirror surface afterwards. It is suggested that 2 types of measurements should be done:

- 1) - Measurement of the surface shape profile (elevation or depth co-ordinate Z via contact probe) along 2 transverse directions (i.e. namely X and Y) respectively referenced to the mirror lateral surfaces (see figure below).

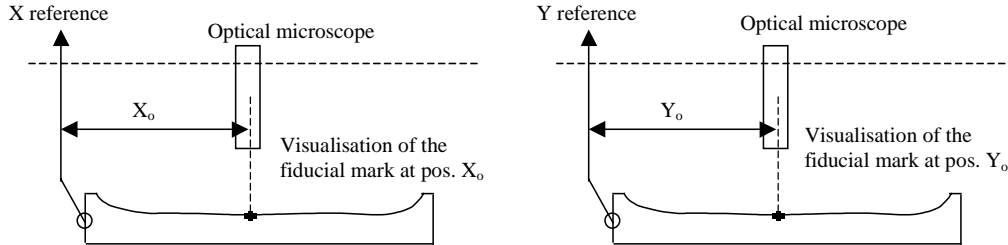


**Figure A1:** schematic of mirror surface shape estimation along 2 cuts via measurement probe.

From the surface definition above, it can be seen that steps of 5 mm in X and Y would lead to a (minimum, at the centre) elevation change in Z between consecutive measurements of ~9 and ~5  $\mu\text{m}$  respectively, which is of the order of the expected surface roughness/form factor. Therefore a finer sampling (smaller steps) would not bring more information<sup>5</sup> because the “signal” obtained would be dominated (assuming the use of measurement machine with precision smaller than  $\pm 10\mu\text{m}$ ) by the surface roughness “noise”. Suggested values for transverse step movements in X and Y would be respectively 5 mm and 8 mm. Considering the dimensions of the mirror front surface, that leads to ~50 points along the surface shape profile in y and ~55 points in x.

<sup>5</sup> Although if the variations due to the surface roughness are completely randomly distributed, more measurements points (smaller step size) would be still relevant (improving statistics) to the overall profile shape. On the other hand, due to the polished central zone, it may be likely that different surface roughness distribution will be encountered during surface profile measurements.

2) – Optical (non-contact) location measurement of the fiducial mark: such a mark is specified in the design and it needs to be checked if it is actually located at the surface centre in X and Y (see figure below).



**Figure A2:** schematic of the optical location of the fiducial mark

From the profiles measurement, an estimation of the centre location can be derived and compared to the fiducial mark position. The fiducial mark will be used as visual reference for the mirror alignment and so knowledge of the eventual corrections (optical displacements) in order to reach the best-estimated location (see below) for the real surface centre. The values  $X_o$  and  $Y_o$  should be obtained with accuracy  $\sim \pm 0.1\text{mm}$  max (TBC).

### A3. Retrieval of the mirror surface optical parameters

From optical tolerancing analysis and in order, for the mirror, to meet the system optical requirements, the optical surface parameters should not depart from the initial design values by more than  $\sim \pm 0.2\%$  for the radii of curvature (especially for  $R_y$  i.e. in the plane of the elliptical shape).

From the designed mirror sag equation, the measured profiles can be fitted with the following functions:

$$z(x, y = 0) = \frac{c_x x^2}{1 + \sqrt{1 - (1 + K_x)c_x^2 x^2}} \text{ with parameters } c_x \text{ and } K_x \text{ (and eventually } z_0 \text{ and } x_0),$$

$$z(x = 0, y) = \frac{c_y y^2}{1 + \sqrt{1 - (1 + K_y)c_y^2 y^2}} \text{ with parameters } c_y \text{ and } K_y \text{ (and eventually } z_0 \text{ and } y_0)$$

This method has the advantage of avoiding a unique fit with all the parameters (therefore reducing the degrees of freedom and constraining the fit procedure for better convergence). The main drawback is that identical behaviour for the rest of mirror surface has to be assumed i.e. there is no “bump” or any large deviation from the fitted surface shape in the non-probed quadrants of the mirror surface.

Some variants of the fitting procedure can be investigated to reduce the error residue. For example,  $K_x$  and  $K_y$  can be assumed to be at the designed values (0 and 1 respectively) and then only the radii of curvature are derived from the fits (or the opposite: fixing the curvatures and fitting the conic constants only).

Some preliminary tests of the fitting procedure have been performed on ideal surface shape profiles (with sampling as indicated above) with a  $\pm 10\mu\text{m}$  magnitude random perturbation to account for the surface roughness/form factor. Results show that the conic constants can not be retrieved, a priori, with accuracy better than around few percents<sup>6</sup>. Quick tolerancing was performed and no consequence related to degradation of the mirror optical performances was noticed when the conic constants were modified by  $\sim 0.05$  to  $0.15$ .

Once the obtained best fit estimation of the location in x and y of the centre can be done and compared to measured value for the fiducial mark. Good retrieval of  $x_0$  and  $y_0$  were always obtained which implies that eventual differences between the surface centre and the fiducial mark should be known to at least  $0.1\text{mm}$  or better. Derivation of the normal to the surface shape profile, especially at the position of the best estimate for the centre, can give access to the optical axis direction. As established in the alignment procedure (RD8) and shown on the mirror mechanical drawing, the optical axis should deviate from the ideal optical axis (with respect to the lateral surfaces references) be no more than  $\pm 1\text{arcmin}$ . This later value can be also used as complementary criterion for post-manufacturing surface quality control.

<sup>6</sup> However this was obtained by averaging after several ( $\sim 10$  to  $20$ ) fits with different sets of “noisy” data representing different possible sets of measurements for one profile and the variance was still very large (example of a test case for the profile in X where only  $K_x$  was fitted leads to a value of  $0.035$ , instead of the designed value of  $0$ , but only after averaging over  $20$  “statistical events” and with a variance of  $0.016$ ).

**SPIRE**

**Technical Note**

**Ref:** SPIRE-RAL-NOT-

000 621

**Issue:** 4.0

**Date:** 03-12-2001

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### A4. Results of the fitting procedures and M2 parameters retrieval

Table 1	Rx	CI Rx	Kx	CI Kx	Ry	CI Ry	Ky	CI Ky	RMS	xo	yo	zo	theta x	theta y
Probe 1mm	1389.57	0.275	-0.2043	0.1	2788.67	1.62	-0.1444	0.56	1	-142.4	-199.7	-6.93	-0.28	-0.42
					2788.28	1.4	-0.159	0.47	3.2	-199.7	-6.93	-0.42		
Probe 2mm	1389.21	0.27	-0.3236	0.1	2788.28	1.4	-0.159	0.47	1	-142.4	-199.7	-6.93	-0.33	-0.42
					2789.24	1.42	0.233	0.49	2.9	-199.7	-6.93	-0.42		
Probe 3mm	1389.25	0.29	-0.2487	0.1	2789.24	1.42	0.233	0.49	1	-142.4	-199.7	-6.93	-0.27	-0.43
					2789.24	1.42	0.233	0.49	2.8	-199.7	-6.93	-0.43		

Table 2	Rx	CI Rx	Kx	CI Kx	Ry	CI Ry	Ky	CI Ky	RMS	xo	yo	zo	theta x	theta y
Probe 1mm	1390.13	0.1	0	N/A	2791.86	0.5	1	N/A	1	-142.4	-199.7	-6.93	N/A	N/A
					2791.48	0.4	1	N/A	3.3	-199.7	-6.93	N/A		
Probe 2mm	1390.11	0.1	0	N/A	2791.48	0.4	1	N/A	1	-142.4	-199.7	-6.93	N/A	N/A
					2791.39	0.4	1	N/A	2.9	-199.7	-6.92	N/A		
Probe 3mm	1389.94	0.2	0	N/A	2791.39	0.4	1	N/A	1	-142.4	-199.7	-6.93	N/A	N/A
					2791.39	0.4	1	N/A	2.9	-199.7	-6.93	N/A		

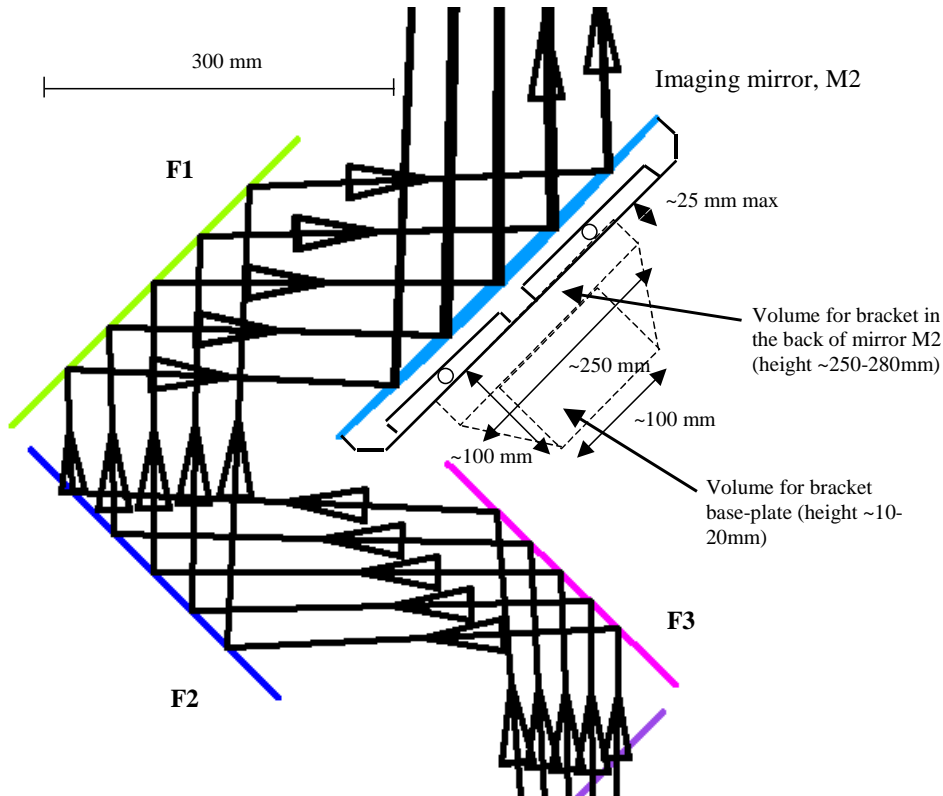
Rx average	<b>1390.06</b>	Ry	<b>2791.58</b>		
Rx spec. diff. (mm.)	<b>1400.64</b> 10.58	average Ry spec. diff. (mm.)	<b>2801.29</b> 9.71		
relative err RSMx	0.76%	relative err RSMy	0.35%		
diff RSM	<b>1965.84</b> 14.97	diff RSM	<b>1973.94</b> 6.87	RSM spec.	<b>1980.81</b>
				average diff. RSM	<b>10.92</b>
				rel. err. diff. RSM	<b>0.55%</b>
				best average RSM	<b>1969.89</b>

Table 1 is for results of the fit of all parameters. In Table 2, Kx and Ky were assumed to be the specified values. For each probe size, 2 sets of values are derived from the 2 profile scans in X and Y performed on the mirror optical surface.

**NB:** RMS is expressed in  $\mu\text{m}$ , thetax and thetay in arcmin. All the others quantities are in mm expect the dimensionless conic constant Kx and Ky. CI stands for the confidence interval of the associated quantities.

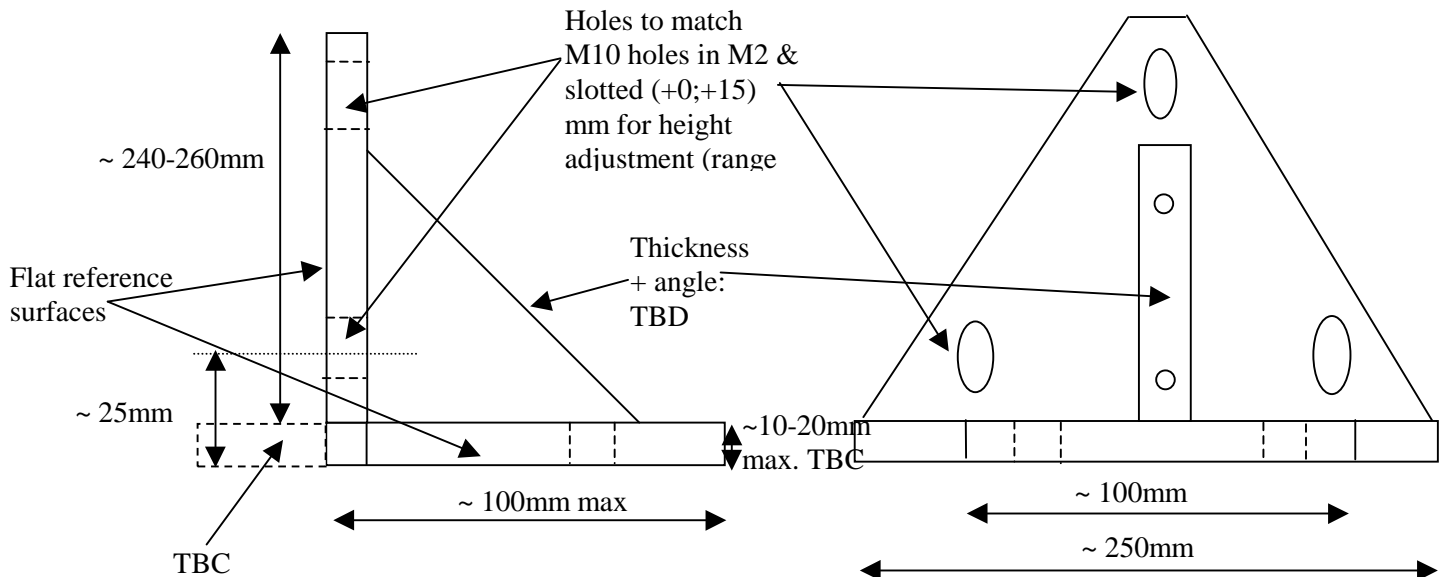
### Appendix B. Design of the M2 height-adjustable bracket system.

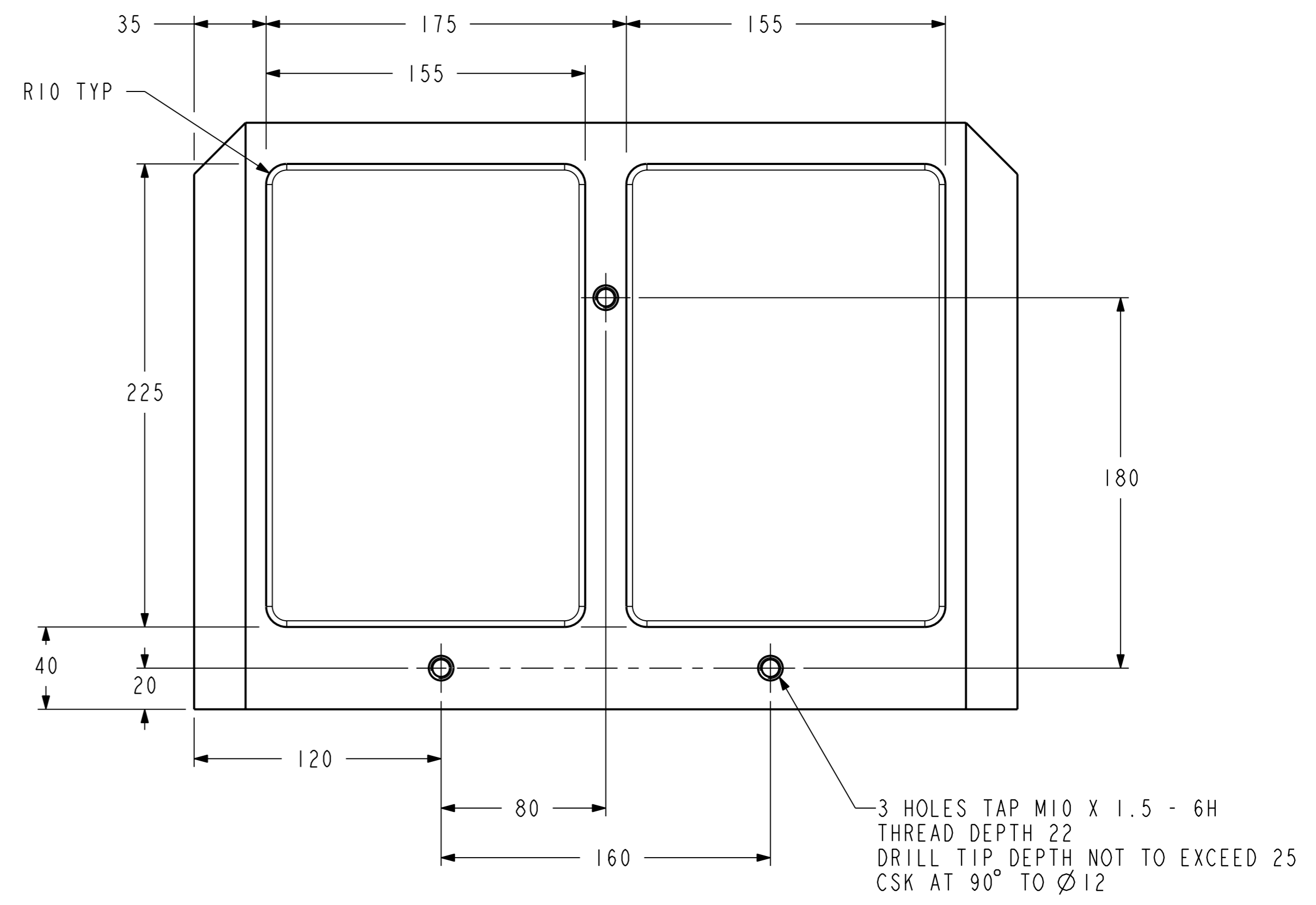
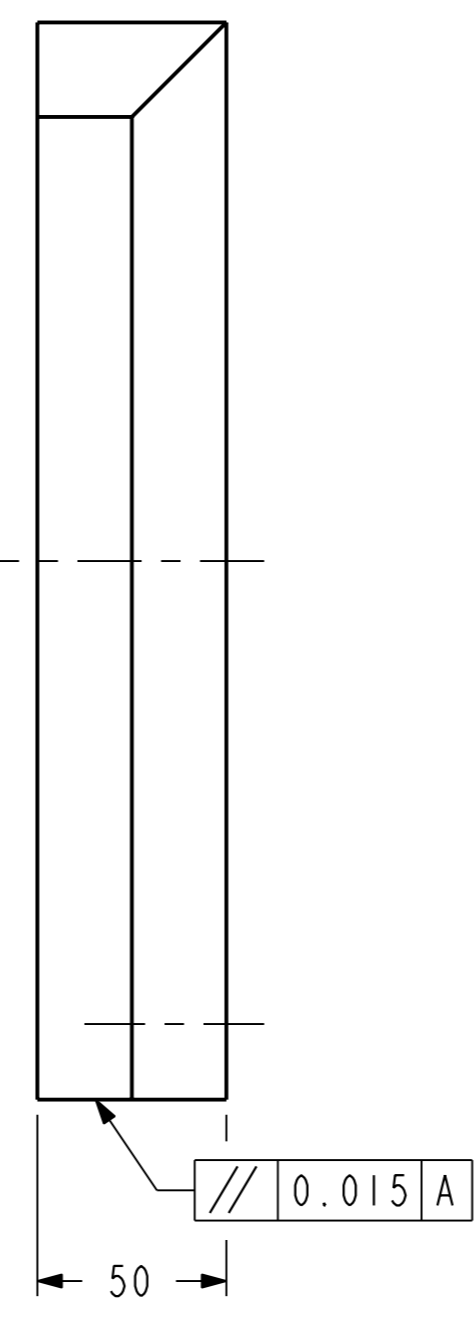
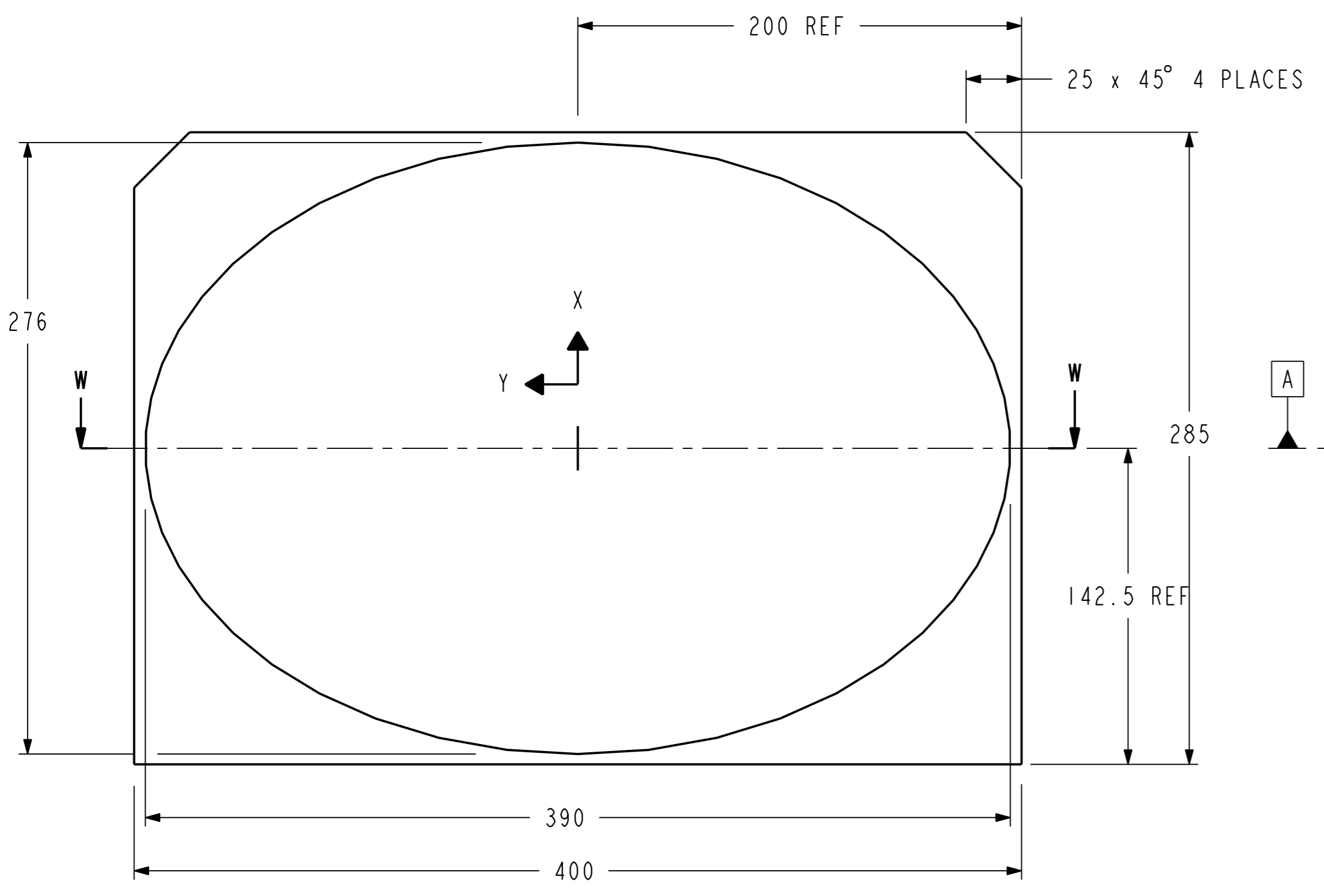
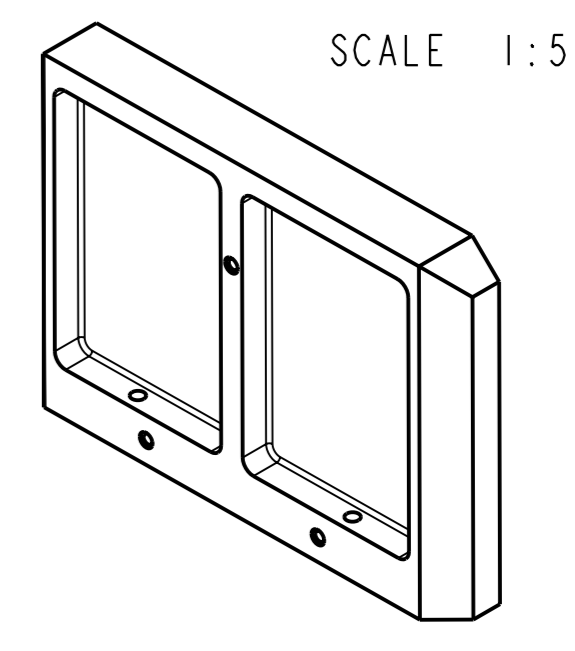
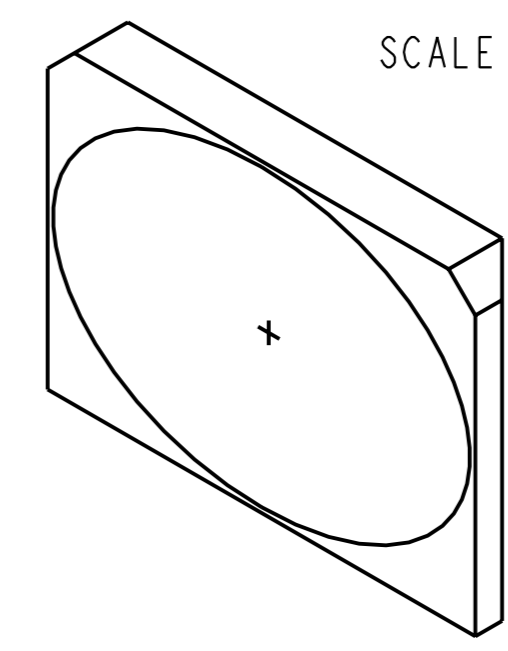
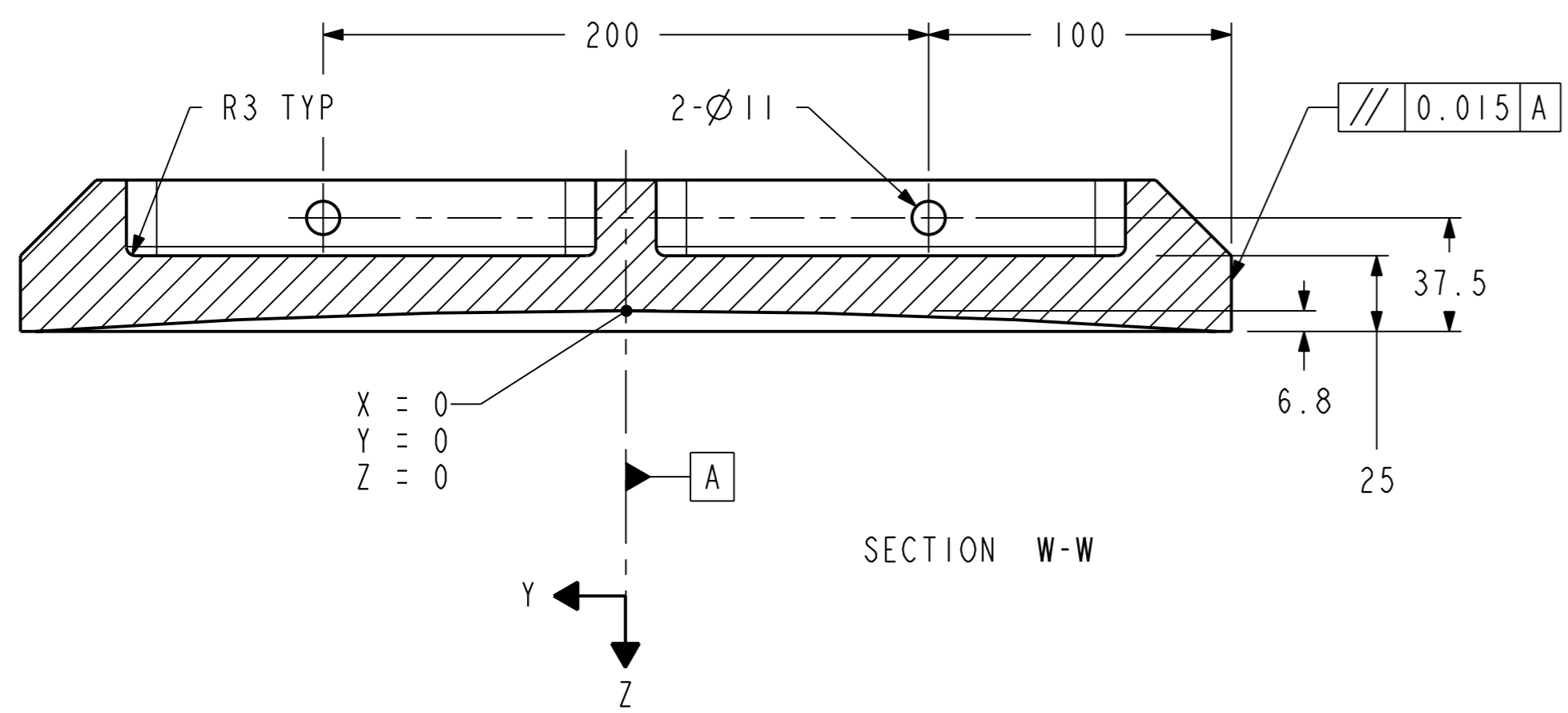
The purpose is to obtain a mechanical design of a height-adjustable bracket system for the Telescope Simulator imaging mirror M2 (see RD7). The M2 bracket in the system should be no larger than the dimensions in figure 1 below, in order to accommodate the other components (namely F3+mount with its 2 actuators, not shown below). Only ~<25mm max in width are available for the interface structure (full height) with M2. However M2 should have a weight of ~12-15kg. So if a large base is needed for stability of the bracket + mirror system, the bracket can seat on a base-plate up to ~100mm max in width (furthest away from M2) but restricted to ~10-20mm max in height.



**Figure 7:** Central part of the Telescope Simulator with the imaging mirror M2 and the 3 rotating flat mirrors F1, F2, and F3. The physical thickness of M2 is shown together with the estimated available space for the mirror bracket system (see below for suggested basic bracket shape).

Below is shown suggested basic idea, not to scale, for the bracket for M2 (with a suggested material: standard aluminium alloy).





- NOTES (FROM DX.REF SPIRE-RAL-NOT 621, 622 AND 734)
1. A FIDUCIAL MARK SHOULD LOCATE THE CENTRE OF THE OPTICAL SURFACE.
  2. EXTRA POLISHING OF THE CENTRE REGION ( $>10\text{cm}^2$ ) IS REQUIRED.
  3. THE OPTICAL SURFACE (BICONIC) MUST ACCURATELY FOLLOW THE EQUATION SPECIFIED BELOW, BUT THE MACHINED SURFACE IS ONLY REQUIRED TO BE IN THE ORDER OF 5 TO 10 MICRONS R.M.S.

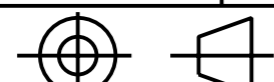
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$$Z = \frac{C_x X^2 + C_y Y^2}{1 + \sqrt{1 - (1 + K_x) C_x^2 X^2 - (1 + K_y) C_y^2 Y^2}}$$

SPIRE MASTER DRAWING	
PROJECT MEMBER	APPROVED
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PA GROUP	
STRESS ENG	
OPTICAL ENG	
THERMAL ENG	
MECHANICAL ENG	

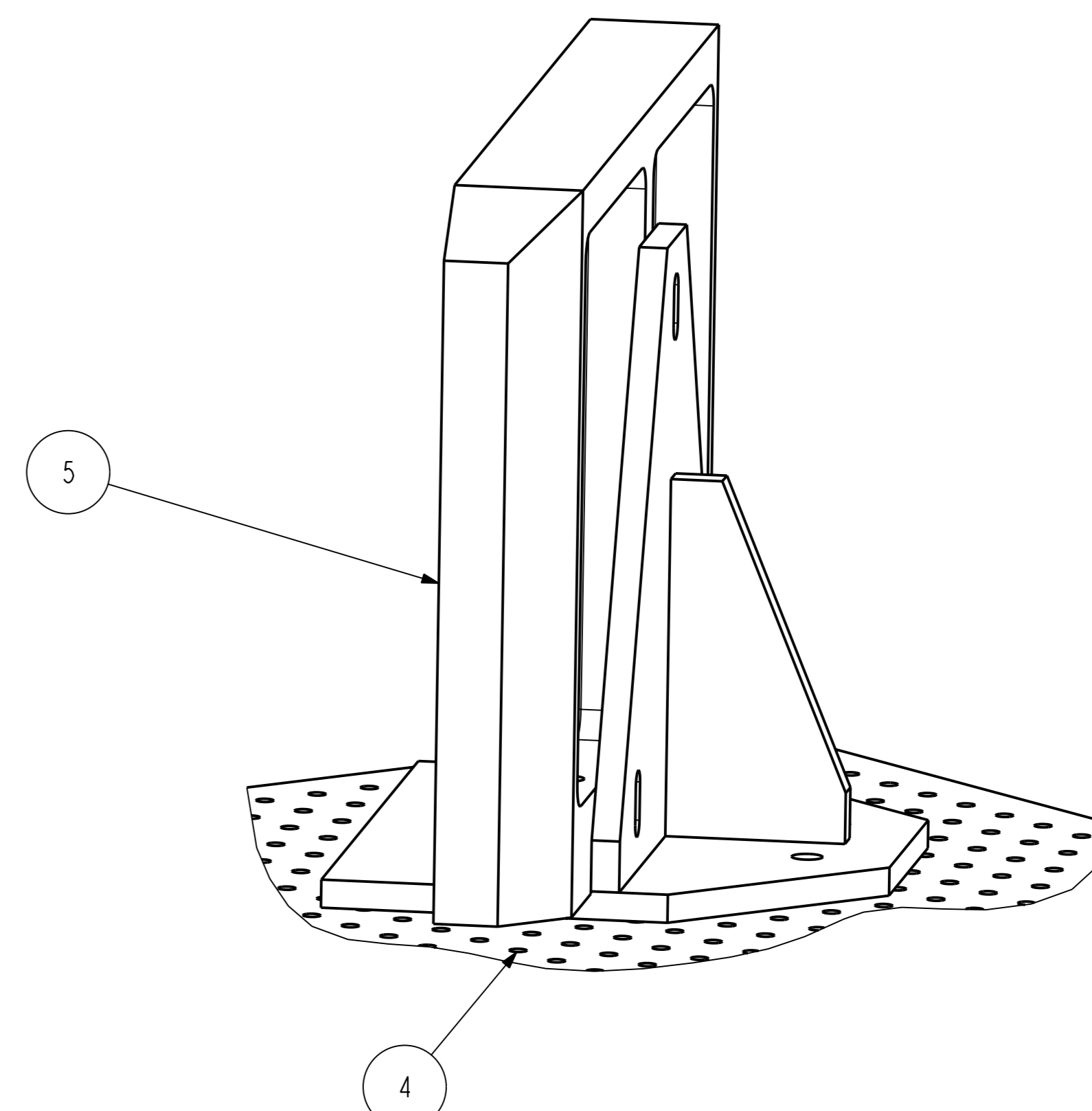
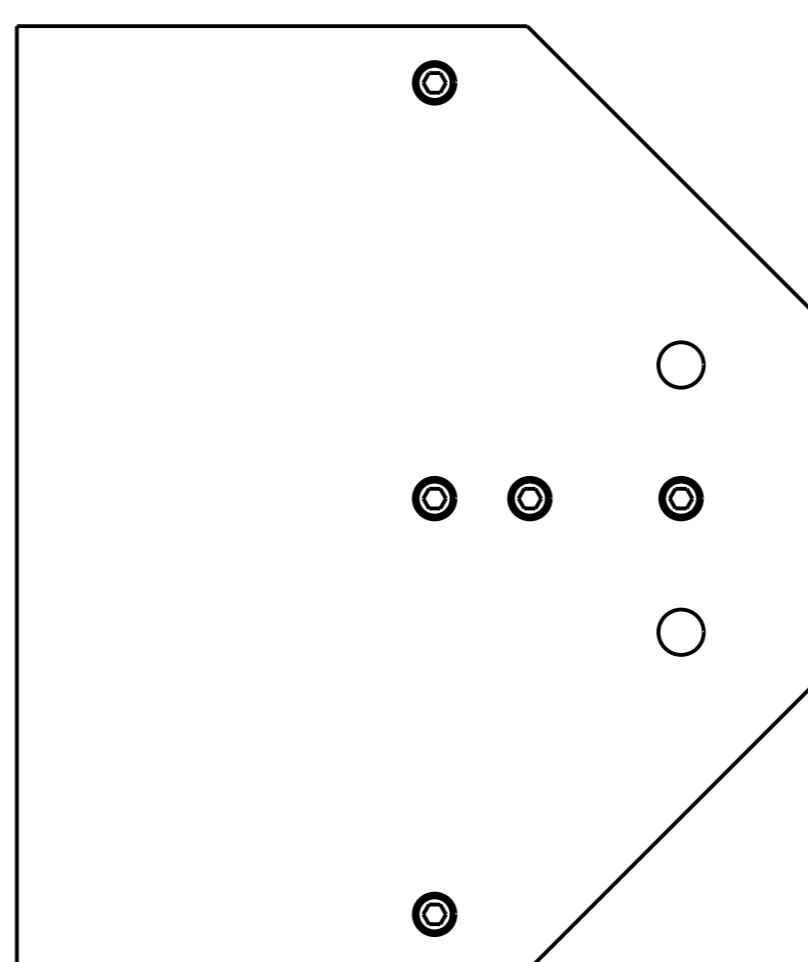
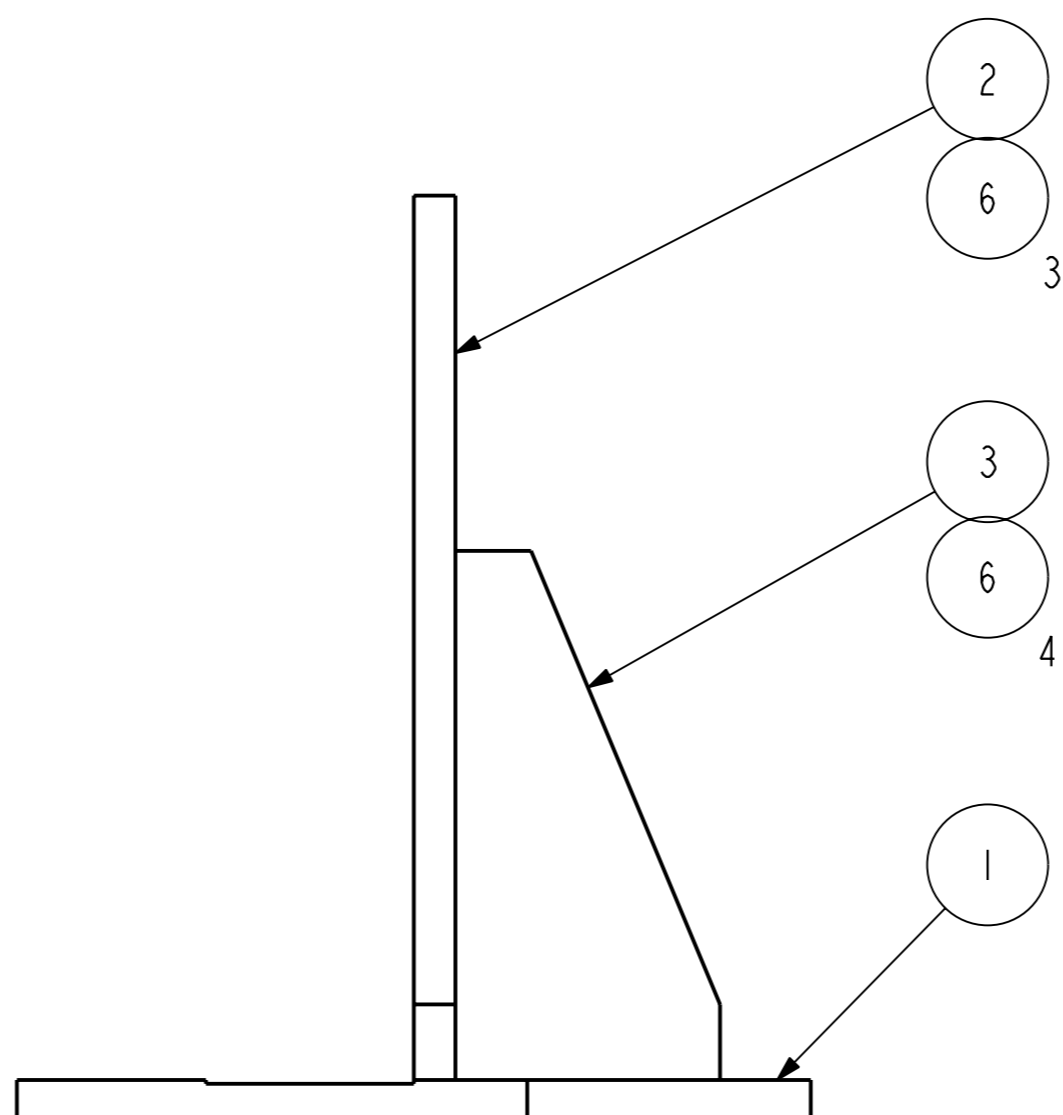
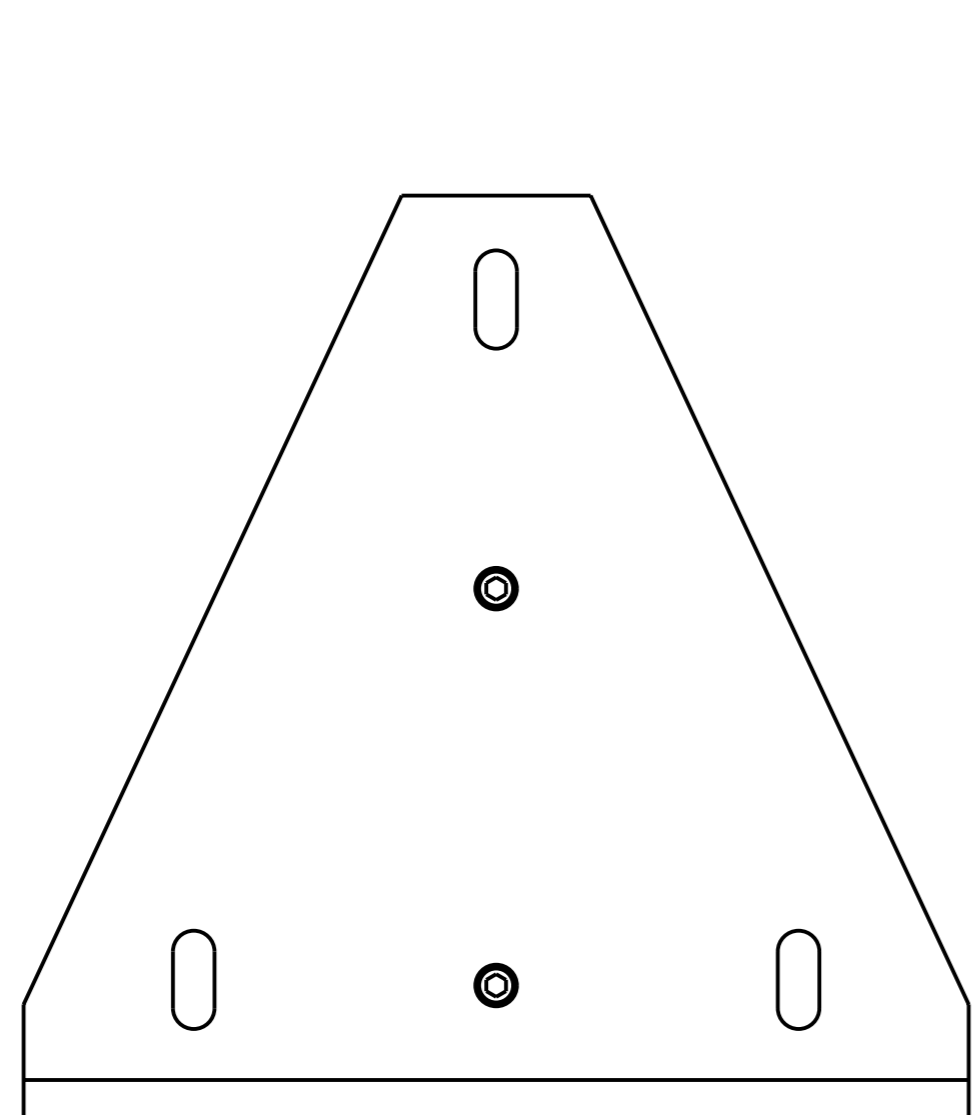
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ALUM ALLOY			SEE NOTES			
BS 1470 6082 (T6)			✓ UNLESS STATED			
USED ON					©CLRC 1996	
CENTRAL LABORATORY OF THE RESEARCH COUNCILS						
TITLE MIRROR M2, TELESCOPE SIMULATOR						
A I-KG-0720-500-B					1 OF 1	

A 1-KE-0104-330-A

PROJECTION 

THIS DRAWING CONFORMS TO B.S. 308

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2	KE-0104-332	A	M2 MOUNT FRONT	1	
3	KE-0104-333	A	M2 MOUNT WEB	1	
4		A	OPTICAL_BASE	1	
5		B	MIRROR M2, TELESCOPE SIMULATOR	1	
6			CAP HEAD SCREW M6 x 25 LONG	7	



A 1-KE-0104-330-A

SPIRE MASTER DRAWING	
PROJECT MEMBER	APPROVED
PROJECT MANAGER	
SYSTEM ENG	
ELECTRONICS ENG	
PA GROUP	
STRESS ENG	
OPTICAL ENG	
THERMAL ENG	
MECHANICAL ENG	

ISSUE	DATE	MOD. NO.	DRN. BY	CHKD.	APPD.	STATUS
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MATERIAL & SPEC. SEE DETAILS			SURFACE TEXTURE µm		0 _____ 50 mm	
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USED ON						©CLRC 1996
CENTRAL LABORATORY OF THE RESEARCH COUNCILS						
TITLE M2 MIRROR MOUNT ASSY						
SPIRE TELESCOPE SIMULATOR						
A 1-KE-0104-330-A					1 OF 1	